

NASA Contractor Report 172303

(NASA-CR-172303-Vol-2) INVESTIGATION OF THE
RELATIONS BETWEEN RESIN AND ADVANCED
COMPOSITE MECHANICAL PROPERTIES. VOLUME 2:
APPENDICES Technical Report, May 1982 - May
1983 (Wyoming Univ.) 195 p HC A09/AF A01

N85-14880

Unclass

G3/24 13026

INVESTIGATION OF THE RELATIONS BETWEEN NEAT RESIN AND ADVANCED COMPOSITE MECHANICAL PROPERTIES

VOLUME II - APPENDICES

Richard S. Zimmerman
Donald F. Adams
David E. Walrath

UNIVERSITY OF WYOMING
Laramie, Wyoming

Grant NA G1-277
November 1984



National Aeronautics and
Space Administration

Langley Research Center
Hampton, Virginia 23665

NASA REPORT NO.
NASA CR-172303

WYOMING REPORT NO.
UWME-DR-301-101-1

INVESTIGATION OF THE RELATIONS BETWEEN
NEAT RESIN AND ADVANCED COMPOSITE MECHANICAL PROPERTIES

VOLUME II - APPENDICES

RICHARD S. ZIMMERMAN
DONALD F. ADAMS
DAVID E. WALRATH

NOVEMBER 1984

TECHNICAL REPORT
NASA-~~L~~ANGLEY RESEARCH CENTER
HAMPTON, VIRGINIA 23665

RESEARCH GRANT NO. NAG-1-277

COMPOSITE MATERIALS RESEARCH GROUP
MECHANICAL ENGINEERING DEPARTMENT
UNIVERSITY OF WYOMING
LARAMIE, WYOMING 82071

APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED

PREFACE

This technical report presents the results of a one-year neat resin characterization and micromechanical prediction program sponsored by NASA-Langley Research Center, under Research Grant NAG-1-277.

All work was performed by the Composite Materials Research Group (CMRG) within the Department of Mechanical Engineering at the University of Wyoming. Co-Principal Investigators were Mr. Richard S. Zimmerman, Staff Engineer and Dr. Donald F. Adams, Professor. The analytical micromechanics studies were under the direction of Mr. David E. Walrath, Staff Engineer. Making major contributions to the program were Larry G. Adams, Beth E. Rogers, and Edward D. Schaffer, graduate students in Mechanical Engineering, and Eric Q. Lewis, Craig H. Johnson, Jeffrey A. Kessler, Thomas A. Ohnstad, Mark Vanderbleek, and Donald B. Hardy, undergraduate students in Mechanical Engineering, and all members of the Composite Materials Research Group. Substantial help in casting neat resin specimens was provided by Mr. Edwin M. Odom, Staff Engineer.

Use of commercial products or names of manufacturers in this report does not constitute official endorsement of such products or manufacturers, either expressed or implied, by the National Aeronautics and Space Administration.

PRECEDING PAGE BLANK NOT FILMED

TABLE OF CONTENTS

VOLUME I
(Under Separate Cover)

| <u>Section</u> | | <u>Page</u> |
|----------------|--|-------------|
| 1 | SUMMARY. | 1 |
| 2 | INTRODUCTION | 9 |
| 3 | EXPERIMENTAL RESULTS | 11 |
| | 3.1 Introduction | 11 |
| | 3.2 Tension Testing Results. | 11 |
| | 3.3 Torsional Shear Results. | 23 |
| | 3.4 Fracture Toughness Results | 43 |
| | 3.5 Coefficient of Thermal Expansion Results | 43 |
| | 3.6 Coefficient of Moisture Expansion Results. | 44 |
| 4 | SCANNING ELECTRON MICROSCOPE RESULTS | 49 |
| | 4.1 Introduction | 49 |
| | 4.2 Specimen Preparation | 49 |
| | 4.3 Explanation of SEM Photographs | 49 |
| | 4.4 Neat Resin Tension | 51 |
| | 4.5 Neat Resin Torsional Shear | 64 |
| | 4.6 Fracture Toughness Tests | 87 |
| 5 | MICROMECHANICS PREDICTIONS OF COMPOSITE RESPONSE | 93 |
| | 5.1 Introduction | 93 |
| | 5.2 Micromechanics Predictions Methodology | 95 |
| | 5.3 Constituent Material Properties. | 96 |
| | 5.3.1 Matrix Materials | 96 |
| | 5.3.2 Fiber Properties | 137 |
| | 5.4 Predicted Unidirectional Composite Response. | 137 |
| | 5.4.1 AS4/2220-1 Unidirectional Composite. | 139 |

~~UNCLASSIFIED~~

TABLE OF CONTENTS
(Continued)

| <u>Section</u> | <u>Page</u> |
|--|-------------|
| 5.4.1.1 Hygrothermal Initial Stress States. . . | 140 |
| 5.4.1.1.1 Cooldown From Curing Temperature. . | 140 |
| 5.4.1.1.2 Heating to 100°C. | 145 |
| 5.4.1.1.3 Moisture Saturation at Room Temperature | 147 |
| 5.4.1.1.4 Moisture Saturation at 100°C. . . . | 150 |
| 5.4.1.2 Mechanical Loadings | 153 |
| 5.4.1.2.1 Longitudinal Tension. | 153 |
| 5.4.1.2.2 Transverse Tension. | 165 |
| 5.4.1.2.3 Longitudinal Shear. | 175 |
| 5.4.2 AS4/3502 Unidirectional Composite. . . . | 196 |
| 5.4.3 AS4/2220-3 Unidirectional Composite. . . | 205 |
| 5.4.4 AS4/914 Unidirectional Composite | 215 |
| 5.5 Comparisons of Predicted Composite Response. . . . | 237 |
| 5.5.1 Hygrothermal Residual Stresses | 241 |
| 5.5.2 Mechanical Loadings. | 246 |
| 5.5.2.1 Longitudinal Tensile Loading. | 246 |
| 5.5.2.2 Transverse Tensile Loading. | 246 |
| 5.5.2.3 Longitudinal Shear Loading. | 250 |
| 6 CONCLUSIONS and RECOMMENDATIONS. | 253 |
| 6.1 Conclusions. | 253 |
| 6.2 Recommendations. | 254 |
| References | 255 |

TABLE OF CONTENTS
(Continued)

VOLUME II

| <u>Section</u> | <u>Page</u> |
|--|-------------|
| Appendices | 1 |
| Appendix A -- Neat Resin Casting Techniques | 1 |
| Appendix B -- Test Methods. | 13 |
| Appendix C -- Individual Test Results and Stress-Strain Curves. | 25 |
| Appendix D -- Additional SEM Fracture Surface Photographs . . . | 61 |
| Appendix E -- Plots of Internal Stress States in AS4 Graphite Fiber-Reinforced Unidirectional Composites Incorporating 3502, 2220-3, and 914 Matrix Materials | 97 |
| E1 -- AS4/3502 Graphite/Epoxy Unidirectional Composite . . | 99 |
| E2 -- AS4/2220-3 Graphite/Epoxy Unidirectional Composite . | 131 |
| E3 -- AS4/914 Graphite/Epoxy Unidirectional Composite. . . | 163 |

APPENDIX A

NEAT RESIN CASTING TECHNIQUES

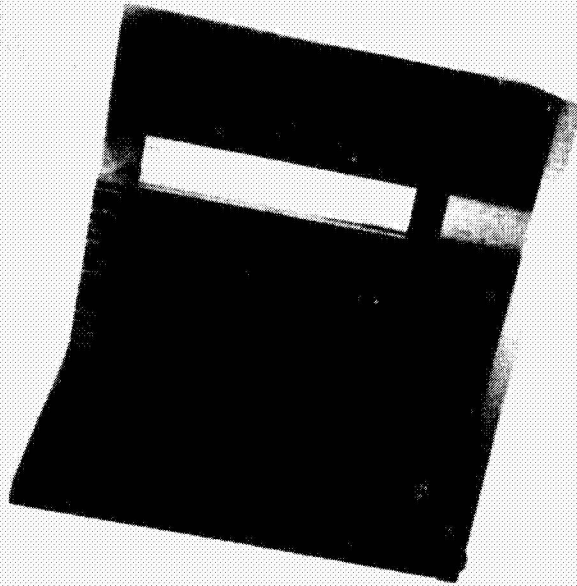
A.1 Introduction

The Composite Materials Research Group has been very active in the development of methods to cast thermoset resins into the necessary shapes for mechanical and physical testing. This fabrication of neat resins has evolved over the last 5-6 years into a fairly routine process. Nine resin systems from several resin formulators have been molded including Hercules 3501-6, 3502, 2220-1, 2220-3, and 4001, Shell Epon 9101, Ciba-Geigy Fibredux 914 and Fiberite 930 and 934. Various specimen configurations were needed for the present program for tensor, torsion, fracture toughness, coefficient of thermal expansion, and coefficient of moisture expansion measurements.

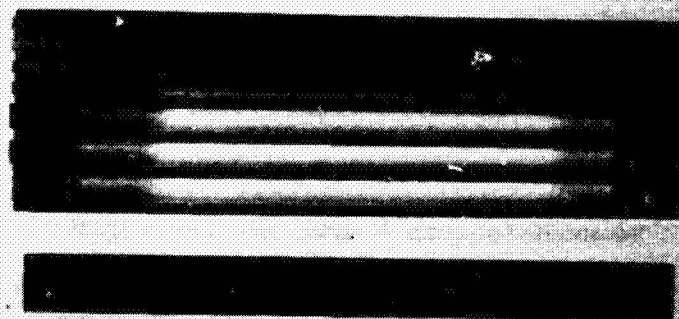
A.2 Fabrication of Tension Specimens

All tension specimens were prepared using a 17.75 cm x 17.75 cm x 22 cm steel box mold shown in Figure Ala. This box mold allows the casting of 32 specimens at one time, each 1.52 cm wide by 0.254 cm thick. As can be seen in both Figure Ala and Figure Alb, thin polished steel dividers are spaced equally in the bottom of the box mold using small steel spacers at each end. All steel pieces were sprayed at room temperature with a fast drying aerosol release agent, Miller-Stephenson MS-122, prior to assembling the mold. The assembled mold was preheated to 99°C in an oven. When the mold reached 100°C, a weighed quantity (up to 300 grams) of frozen resin was placed in the mold and the assembly was placed in a vacuum oven at 100°C. When the resin was completely melted, a vacuum was drawn on the system. The resin foamed vigorously

OF FOUR QUALITY



a.) Steel Box Mold With Dividers and Spacers.



b.) Steel Dividers and Spacers Used in Box Molds.

Figure A1. Steel Box Mold Used to Cast Flat Neat Resin Specimen Blanks.

and care was required to prevent overflowing the mold. This step was necessary to remove entrapped air and volatiles from the resin and is the reason for such a tall mold wall. Vacuum was continued until the resin was free of air and bubbles in the bottom of the mold. Then excess resin above the steel dividers was squeegeed off and the mold placed in an air-circulating oven at approximately 135°C for 5 hours. The mold was then disassembled and the individual moldings sanded to remove flashing. The moldings were then placed in a preheated oven while being held snugly between two smooth pieces of steel and postcured at 177°C between 1 and 5 hours depending upon the resin. An auxiliary thermocouple was placed on the surface of one of the specimens to ensure proper temperature control for the entire time. After postcure, the specimens were ground to uniform widths using a surface grinder. The tensile bars were then placed in a jig and routed with a diamond-coated router bit to the proper dogbone shape. The specimens were then placed in desiccators or moisture chambers for storage or moisture saturation. Finished tensile specimens are shown in Figure A2.

A.3 Fracture Toughness Specimen Casting Procedure

The notched-bend toughness specimens were cast in a similar fashion as the tensile specimens. Only the small spacers were changed in the square steel mold to allow nineteen 0.635 cm thick specimens to be cast. The 1.52 cm width was sufficient for these fracture toughness specimens. After grinding the specimens, three notches were cut in one side of the specimen to approximately one-half the width. Figure A3 is a photograph of the fracture toughness bars with notches cut. Three independent tests can be made from one bar since the span in the three-point bend fixture is only 5.00 cm and the two outer notches are totally out of the loading

ORIGINAL
OF POOR QUALITY

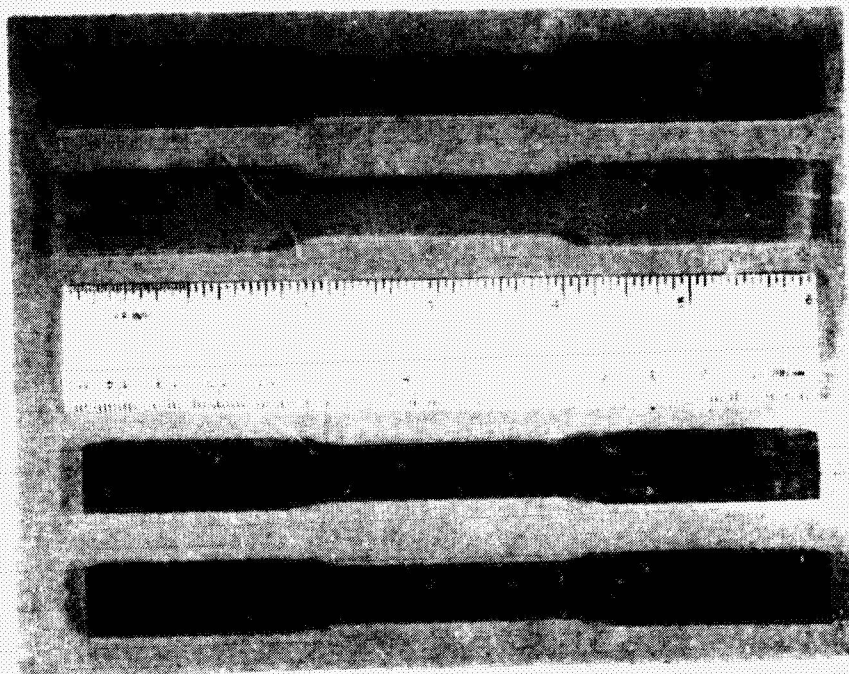


Figure A2. Typical Tensile Specimens Routed to the Required Dogbone Shape.

ORIGINAL SPECIMEN
OF POOR QUALITY

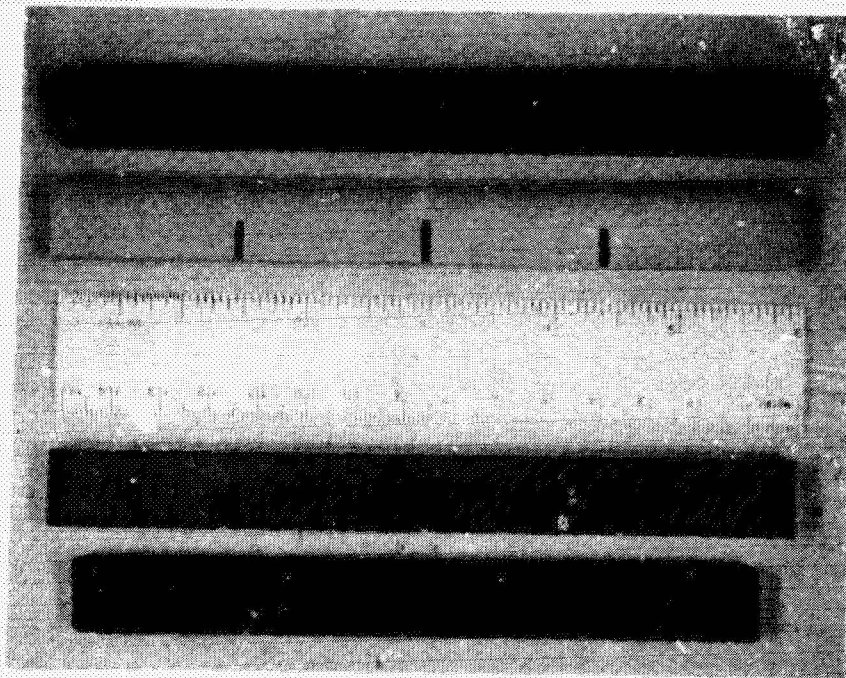


Figure A3. Typical Notched Bend Fracture Toughness Specimens.

area.

A.4 Fabrication of Thin Flat Specimens

Coefficient of moisture expansion (CME) specimens were cast in the same square steel box molds used for the tensile bars, without spacers or dividers. Approximately 100 grams of frozen resin was placed in the preheated mold and the curing times and temperatures duplicated from the tensile bar casting procedure. This mold yielded four specimens about 0.125 cm thick and 7 cm square after cutting the large square piece using a diamond-coated saw blade. A surface grinder was used to reduce the thickness to the 0.080 cm thickness used in the experiments. Figure A4 is a photograph of typical specimens used in this program.

Large flat specimens were also fabricated using procedures similar to those used for the CME specimens. Approximately 200 grams of frozen resin were used in an open box mold to make one 17 cm x 17 cm x 0.32 cm specimen for each of the four resin systems.

A.5 Fabrication of Cylindrical Torsion Specimens

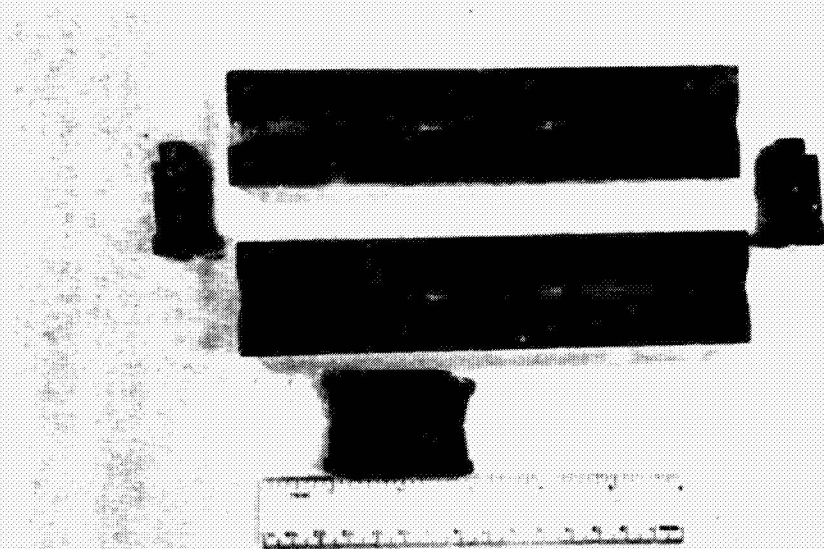
The Hercules 3502, 2220-1, and 2220-3 epoxies were cast using single-cavity split steel molds which were ground to the required dogbone shape. Figure A5 contains photographs of these steel molds. As can be seen in the figure, the mold consisted of three pieces screwed together to form the mold cavity. A silicone rubber funnel (Figure A5b) was placed over the open end of the mold and the mold preheated to 100°C after being sprayed with the Miller-Stephenson release agent. Figure A5b shows the assembled mold and rubber funnel. Approximately 21 grams of frozen resin were placed in the rubber funnel and allowed to melt into the mold. A vacuum was then pulled in the oven and regulated to prevent foam from overflowing the funnel. When the resin had been sufficiently

GROUP
OF POOR QUALITY



Figure A4. Typical Coefficient of Moisture Expansion (CME)
Specimens.

ORIGINAL PHOTO
OF POOR QUALITY



a.) Split Steel Mold Unassembled (Fastener Bolts also Shown).



b.) Assembled Mold With Rubber Funnel on Top for Holding Resin During Melting and Volatiles Removal.

Figure A5. Split Steel Mold Used to Cast Cylindrical Dogbone Specimens.

degassed, which required about 20-30 minutes, the molds were placed in an air-circulating oven at 135°C with the rubber funnels removed. The specimens were cured for 2-5 hours, depending on the resin system, and then removed from the split molds after being allowed to cool. The round dogbones were then postcured at 177°C, placing them between two stiff rubber sheets having a middle rib to support the center section should the specimens become slightly soft during postcure. A thermocouple was placed between the sheets to monitor specimen temperature and ensure the epoxy was maintained at the proper temperature.

After postcuring, the round specimens were polished with 400-grit and then 600-grit wet/dry emery paper using a small lathe to turn the specimens. This process removed all mold flashing and produced a consistent finish for all test specimens. After inspection the specimens were labeled and placed in desiccators or environmental chambers as appropriate for proper conditioning. Figure A6 shows typical round dogbone specimens ready for torsion testing.

The Fibredux 914 epoxy required slightly different processing to assure a reasonable test specimen without bubbles or voids. Silicone rubber molds were cast around a steel mandrel shaped to the round dogbone form. This produced a hard rubber form to cast the 914 epoxy in. The rubber molds were slit longitudinally to remove the steel mandrel, using a sharp knife, and the internal surface of the rubber mold was inspected for defects. After inspection the rubber molds were used in the same fashion as the steel molds to cast the 914 specimens. After curing, the 914 specimens were removed from the slit rubber molds and fully cured for one hour at 177°C held between the flat rubber sheets used with the other three resin systems. The 914 epoxy presented more

OF POOR QUALITY

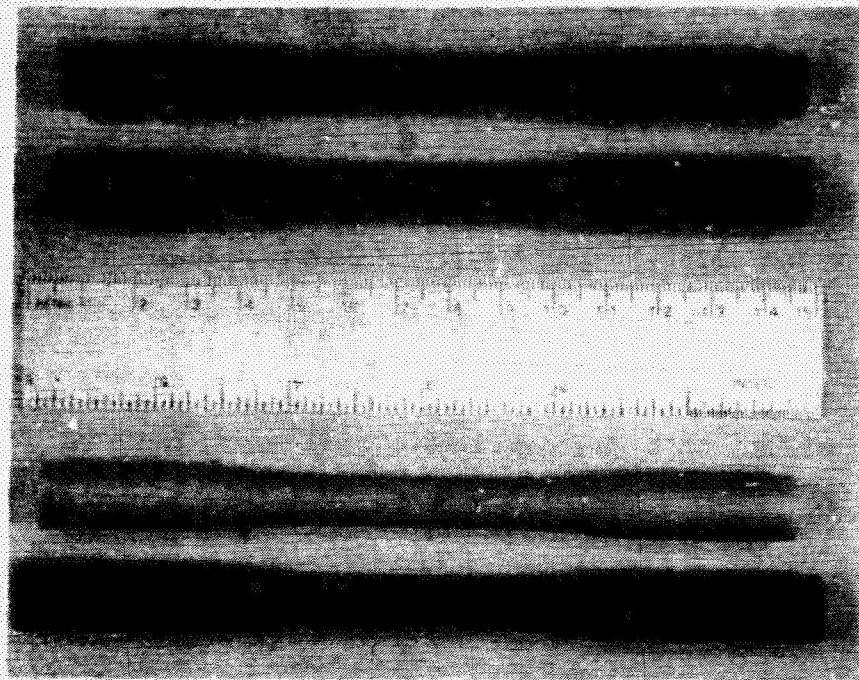


Figure A6. Typical Cylindrical Dogbone Specimens Used for Torsion Testing.

problems in the casting process due to its greater degree of shrinkage during cure. The rubber molds, having a much higher coefficient of thermal expansion, better compensated for the shrinkage in the 914 epoxy than the steel molds. A rubber mold is shown cut into two halves for viewing purposes in Figure A7.

A.6 Problems

Several problems in resin casting were encountered during this program. The 914 epoxy shrank more than the three other systems and seemed to be much more sensitive to processing temperature, i.e., it had a strong tendency to increase in viscosity before the air and volatiles had been fully removed. This resulted in more rejected specimens.

Some evidence of foreign particle contamination was encountered in all four resin systems. The fabrication process used by the Composite Materials Research Group was carefully monitored; it was concluded that the foreign particles were present in the frozen resin when received from the manufacturers and could not be removed by filtration prior to molding because of the high viscosity of these resins. These foreign particles occurred in the gage section of about one out of every twenty cast resin specimens. These specimens were discarded.

ORIGINAL FILED IN
OF POOR QUALITY

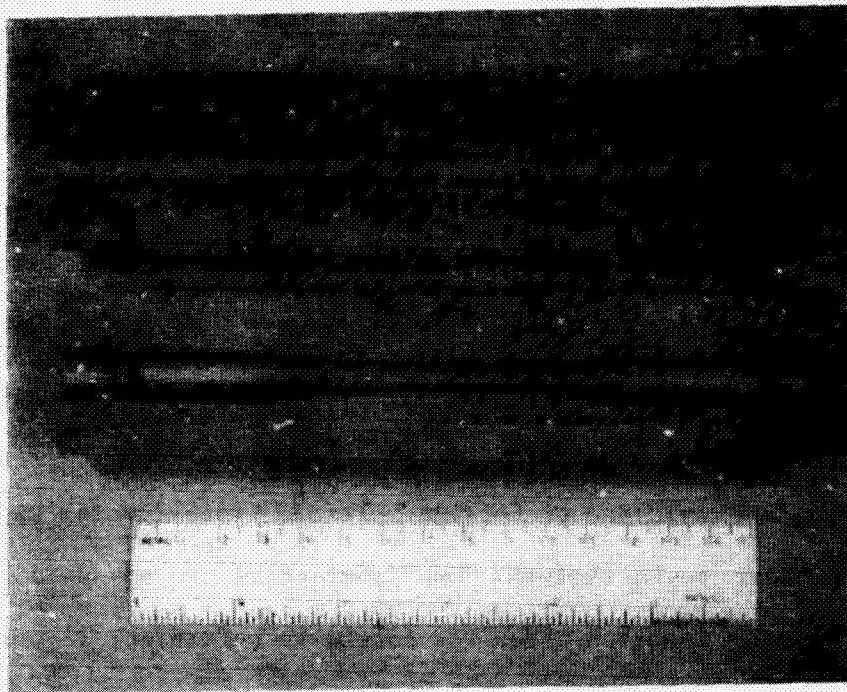


Figure A7. Silicone Rubber Mold Cut into Two Halves to Show the Dogbone Shape.

APPENDIX B

TEST METHODS

B.1 Tension Test Method

Tension tests were performed in an Instron Model 1125 electromechanical testing machine, using a loading rate of 2 mm/min. All data were recorded on a Hewlett-Packard HP 21 MX-E minicomputer which was used to plot the stress-strain curves presented in Appendix C. Tensile strains were measured using an extensometer or strain gage rosette depending on moisture conditioning and test temperature. Only an extensometer was used with the moisture-saturated specimens because of problems with gage adherence to the moisture-saturated surface. A transverse extensometer was used in conjunction with the longitudinal extensometer for calculation of Poisson's ratio. Figure B1 shows a typical test set-up using longitudinal and transverse extensometers. A 0°/90° strain gage rosette was used on all the dry specimens to measure the longitudinal and transverse strains. To verify that both of the strain measurement methods would yield equivalent strains on these materials of fairly low moduli, the room temperature, dry specimens were tested using both extensometers and strain gage rosettes.

Wedge grips were used on the flat dogbone tensile specimens. A 120-grit emery cloth was used in the grips to reduce the stress concentrations induced by the serrations on the grip faces, thereby avoiding the tendency of the neat resin specimens to fail in the grip area.

The dry neat resin specimens were stored in a desiccator to ensure the specimens remained dry. The moisture-saturated specimens were conditioned in a glass container filled with distilled water to a level

ORIGINAL FILED IN
OF POOR QUALITY

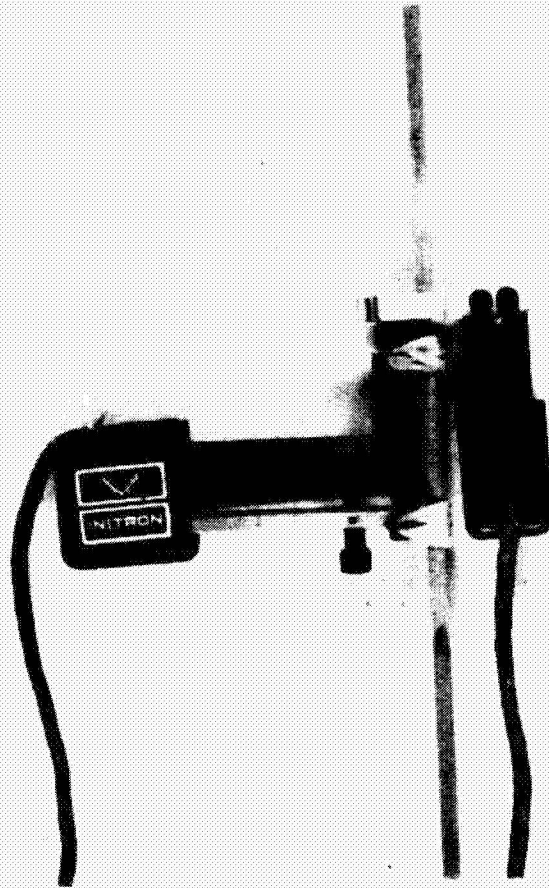


Figure B1. Typical Tension Test Setup With Two Extensometers.

The longitudinal extensometer with a one-inch gage length is on the left. The transverse extensometer is on the right using the specimen width as its gage length.

just below where the specimens were suspended, to maintain approximately 98% relative humidity. A temperature of 74°C was maintained by placing the container in a Tenney Benchmaster environmental chamber. Periodic weighings of witness specimens were done to determine when saturation of the neat resin specimens was achieved. After they reached saturation the specimens were removed from the elevated temperature chamber and stored in moist containers at room temperature to maintain them at their equilibrium levels. The specimens were moved in a beaker of water to the test machine in groups of five to ensure a moisture-saturated condition was maintained up to the point of the actual test. Only for the short time required to place the specimens in the grips and then perform the test were they in a dry environment.

A BEMCO environmental chamber was used for elevated temperature tests. This chamber has a temperature range from -129°C to 438°C. Three test temperatures were chosen for this program, viz, 23°C, 54°C, and 82°C.

B.2 Torsion Test Method

Solid rod torsion tests were performed in an Instron Model 1125 electromechanical testing machine configured for torsion testing. Collets were used to grip the round specimens with 120-grit emery paper being used to minimize the stress concentration on the test specimen and minimize slippage.

Angle of twist was measured using a "rotometer" developed at the University of Wyoming for this purpose. Two cams are attached to the specimen and an arrangement of four LVDT's convert the rotary movement into proportional linear signals. These signals are then subtracted in a Daytronix Model 9132 amplifier module to yield shear strain for input to

the 2IMX-E minicomputer. Figure B2 shows the specimen, cam and LVDT arrangement. A BEMCO environmental chamber was used for the elevated temperature.

B.3 Fracture Toughness Test Method

Fracture toughness testing was performed using ASTM Standard E 399-81. A three-point bend fixture was fabricated from the specifications found in the ASTM standard and is shown in Figure B3. A razor blade was used to scribe the notch tip. A fixture was developed to ensure the razor blade would cut the specimens equally when drawn through the fixture. Figure B4 is a photograph showing the razor blade held in a spring loading arm. The fixture was held firmly and the specimen drawn through the blade, providing a uniform cut at the notch tip. The specimens were then placed in the test fixture and loaded to failure while recording the loading history. Calculation of the plane strain fracture toughness K_{IC} was done using the following equation:

$$K_{IC} = \frac{3PL}{BW^2} a^{\frac{1}{2}} \left[1.93 - 3.07 \left(\frac{a}{W} \right) + 14.53 \left(\frac{a}{W} \right)^2 - 25.11 \left(\frac{a}{W} \right)^3 + 25.80 \left(\frac{a}{W} \right)^4 \right]$$

where P = load

L = specimen length

W = specimen width

B = specimen thickness

a = length of pre-crack and notch

The critical strain energy release rate G_{IC} is calculated from the following equation:

$$G_{IC} = \frac{K_{IC}^2}{E}$$

where E = elastic modulus

ORIENTED
DE POOR QUALITEIT

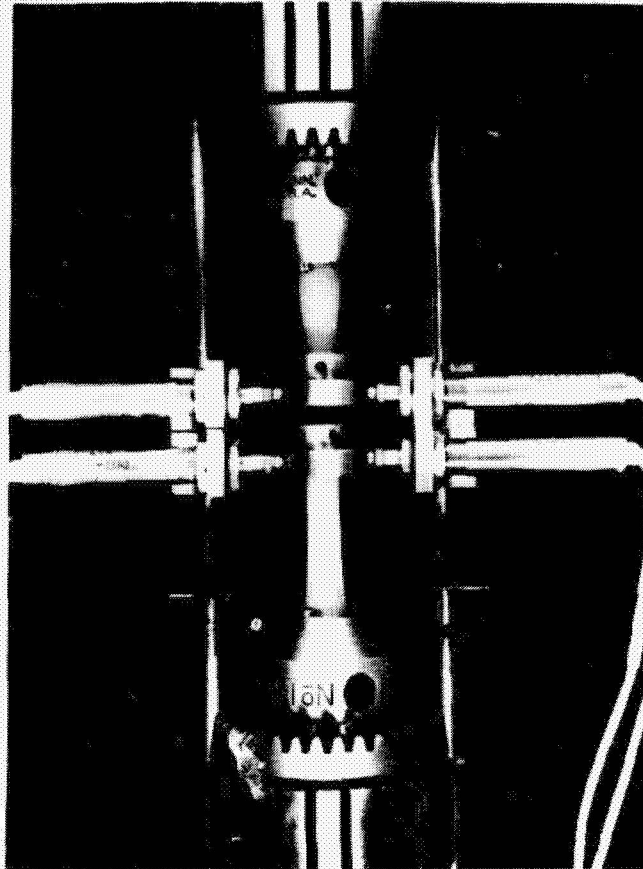


Figure B2. Torsional Shear Test Setup Showing Rotometer Mounted on Specimen.

ORIGINAL PHOTOGRAPH
OF POOR QUALITY

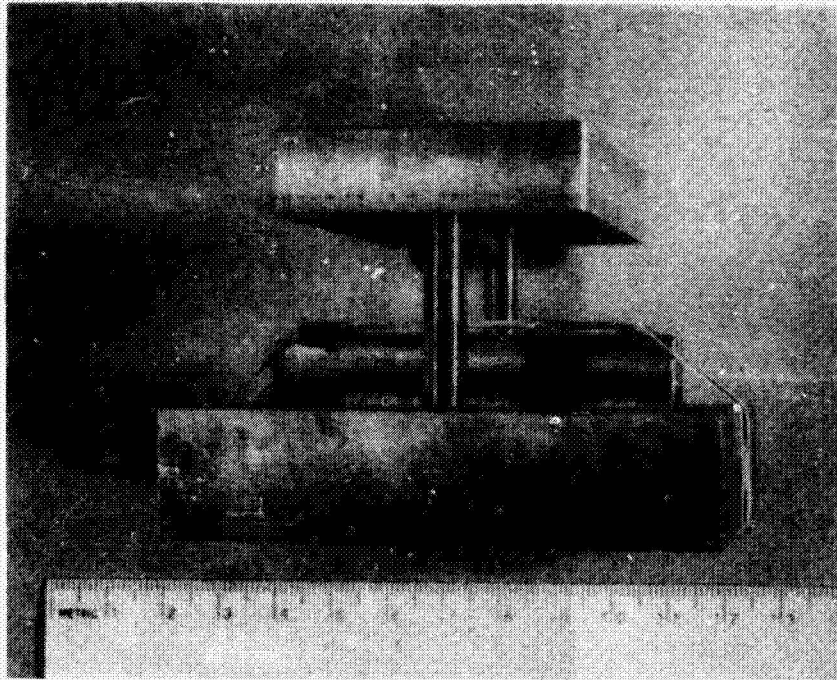


Figure B3. Single-Edge Notch Bend Test Fixture.

ORIGINAL PAGE IS
OF POOR QUALITY

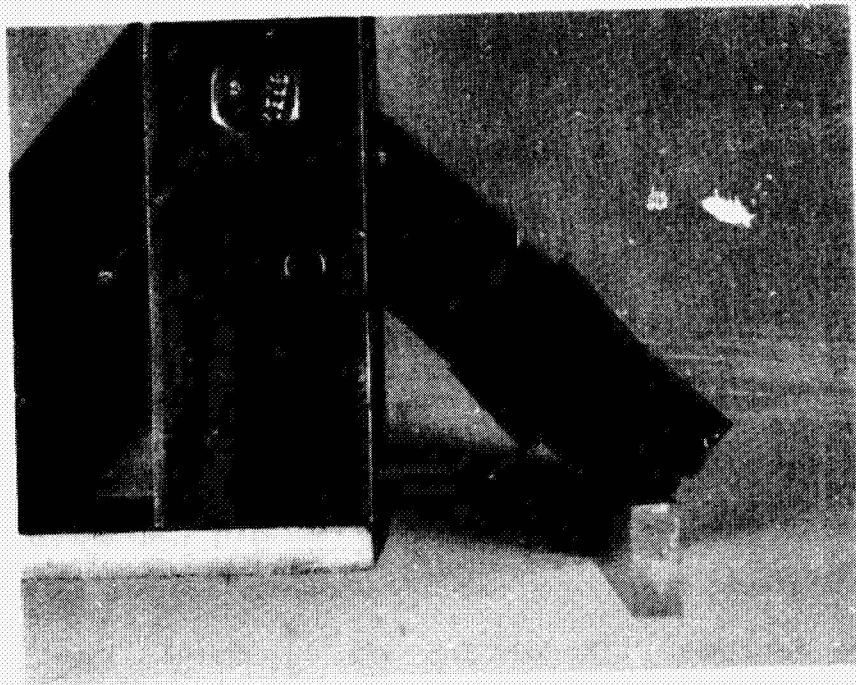


Figure B4. Notch Crack Initiator Apparatus For Single-Edge
Notch Bend Test Specimen.

B4. Coefficient of Thermal Expansion Test Method

Coefficient of thermal expansion (CTE) measurements were performed using a glass tube dilatometer linked to an LVDT. Figure B5 is a photograph of the test apparatus. The LVDT signal was amplified through a Daytronics Model 9130 amplifier and then to the Y-axis of a Hewlett-Packard Model 7004-B X-Y plotter. Temperature change was monitored using a T-type thermocouple attached to specimen and amplified through a Daytronics Model 9110 AT module then to the X-axis of the Hewlett-Packard Model 7004B X-Y plotter. A plot of expansion vs. temperature was thus generated. A heat-up rate of 1.1°C/min was used, over a range from room temperature to 93°C. Dry and moisture-saturated specimens were tested to determine the change in CTE due to moisture absorption. After the tests were completed, the data points were input to a computer curve-fit routine and an equation for the nonlinear CTE behavior was calculated. These equations were subsequently used in the micromechanics analysis program.

B.5 Coefficient of Moisture Expansion Test Method

Coefficient of moisture expansion (CME) measurements were performed on the four unreinforced epoxy systems using a glass tube dilatometer/LVDT apparatus to measure moisture expansion and an electronic balance to concurrently measure the weight gain of an identical witness specimen hung within the same environmental chamber. A photograph of the three-station apparatus is shown in Figure B6.

Moisture conditions were maintained by means of distilled water in open containers inside a plexiglass insert in the test ovens. A T-type thermocouple placed close to the two specimens monitored the constant 66°C temperature during the tests.

ORIGINAL PHOTOGRAPH
OF POOR QUALITY

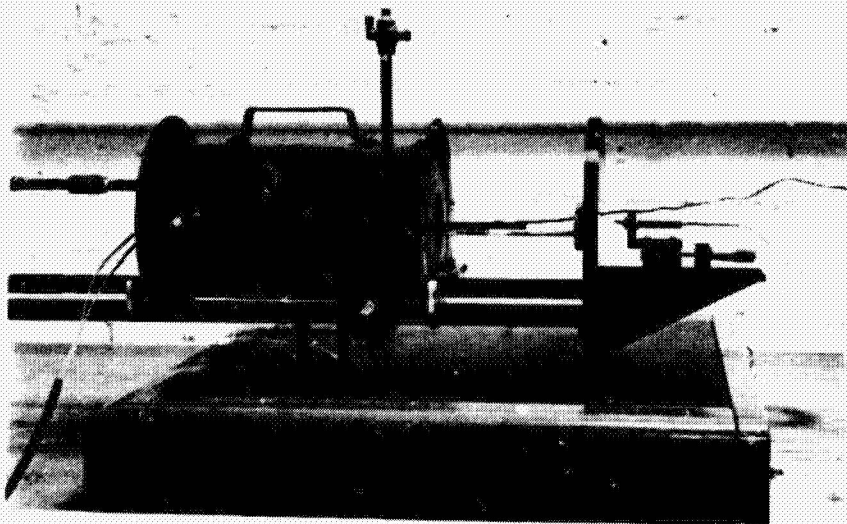


Figure B5. Coefficient of Thermal Expansion Test Apparatus.

ORIGINAL PAGE IS
OF POOR QUALITY

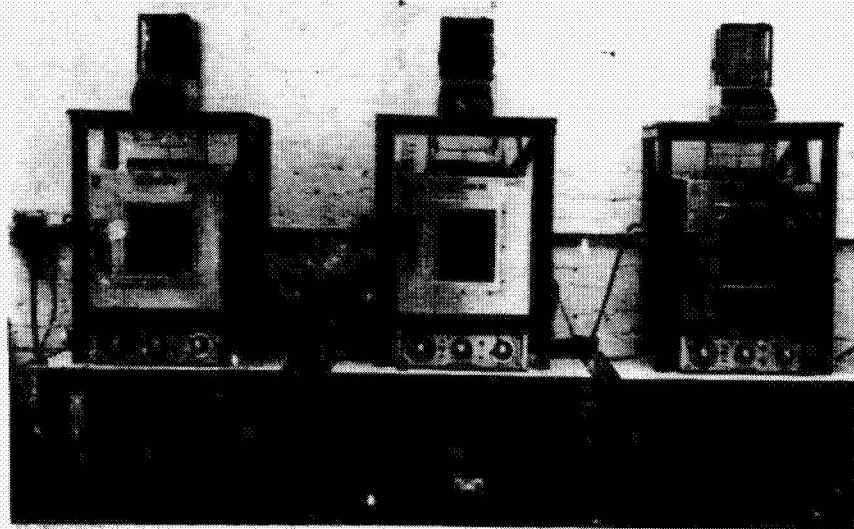


Figure B6. Coefficient of Moisture Expansion Test Apparatus.

A hole was drilled through the top of each oven to allow a linkage to a Sartorius balance mounted above, for weight gain measurement. A hole drilled in the side of each oven allowed the passage of the quartz glass rod of a dilatometer to the LVDT assembly for measurement of moisture expansion.

A 7 cm x 7 cm x 0.08 cm specimen configuration was used which was large enough to permit the assumption of one-dimensional moisture diffusion and the calculation of β , the coefficient of moisture expansion. The specimens were surface ground to size and then dried in a desiccator maintained at 93°C prior to testing to ensure a dry initial state.

Data were periodically recorded and then input to a computer curve-fit routine to calculate the relation to best fit the data. These equations were subsequently input to the micromechanics analysis program to predict moisture effects in the graphite/epoxy composites.

APPENDIX C

INDIVIDUAL TEST RESULTS AND STRESS-STRAIN CURVES FOR EACH TEST CONDITION

All test results were included in an initial average. If the standard deviation was greater than 10%, all values falling outside this standard deviation were removed, noted with an asterisk, and a new average and standard deviation were calculated.

PRECEDING PAGE BLANK NOT FILMED

~~FROM~~ 24 INTENTIONALLY BLANK

TABLE C1

INDIVIDUAL 3502 DRY TENSION RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (MPa) | Ultimate Stress (ksi) | Ultimate Strain | Modulus (GPa) | Modulus (Msi) | Poisson's Ratio |
|--------------------|--------------------------|--------------------------|--------------------------|-----------------|------------------|------------------|-----------------|
| LTDA 31 | 23 | 40.0 | 5.8 | 0.011 | 3.65 | 0.53 | 0.37 |
| 32 | | 31.7* | 4.6* | 0.009 | 3.65 | 0.53 | 0.37 |
| 33 | | 35.1 | 5.1 | 0.009 | 3.79 | 0.55 | 0.36 |
| 34 | | 32.4 | 4.7 | 0.009 | 3.86 | 0.56 | 0.39 |
| 35 | | 37.2 | 5.4 | 0.010 | 3.86 | 0.56 | 0.37 |
| 36 | | 43.4* | 6.3* | 0.012 | 3.65 | 0.53 | 0.35 |
| 37 | | 40.7 | 5.9 | 0.012 | 3.45 | 0.50 | 0.34 |
| average | | 37.2 | 5.4 | 0.010 | 3.65 | 0.53 | 0.36 |
| standard deviation | | 3.5 | 0.5 | 0.001 | 0.14 | 0.02 | 0.02 |
| LTDB 31 | 54 | 30.3 | 4.4 | 0.007 | 3.31 | 0.48 | 0.36 |
| 32 | | 28.3 | 4.1 | 0.009 | 3.24 | 0.47 | 0.38 |
| 33 | | 24.1* | 3.5* | 0.007 | 3.31 | 0.48 | 0.36 |
| 34 | | 31.7 | 4.6 | 0.011 | 3.31 | 0.48 | 0.36 |
| 35 | | 37.2* | 5.4* | 0.011 | 3.17 | 0.46 | 0.36 |
| average | | 30.3 | 4.4 | 0.009 | 3.24 | 0.47 | 0.36 |
| standard deviation | | 1.7 | 0.3 | 0.002 | 0.07 | 0.01 | 0.01 |
| LTDC 31 | 82 | 32.4 | 4.7 | 0.009* | 3.17 | 0.46 | 0.35 |
| 32 | | 39.3 | 5.7 | 0.013 | 3.10 | 0.45 | 0.37 |
| 33 | | 54.5* | 7.9* | 0.019 | 3.10 | 0.45 | 0.39 |
| 34 | | 46.2 | 6.7 | 0.016 | 3.17 | 0.46 | 0.37 |
| 35 | | 40.7 | 5.9 | 0.014 | 3.03 | 0.44 | 0.36 |
| average | | 42.1 | 6.1 | 0.016 | 3.10 | 0.45 | 0.37 |
| standard deviation | | 3.6 | 0.5 | 0.002 | 0.07 | 0.01 | 0.01 |

* Not included in average

TABLE C2

INDIVIDUAL FIBREDUX 914 DRY TENSION RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (MPa) | Ultimate Stress (ksi) | Ultimate Strain | Modulus (GPa) | Modulus (Msi) | Poisson's Ratio |
|--------------------|--------------------------|--------------------------|--------------------------|-----------------|------------------|------------------|-----------------|
| LIDA 21 | 23 | 25.5 | 3.7 | 0.006 | 4.27 | 0.62 | 0.38 |
| 22 | | 29.6 | 4.3 | 0.007 | 4.48 | 0.65 | 0.38 |
| 23 | | 29.6 | 4.3 | 0.007 | 4.00 | 0.58 | 0.37 |
| 24 | | 22.8* | 3.3* | 0.006 | 4.00 | 0.58 | 0.36 |
| 25 | | 22.8* | 3.3* | 0.006 | 4.00 | 0.58 | 0.37 |
| 26 | | 24.8 | 3.6 | 0.009 | 3.86 | 0.56 | 0.35 |
| 27 | | 33.8* | 4.9* | 0.009 | 3.79 | 0.55 | 0.34 |
| 28 | | 31.7 | 4.6 | 0.009 | 3.65 | 0.53 | 0.34 |
| average | | 28.3 | 4.1 | 0.007 | 4.02 | 0.58 | 0.36 |
| standard deviation | | 2.9 | 0.4 | 0.001 | 0.28 | 0.04 | 0.02 |
| LIDB 21 | 54 | 36.5* | 5.3* | 0.011 | 3.31 | 0.48 | 0.36 |
| 22 | | 28.9 | 4.2 | 0.009 | 3.38 | 0.49 | 0.35 |
| 23 | | 31.0 | 4.5 | 0.009 | 3.45 | 0.50 | 0.38 |
| 24 | | 31.0 | 4.5 | 0.007 | 3.31 | 0.48 | - |
| 25 | | 29.0 | 4.2 | 0.009 | 3.38 | 0.49 | 0.36 |
| average | | 30.3 | 4.4 | 0.009 | 3.37 | 0.49 | 0.36 |
| standard deviation | | 1.2 | 0.2 | 0.001 | 0.06 | 0.01 | 0.01 |
| LTDC 21 | 82 | 40.1* | 5.9* | 0.014 | 3.10 | 0.45 | 0.36 |
| 22 | | 31.7 | 4.6 | 0.010 | 3.31 | 0.48 | 0.37 |
| 23 | | 30.3 | 4.4 | 0.010 | 2.96 | 0.43 | 0.36 |
| 24 | | 33.1 | 4.8 | 0.012 | 2.49 | 0.42 | 0.37 |
| 25 | | 14.5* | 2.1* | 0.004* | 3.52 | 0.51 | 0.37 |
| average | | 31.7 | 4.6 | 0.012 | 3.17 | 0.46 | 0.37 |
| standard deviation | | 1.4 | 0.2 | 0.002 | 0.28 | 0.04 | 0.01 |

* Not included in average

TABLE C3

INDIVIDUAL 2220-1 DRY TENSION RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (MPa) | Ultimate Stress (ksi) | Ultimate Strain | Modulus (GPa) | Modulus (Msi) | Poisson's Ratio |
|--------------------|--------------------------|--------------------------|--------------------------|-----------------|------------------|------------------|-----------------|
| LTDA 01 | 23 | 42.7 | 6.2 | 0.004* | 9.72* | 1.41* | 0.34 |
| 2 | | 42.7 | 6.2 | 0.014 | 2.55 | 0.37 | 0.37 |
| 3 | | 44.1 | 6.4 | 0.014 | 3.31 | 0.48 | 0.37 |
| 4 | | 41.4 | 6.0 | 0.014 | 3.03 | 0.44 | 0.36 |
| 5 | | 45.5 | 6.6 | 0.015 | 3.10 | 0.45 | 0.36 |
| average | | 43.4 | 6.3 | 0.014 | 2.96 | 0.43 | 0.36 |
| standard deviation | | 1.4 | 0.2 | 0.001 | 0.34 | 0.05 | 0.01 |
| LTDB 01 | 54 | 57.2 | 8.3 | 0.008* | 2.96 | 0.43 | 0.19* |
| 2 | | 49.6 | 7.2 | 0.020 | 2.90 | 0.42 | 0.38 |
| 3 | | 60.7 | 8.8 | 0.019 | 2.96 | 0.43 | 0.36 |
| 4 | | 49.0* | 7.1* | 0.015 | 3.17 | 0.46 | 0.38 |
| 5 | | 68.9* | 10.0* | 0.019 | 2.96 | 0.43 | 0.37 |
| average | | 55.8 | 8.1 | 0.018 | 2.96 | 0.43 | 0.37 |
| standard deviation | | 5.6 | 0.8 | 0.002 | 0.07 | 0.01 | 0.01 |
| LTDC 01 | 82 | 73.8 | 10.7 | 0.024 | 2.48 | 0.36 | 0.19 |
| 2 | | 71.7 | 10.4 | 0.038* | 2.48 | 0.36 | 0.38 |
| 3 | | 73.8 | 10.7 | 0.006* | 2.83 | 0.41 | 0.36 |
| 4 | | 50.3* | 7.3* | 0.015 | 2.48 | 0.36 | 0.38 |
| 5 | | 51.7* | 7.5* | 0.023 | 2.83 | 0.41 | 0.37 |
| average | | 73.1 | 10.6 | 0.021 | 2.62 | 0.38 | 0.36 |
| standard deviation | | 1.2 | 0.2 | 0.005 | 0.21 | 0.03 | 0.02 |

* Not included in average

TABLE C4

INDIVIDUAL 2220-3 DRY TENSION RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (MPa) | Ultimate Stress (ksi) | Ultimate Strain | Modulus (GPa) | Modulus (Msi) | Poisson's Ratio |
|--------------------|--------------------------|--------------------------|--------------------------|-----------------|------------------|------------------|-----------------|
| LTDA 11 | 23 | 25.5* | 3.7* | 0.008* | 3.24 | 0.47 | 0.37 |
| 12 | | 42.7 | 6.2 | 0.014 | 3.17 | 0.46 | 0.37 |
| 13 | | 45.5 | 6.6 | 0.015 | 3.10 | 0.45 | 0.34 |
| 14 | | 35.1 | 5.1 | 0.011 | 3.31 | 0.48 | 0.37 |
| 15 | | 51.7 | 7.5 | 0.017 | 3.38 | 0.49 | 0.37 |
| 16 | | 51.0 | 7.4 | 0.018 | 3.24 | 0.47 | 0.36 |
| 17 | | 64.1* | 9.3* | 0.025* | 2.96 | 0.43 | 0.35 |
| average | | 45.5 | 6.6 | 0.015 | 3.17 | 0.46 | 0.36 |
| standard deviation | | 6.8 | 1.0 | 0.003 | 0.14 | 0.02 | 0.01 |
| LTDB 11 | 54 | 59.3 | 8.6 | 0.027 | 2.69 | 0.39 | 0.33 |
| 12 | | 82.0* | 11.9* | 0.038* | 2.96 | 0.43 | 0.37 |
| 13 | | 51.7 | 7.5 | 0.018* | 3.45 | 0.50 | 0.42 |
| 14 | | 49.6 | 7.2 | 0.021 | 2.96 | 0.43 | 0.35 |
| 15 | | 66.9 | 9.7 | 0.030 | 2.83 | 0.41 | 0.35 |
| average | | 56.5 | 8.2 | 0.026 | 2.96 | 0.43 | 0.36 |
| standard deviation | | 7.9 | 1.1 | 0.004 | 0.28 | 0.04 | 0.03 |
| LTD 11 | 82 | 54.5* | 7.9* | 0.029 | 2.48 | 0.36 | 0.39 |
| 12 | | 73.1 | 10.6 | 0.018 | 2.62 | 0.38 | 0.36 |
| 13 | | 64.8 | 9.4 | 0.031 | 2.28 | 0.33 | 0.34 |
| 14 | | 71.0 | 10.3 | 0.018 | 2.48 | 0.36 | 0.34 |
| 15 | | 71.7 | 10.4 | 0.036* | 2.41 | 0.35 | 0.35 |
| average | | 70.3 | 10.2 | 0.024 | 2.45 | 0.36 | 0.35 |
| standard deviation | | 3.7 | 0.5 | 0.007 | 0.13 | 0.02 | 0.02 |

* Not included in average

TABLE C5

INDIVIDUAL 3502 WET TENSION RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (MPa) | Ultimate Stress (ksi) | Ultimate Strain | Modulus (GPa) | Modulus (Msi) | Poisson's Ratio |
|--------------------|--------------------------|--------------------------|--------------------------|-----------------|------------------|------------------|-----------------|
| LTWA 31 | 23 | 37.2 | 5.4 | 0.011 | 3.45 | 0.50 | 0.42 |
| 32 | | 33.1 | 4.8 | 0.011 | 3.38 | 0.49 | 0.42 |
| 33 | | 34.4 | 5.0 | 0.010 | 3.79 | 0.55 | 0.43 |
| 34 | | 37.9 | 5.5 | - | - | - | - |
| 35 | | 44.8* | 6.5* | 0.014 | 3.52 | 0.51 | 0.44 |
| average | | 35.9 | 5.2 | 0.012 | 3.52 | 0.51 | 0.43 |
| standard deviation | | 2.3 | 0.3 | 0.001 | 0.14 | 0.02 | 0.01 |
| LTWB 31 | 54 | 26.2 | 3.8 | 0.009 | 2.96 | 0.43 | 0.40 |
| 32 | | 22.8 | 3.3 | 0.008 | 3.03 | 0.44 | - |
| 33 | | 32.4* | 4.7* | 0.012* | 2.96 | 0.43 | 0.38 |
| 34 | | 21.4 | 3.1 | 0.007 | 2.96 | 0.45 | 0.38 |
| 35 | | 22.1 | 3.2 | 0.007 | 3.10 | 0.45 | 0.38 |
| average | | 22.8 | 3.3 | 0.008 | 3.03 | 0.44 | 0.38 |
| standard deviation | | 2.0 | 0.3 | 0.001 | 0.07 | 0.01 | 0.01 |
| LTWC 31 | 82 | 19.3* | 2.8* | 0.008 | 2.55 | 0.37 | 0.41 |
| 32 | | 25.5 | 3.7 | 0.011 | 2.43 | 0.36 | 0.40 |
| 33 | | 27.6* | 4.0* | 0.011 | 2.48 | 0.36 | 0.42 |
| 34 | | 24.8 | 3.6 | 0.009 | 2.76 | 0.40 | 0.45 |
| 35 | | 24.1 | 3.5 | 0.010 | 2.55 | 0.37 | 0.44 |
| 36 | | 24.8 | 3.6 | - | - | - | - |
| average | | 24.8 | 3.6 | 0.010 | 2.58 | 0.37 | 0.42 |
| standard deviation | | 0.6 | 0.1 | 0.001 | 0.14 | 0.02 | 0.02 |

* Not included in average

TABLE C6

INDIVIDUAL FIBREDUX 914 WET TENSION RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (ksi) | Ultimate Strain | Modulus (GPa) | Modulus (Msi) | Poisson's Ratio |
|--------------------|--------------------------|--------------------------|-----------------|------------------|------------------|-----------------|
| LTTA 21 | 23 | 40.0 | 0.014 | 3.03 | 0.44 | 0.44 |
| 22 | | 32.4* | 0.010* | 3.24 | 0.47 | 0.44 |
| 23 | | 54.5 | 0.020 | 3.10 | 0.45 | 0.43 |
| 24 | | 44.8 | 0.016 | 3.03 | 0.44 | 0.40 |
| 25 | | 53.8 | 0.019 | 3.03 | 0.44 | 0.44 |
| average | | 48.3 | 0.017 | 3.10 | 0.45 | 0.43 |
| standard deviation | | 7.1 | 0.003 | 0.70 | 0.01 | 0.02 |
| LTTB 21 | 54 | 31.7 | 0.011 | 3.31 | 0.39 | 0.43 |
| 22 | | 31.7 | 0.009 | 3.38 | 0.37 | 0.40 |
| 23 | | 27.6 | 0.009 | 3.45 | 0.37 | 0.41 |
| 24 | | 32.4 | 0.007 | 3.31 | 0.37 | 0.45 |
| 25 | | 31.7 | 0.009 | 3.38 | 0.36 | 0.41 |
| 26 | | 35.2 | - | - | - | - |
| average | | 31.8 | 0.009 | 2.56 | 0.37 | 0.42 |
| standard deviation | | 2.4 | 0.001 | 0.70 | 0.01 | 0.02 |
| LTTT 21 | 82 | 35.9 | 0.020* | 2.14 | 0.31 | 0.41 |
| 22 | | 31.7 | 0.017 | 2.07 | 0.30 | 0.41 |
| 23 | | 29.6 | 0.014 | 2.28 | 0.33 | 0.40 |
| 24 | | 31.7 | 0.017 | 2.00 | 0.29 | 0.39 |
| 25 | | 30.3 | 0.015 | 2.14 | 0.31 | 0.38 |
| average | | 31.7 | 0.016 | 2.14 | 0.31 | 0.40 |
| standard deviation | | 2.1 | 0.002 | 0.70 | 0.01 | 0.01 |

* Not included in average

TABLE C7

INDIVIDUAL 2220-1 WET TENSION RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (MPa) | Ultimate Stress (ksi) | Ultimate Strain | Modulus (GPa) | Modulus (Msi) | Poisson's Ratio |
|--------------------|--------------------------|--------------------------|--------------------------|-----------------|------------------|------------------|-----------------|
| LTWA 01 | 23 | 62.1 | 9.0 | 0.024 | 3.10 | 0.45 | 0.40 |
| 2 | | 62.1 | 9.0 | 0.025 | 3.03 | 0.44 | 0.39 |
| 3 | | 81.4* | 11.8* | 0.029 | 3.17 | 0.46 | 0.41 |
| 4 | | 79.5 | 11.5 | 0.034 | 3.17 | 0.46 | 0.42 |
| 5 | | 68.3 | 9.9 | 0.029 | 3.10 | 0.45 | 0.41 |
| average | | 68.3 | 9.9 | 0.028 | 3.11 | 0.45 | 0.41 |
| standard deviation | | 8.3 | 1.2 | 0.004 | 0.06 | 0.01 | 0.01 |
| LTWB 01 | 54 | 52.4 | 7.6 | 0.026 | 2.76 | 0.40 | 0.46 |
| 2 | | 53.8 | 7.8 | 0.030 | 2.48 | 0.36 | 0.42 |
| 3 | | 57.9 | 8.4 | 0.034 | 2.55 | 0.37 | 0.43 |
| 4 | | 57.9 | 8.4 | 0.034 | 2.48 | 0.36 | 0.42 |
| 5 | | 62.1 | 9.0 | 0.038 | 2.56 | 0.37 | 0.43 |
| average | | 56.5 | 8.2 | 0.032 | 2.56 | 0.37 | 0.43 |
| standard deviation | | 4.1 | 0.6 | 0.004 | 0.12 | 0.02 | 0.02 |
| LTWC 01 | 82 | 46.2 | 6.7 | 0.038 | 2.14 | 0.31 | 0.44 |
| 2 | | 45.5 | 6.6 | 0.034 | 2.00 | 0.29 | 0.41 |
| 3 | | 46.2 | 6.7 | 0.037 | 2.07 | 0.30 | 0.42 |
| 4 | | 45.5 | 6.6 | 0.033 | 2.00 | 0.29 | 0.43 |
| 5 | | 47.6 | 6.9 | 0.042 | 2.07 | 0.30 | 0.43 |
| average | | 46.2 | 6.7 | 0.037 | 2.07 | 0.30 | 0.43 |
| standard deviation | | 0.7 | 0.1 | 0.003 | 0.07 | 0.01 | 0.01 |

* Not included in average

TABLE C8

INDIVIDUAL 2220-3 WET TENSION RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (MPa) | Ultimate Stress (ksi) | Ultimate Strain | Modulus (GPa) | Modulus (Msi) | Poisson's Ratio |
|--------------------|--------------------------|--------------------------|--------------------------|-----------------|------------------|------------------|-----------------|
| LTWA 11 | 23 | 75.1 | 10.9 | 0.036 | 3.03 | 0.44 | 0.44 |
| 12 | | 54.5 | 7.9* | 0.020* | 3.03 | 0.44 | 0.36 |
| 13 | | 61.4 | 8.9 | 0.024 | 3.17 | 0.46 | 0.44 |
| 14 | | 82.7 | 12.0* | 0.034 | 3.10 | 0.45 | 0.43 |
| 15 | | 63.4 | 9.2 | 0.037 | 2.96 | 0.43 | 0.45 |
| average | | <u>66.7</u> | <u>9.7</u> | <u>0.033</u> | <u>3.03</u> | <u>0.44</u> | <u>0.43</u> |
| standard deviation | | 7.4 | 1.1 | 0.006 | 0.07 | 0.01 | 0.03 |
| LTWB 11 | 54 | 63.4 | 9.2 | 0.037 | 2.55 | 0.37 | 0.46 |
| 12 | | 62.7 | 9.1 | 0.043 | 2.48 | 0.36 | 0.42 |
| 13 | | 55.8 | 8.1 | 0.032 | 2.55 | 0.37 | 0.43 |
| 14 | | 64.1 | 9.3 | 0.051* | 2.41 | 0.35 | 0.31* |
| 15 | | - | - | 0.035 | 2.21 | 0.32 | - |
| average | | <u>61.4</u> | <u>8.9</u> | <u>0.037</u> | <u>2.41</u> | <u>0.35</u> | <u>0.44</u> |
| standard deviation | | 4.1 | 0.6 | 0.005 | 0.14 | 0.02 | 0.02 |
| LTWC 11 | 82 | 37.9 | 5.5 | 0.052 | 2.28 | 0.33 | 0.45 |
| 12 | | 46.2 | 6.7 | 0.068* | 2.00 | 0.29 | 0.49 |
| 13 | | 43.4 | 6.3 | 0.042 | 2.00 | 0.29 | 0.46 |
| 14 | | 44.1 | 6.4 | 0.049 | 2.21 | 0.32 | 0.46 |
| 15 | | 48.3 | 7.0 | 0.048 | 2.34 | 0.34 | 0.49 |
| average | | <u>44.1</u> | <u>6.4</u> | <u>0.048</u> | <u>2.14</u> | <u>0.31</u> | <u>0.47</u> |
| standard deviation | | 3.9 | 0.6 | 0.004 | 0.14 | 0.02 | 0.02 |

* Not included in average

TABLE C9

INDIVIDUAL 3502 DRY SHEAR RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (MPa) | Ultimate Stress (ksi) | Ultimate Strain | Shear Modulus (GPa) | Shear Modulus (Msi) |
|--------------------|--------------------------|--------------------------|--------------------------|-----------------|------------------------|------------------------|
| LSDA 31 | 23 | 33.8 | 4.9* | 0.022 | 1.59 | 0.23 |
| 32 | | 64.8 | 9.4 | 0.042 | 2.07 | 0.30 |
| 33 | | 46.9 | 6.8 | 0.014* | 2.07 | 0.30 |
| 34 | | 68.3 | 9.9 | 0.047 | 1.52 | 0.22 |
| 35 | | 41.4* | 6.0* | 0.028 | 1.59 | 0.23 |
| 36 | | 79.3* | 11.5* | 0.051* | 2.27* | 0.33* |
| 37 | | 63.4 | 9.2 | 0.040 | 1.86 | 0.27 |
| 38 | | 57.2 | 8.3 | 0.039 | 1.45* | 0.21* |
| average | | 60.0 | 8.7 | 0.036 | 1.79 | 0.26 |
| standard deviation | | 8.3 | 1.2 | 0.009 | 0.20 | 0.03 |
| LSDB 31 | 54 | 90.3* | 13.1* | 0.067 | 1.59 | 0.23 |
| 32 | | 61.4 | 8.9 | — | 1.79* | 0.26* |
| 33 | | 69.6 | 10.1 | 0.057 | 1.03* | 0.15* |
| 34 | | 62.0 | 9.0 | 0.043 | 1.52 | 0.22 |
| 35 | | 76.5 | 11.1 | 0.060 | 1.59 | 0.23 |
| average | | 67.6 | 9.8 | 0.057 | 1.59 | 0.23 |
| standard deviation | | 6.9 | 1.0 | 0.010 | 0.03 | 0.01 |
| LSDC 31 | 82 | 71.0 | 10.3 | 0.058 | 1.52 | 0.22 |
| 32 | | 73.1 | 10.6 | 0.055 | 1.59 | 0.23 |
| 33 | | 60.0 | 8.7 | 0.047 | 1.59 | 0.23 |
| 34 | | 57.9* | 8.4* | 0.040* | 1.52 | 0.22 |
| 35 | | 75.8 | 11.0 | 0.058 | 1.59 | 0.23 |
| average | | 70.4 | 10.2 | 0.055 | 1.59 | 0.23 |
| standard deviation | | 7.0 | 1.0 | 0.005 | 0.03 | 0.01 |

* Not included in average

TABLE C10

INDIVIDUAL FIBREDUX 914 DRY SHEAR RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (MPa) | Ultimate Stress (ksi) | Ultimate Strain | Modulus (GPa) | Modulus (Msi) |
|--------------------|--------------------------|--------------------------|--------------------------|-----------------|------------------|------------------|
| LSDA 21 | 23 | 81.4 | 11.8 | 0.072 | 1.17* | 0.17* |
| 22 | | 74.5 | 10.8 | 0.050 | 1.86 | 0.27 |
| 23 | | 60.7* | 8.7* | 0.040 | 1.52 | 0.22 |
| 24 | | 83.4 | 12.1 | 0.074 | 1.24 | 0.18 |
| 25 | | 88.9* | 12.9* | 0.061 | 1.59 | 0.23 |
| 26 | | 59.3* | 8.6* | 0.035* | 2.00* | 0.29* |
| average | | 80.0 | 11.6 | 0.059 | 1.52 | 0.22 |
| standard deviation | | 4.8 | 0.7 | 0.014 | 0.28 | 0.04 |
| LSDB 21 | 54 | 63.4 | 9.2 | 0.049 | 1.59 | 0.23 |
| 22 | | 54.5 | 7.9 | 0.039 | 1.52 | 0.22 |
| 23 | | 86.2* | 12.5* | 0.068* | 1.72 | 0.25 |
| 24 | | 55.2 | 8.0 | 0.046 | 1.38 | 0.20 |
| 25 | | 68.3 | 9.9 | 0.045 | 1.45 | 0.21 |
| average | | 60.7 | 8.8 | 0.045 | 1.52 | 0.22 |
| standard deviation | | 6.8 | 1.0 | 0.004 | 0.14 | 0.02 |
| LSDC 21 | 82 | 51.7* | 7.5* | - | 0.83 | 0.12 |
| 22 | | 69.6 | 10.1 | 0.055 | 1.45 | 0.21 |
| 23 | | 73.1 | 10.6 | 0.070 | 0.76* | 0.11* |
| 24 | | 76.5 | 11.1 | 0.059 | 1.59 | 0.23 |
| 25 | | 95.8 | 13.9* | 0.060 | 1.52 | 0.22 |
| 26 | | 81.4* | 11.8 | - | - | - |
| average | | 75.1 | 10.9 | 0.061 | 1.17 | 0.20 |
| standard deviation | | 5.0 | 0.7 | 0.006 | 0.41 | 0.06 |

* Not included in average

TABLE C11

INDIVIDUAL 2220-1 DRY SHEAR RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (MPa) | Ultimate Stress (ksi) | Ultimate Strain | Modulus (GPa) | Modulus (Msi) |
|--------------------|--------------------------|--------------------------|--------------------------|-----------------|------------------|------------------|
| LSDA 01 | 23 | 47.6* | 6.9* | 0.012* | - | - |
| 2 | | 71.7 | 10.4 | 0.042 | - | - |
| 3 | | 32.4* | 4.7* | 0.022 | 1.52 | 0.22 |
| 4 | | 85.5 | 12.4 | 0.098 | 1.65 | 0.24 |
| 5 | | 68.9 | 10.0 | 0.056 | 1.65 | 0.24 |
| 6 | | 90.3 | 13.1 | 0.096 | 0.97* | 0.14* |
| 7 | | 66.2 | 9.6 | 0.055 | 1.45 | 0.21 |
| 8 | | 100.6* | 14.6* | 0.126* | 1.38 | 0.20 |
| average | | <u>76.5</u> | <u>11.1</u> | <u>0.062</u> | <u>1.52</u> | <u>0.22</u> |
| standard deviation | | 10.7 | 1.6 | 0.030 | 0.14 | 0.02 |
| LSDB 01 | 54 | 67.6 | 9.8 | 0.067 | 1.38 | 0.20 |
| 2 | | 86.9 | 12.6 | 0.143* | 1.24 | 0.18 |
| 3 | | 81.4 | 11.9 | 0.104 | 1.52 | 0.22 |
| 4 | | 31.7* | 4.6* | 0.024* | 1.31 | 0.19 |
| 5 | | 89.6 | 13.2 | 0.132 | 1.65 | 0.24 |
| 6 | | 56.0 | 8.1 | 0.010* | - | - |
| average | | <u>77.2</u> | <u>11.2</u> | <u>0.101</u> | <u>1.38</u> | <u>0.21</u> |
| standard deviation | | 14.5 | 2.1 | 0.033 | 0.14 | 0.02 |
| LSDC 01 | 82 | 44.1* | 6.4* | - | - | - |
| 2 | | 81.4 | 11.8 | 0.155 | 1.38 | 0.20 |
| 3 | | 82.0 | 11.9 | 0.152 | 0.90 | 0.13 |
| 4 | | 71.7 | 10.4 | 0.128 | 1.17 | 0.17 |
| 5 | | 62.7 | 9.1 | 0.048* | 0.69* | 0.10* |
| average | | <u>74.5</u> | <u>10.8</u> | <u>0.145</u> | <u>1.17</u> | <u>0.17</u> |
| standard deviation | | 8.9 | 1.3 | 0.015 | 0.21 | 0.03 |

* Not included in average

TABLE C1.2

INDIVIDUAL 2220-3 DRY SHEAR RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (MPa) | Ultimate Stress (ksi) | Ultimate Strain | Modulus (GPa) | Modulus (Msi) |
|--------------------|--------------------------|--------------------------|--------------------------|-----------------|------------------|------------------|
| LSDA 11 | 23 | 93.8* | 13.6* | 0.111* | 1.45 | 0.21 |
| 12 | | 91.0 | 13.2 | 0.107* | 1.45 | 0.21 |
| 13 | | 93.1* | 13.5* | 0.091 | 1.46 | 0.22 |
| 14 | | 57.4 | 8.3 | 0.053 | 1.31 | 0.19 |
| 15 | | 57.4 | 8.3 | 0.052 | 1.17 | 0.17 |
| 16 | | 37.2* | 5.4* | 0.020* | 1.86* | 0.27* |
| 17 | | 34.5* | 5.0* | 0.044 | 0.97* | 0.14* |
| average | | 68.5 | 9.9 | 0.065 | 1.38 | 0.20 |
| standard deviation | | 19.5 | 2.8 | 0.020 | 0.14 | 0.02 |
| LSDB 11 | 54 | 31.0* | 4.5* | 0.015* | 2.41* | 0.35* |
| 12 | | 79.5 | 11.5 | 0.099 | 1.17 | 0.17 |
| 14 | | 66.2 | 9.6 | 0.154* | - | - |
| 15 | | 75.8 | 11.0 | 0.115 | 1.17 | 0.17 |
| 16 | | 62.7 | 9.1 | 0.079 | 1.38 | 0.20 |
| average | | 71.0 | 10.3 | 0.098 | 1.24 | 0.18 |
| standard deviation | | 7.8 | 1.1 | 0.018 | 0.14 | 0.02 |
| LSDC 11 | 82 | 57.2 | 8.3 | 0.144 | 1.03 | 0.15 |
| 12 | | 70.3 | 10.2 | 0.168* | 1.17 | 0.17 |
| 13 | | 60.7 | 8.8 | 0.085* | 1.10 | 0.16 |
| 14 | | 70.3 | 10.2 | 0.142 | - | - |
| 15 | | 69.6 | 10.1 | 0.099 | 1.17 | 0.17 |
| average | | 66.2 | 9.6 | 0.128 | 1.10 | 0.16 |
| standard deviation | | 6.3 | 0.9 | 0.025 | 0.07 | 0.01 |

* Not included in average

TABLE C13

INDIVIDUAL 3502 WET SHEAR RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (MPa) | Ultimate Stress (ksi) | Ultimate Strain | Modulus (GPa) | Modulus (Msi) |
|--------------------|--------------------------|--------------------------|--------------------------|-----------------|------------------|------------------|
| LSWA 31 | 23 | 44.8 | 6.5 | 0.027 | 1.72 | 0.25 |
| 32 | | 45.5 | 6.6 | 0.026 | 1.79 | 0.26 |
| 33 | | 53.8 | 7.8 | 0.036 | 0.76* | 0.11* |
| 34 | | 55.2 | 8.0 | 0.043 | 1.52 | 0.22 |
| 35 | | 64.1* | 9.3* | 0.058* | 1.31 | 0.19 |
| average | | 49.8 | 7.2 | 0.033 | 1.58 | 0.23 |
| standard deviation | | 5.4 | 0.8 | 0.008 | 0.21 | 0.03 |
| LSWB 31 | 54 | 46.2 | 6.7 | 0.042 | 1.52 | 0.22 |
| 32 | | 58.6 | 8.5 | 0.052 | 1.03* | 0.15* |
| 33 | | 51.0 | 7.4 | 0.051 | 1.31 | 0.19 |
| 34 | | 57.9 | 8.4 | 0.046 | 1.31 | 0.19 |
| 35 | | 27.6* | 4.0* | 0.022* | 1.24 | 0.18 |
| 36 | | 57.2 | 8.3 | 0.047 | 1.52 | 0.22 |
| average | | 54.5 | 7.9 | 0.048 | 1.38 | 0.20 |
| standard deviation | | 5.4 | 0.8 | 0.004 | 0.14 | 0.02 |
| LSWC 31 | 82 | 59.3* | 8.6* | 0.076* | 1.03 | 0.15 |
| 32 | | 44.1 | 6.4 | 0.054 | 0.69 | 0.10 |
| 33 | | 49.6 | 7.2 | 0.052 | 1.03 | 0.15 |
| 34 | | 42.7 | 6.2 | 0.030 | 2.21* | 0.32* |
| 35 | | 17.2* | 2.5* | 0.015* | 1.24 | 0.18 |
| average | | 45.5 | 6.6 | 0.045 | 0.97 | 0.14 |
| standard deviation | | 3.7 | 0.5 | 0.013 | 0.21 | 0.03 |

* Not included in average

TABLE C 14

INDIVIDUAL FIBREDUX 914 WET SHEAR RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (MPa) | Ultimate Stress (ksi) | Ultimate Strain | Modulus (GPa) | Modulus (Msi) |
|--------------------|--------------------------|--------------------------|--------------------------|-----------------|------------------|------------------|
| LSWA 21 | 23 | 87.0 | 11.9 | 0.092 | 1.38 | 0.20 |
| 22 | | 28.3* | 4.1* | 0.014* | 1.72* | 0.25* |
| 23 | | 78.6 | 11.4 | 0.106* | 1.52 | 0.22 |
| 24 | | 59.3 | 8.6 | 0.044 | 1.38 | 0.20 |
| 25 | | <u>56.5</u> | <u>8.2</u> | <u>0.054</u> | <u>1.03*</u> | <u>0.15*</u> |
| average | | <u>68.9</u> | <u>10.0</u> | <u>0.063</u> | <u>1.45</u> | <u>0.21</u> |
| standard deviation | | 13.1 | 1.9 | 0.025 | 0.07 | 0.01 |
| LSWB 21 | 54 | 40.0 | 5.8 | 0.032 | 1.03 | 0.15 |
| 22 | | 45.5 | 6.6 | 0.041 | 1.31 | 0.19 |
| 23 | | 40.7 | 5.9 | 0.036 | 1.24 | 0.18 |
| 24 | | 44.8 | 6.5 | 0.031 | 1.93* | 0.28* |
| 25 | | <u>41.4</u> | <u>6.0</u> | <u>0.034</u> | <u>1.31</u> | <u>0.19</u> |
| average | | <u>42.7</u> | <u>6.2</u> | <u>0.035</u> | <u>1.24</u> | <u>0.18</u> |
| standard deviation | | 2.7 | 0.4 | 0.004 | 0.14 | 0.02 |
| LSWC 21 | 82 | 38.6 | 5.6 | 0.043 | 1.10 | 0.16 |
| 22 | | 44.8 | 6.5 | 0.064 | 1.17 | 0.17 |
| 23 | | 54.5* | 7.9* | 0.040 | 1.24 | 0.18 |
| 24 | | 33.1 | 4.8 | 0.032 | 1.24 | 0.18 |
| 25 | | <u>28.3*</u> | <u>4.1*</u> | <u>0.098*</u> | <u>1.10</u> | <u>0.16</u> |
| average | | <u>38.6</u> | <u>5.6</u> | <u>0.045</u> | <u>1.17</u> | <u>0.17</u> |
| standard deviation | | 5.9 | 0.9 | 0.010 | 0.70 | 0.01 |

* Not included in average

TABLE C15

INDIVIDUAL 2220-1 WET SHEAR RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (MPa) | Ultimate Stress (ksi) | Ultimate Strain | Modulus (GPa) | Modulus (Msi) |
|--------------------|--------------------------|--------------------------|--------------------------|-----------------|------------------|------------------|
| LSWA 01 | 23 | 66.9 | 9.7 | 0.070 | 1.38 | 0.20 |
| 2 | | 78.6 | 11.4 | 0.114 | 1.38 | 0.20 |
| 3 | | 60.0 | 8.7 | 0.052 | 1.72 | 0.25 |
| 4 | | 42.7 | 6.2* | 0.032 | 1.59 | 0.23 |
| 5 | | 68.3 | 9.9 | - | - | - |
| 6 | | 84.8* | 12.3* | 0.174* | 1.52 | 0.22 |
| average | | 68.2 | 9.9 | 0.067 | 1.52 | 0.22 |
| standard deviation | | 7.6 | 1.1 | 0.035 | 0.14 | 0.02 |
| LSWB 01 | 54 | 68.3 | 9.9 | 0.021* | 1.24 | 0.18 |
| 2 | | 67.6 | 9.8 | 0.155 | 1.10* | 0.16* |
| 3 | | 69.6 | 10.1 | 0.152 | 1.38 | 0.20 |
| 4 | | 64.8 | 9.4 | 0.128 | 1.24 | 0.20 |
| 5 | | 59.3 | 8.6 | 0.049 | 1.24 | 0.18 |
| average | | 66.2 | 9.6 | 0.121 | 1.31 | 0.19 |
| standard deviation | | 4.3 | 0.6 | 0.049 | 0.07 | 0.01 |
| LSWC 01 | 82 | 56.5 | 8.2 | 0.062 | 1.10 | 0.16 |
| 2 | | 51.7 | 7.5 | 0.090 | 1.45 | 0.21 |
| 3 | | 57.9 | 8.4 | 0.094 | 1.38 | 0.20 |
| 4 | | 53.8 | 7.8 | 0.105 | 1.03 | 0.15 |
| 5 | | 54.5 | 7.9 | 0.089 | 1.10 | 0.16 |
| average | | 55.2 | 8.0 | 0.095 | 1.24 | 0.18 |
| standard deviation | | 2.5 | 0.4 | 0.007 | 0.21 | 0.03 |

* Not included in average

TABLE C16

INDIVIDUAL 2220-3 WET SHEAR RESULTS

| Specimen No. | Test Temperature (°C) | Ultimate Stress (MPa) | Ultimate Stress (ksi) | Ultimate Strain | Modulus (GPa) | Modulus (Msi) |
|--------------------|--------------------------|--------------------------|--------------------------|-----------------|------------------|------------------|
| LSWA 11 | 23 | 71.0 | 10.3 | 0.089* | 1.65 | 0.24 |
| 12 | | 71.0 | 10.3 | 0.123 | 1.45 | 0.21 |
| 13 | | 78.6 | 11.4 | 0.141 | 1.38 | 0.20 |
| 14 | | 81.4 | 11.8 | 0.170* | 0.69* | 0.10* |
| 15 | | 79.3 | 11.5 | 0.130 | 0.76* | 0.11* |
| average | | 75.8 | 11.1 | 0.131 | 1.17 | 0.22 |
| standard deviation | | 4.7 | 0.7 | 0.009 | 0.38 | 0.02 |
| LSWB 11 | 54 | 63.4 | 9.2 | 0.120 | 1.17 | 0.17 |
| 12 | | 65.5 | 9.5 | 0.159 | 1.38 | 0.20 |
| 13 | | 65.5 | 9.5 | 0.140 | 1.24 | 0.18 |
| 14 | | 53.8 | 7.8 | 0.059* | 1.38 | 0.20 |
| 15 | | 66.9 | 9.7 | 0.158 | 1.10 | 0.16 |
| average | | 62.7 | 9.1 | 0.144 | 1.24 | 0.18 |
| standard deviation | | 5.4 | 0.8 | 0.018 | 0.14 | 0.02 |
| LSWC 11 | 82 | 53.8 | 7.8 | 0.055* | 1.38 | 0.20 |
| 12 | | 49.0 | 7.1 | 0.098 | 0.69* | 0.10* |
| 13 | | 46.2 | 6.7 | 0.124 | 1.31 | 0.19 |
| 14 | | 49.0 | 7.1 | 0.137 | 1.31 | 0.19 |
| 15 | | 49.0 | 7.1 | 0.153 | 1.31 | 0.19 |
| average | | 49.6 | 7.2 | 0.128 | 1.31 | 0.19 |
| standard deviation | | 2.7 | 0.4 | 0.023 | 0.03 | 0.01 |

* Not included in average

TABLE C17
COEFFICIENTS OF THERMAL EXPANSION OF THE
VARIOUS EPOXY MATRIX MATERIALS

| Resin System | Temp (°C) | Dry Condition (10 ⁻⁶ in/in-°C) | Moisture-Saturated Condition (10 ⁻⁶ in/in-°C) |
|--------------|-----------|--|---|
| 3502 | 23 | 37.4 | 54.6 |
| | 54 | 44.1 | 61.3 |
| | 82 | 54.2 | 58.8 |
| | 93 | 59.3 | 55.6 |
| 914 | 23 | 65.9 | 58.0 |
| | 54 | 77.5 | 57.8 |
| | 82 | 96.8 | 58.7 |
| | 93 | 106.7 | 59.3 |
| 2220-1 | 23 | 59.4 | 59.9 |
| | 54 | 56.3 | 64.0 |
| | 82 | 59.2 | 55.1 |
| | 93 | 61.8 | 48.3 |
| 2220-3 | 23 | 48.2 | 61.2 |
| | 54 | 55.5 | 68.3 |
| | 82 | 56.8 | 66.5 |
| | 93 | 56.0 | 63.6 |

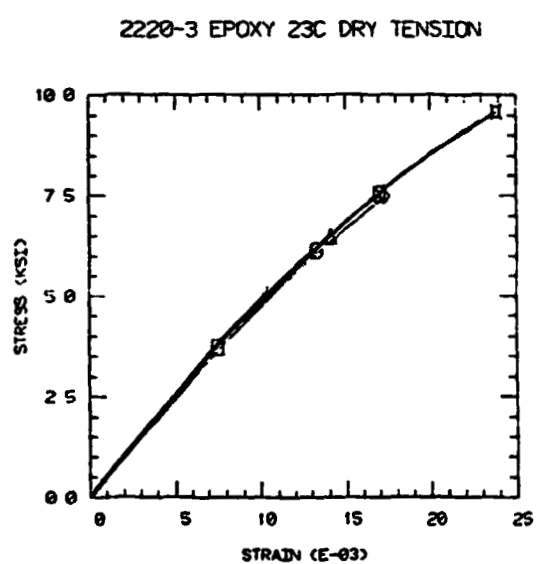
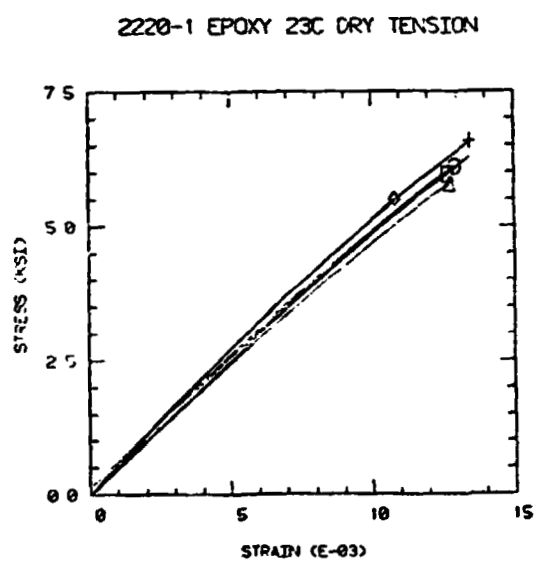
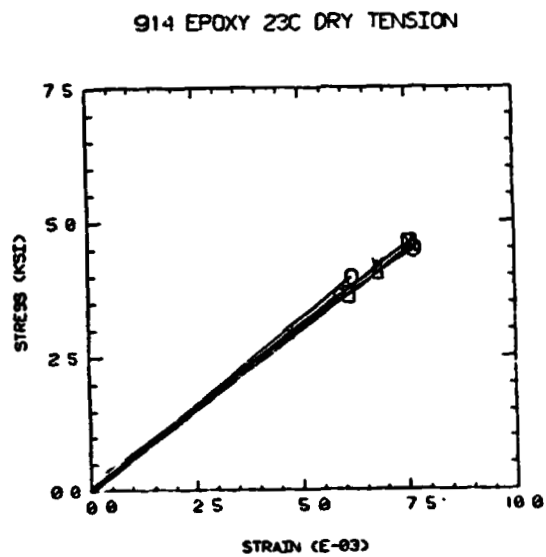
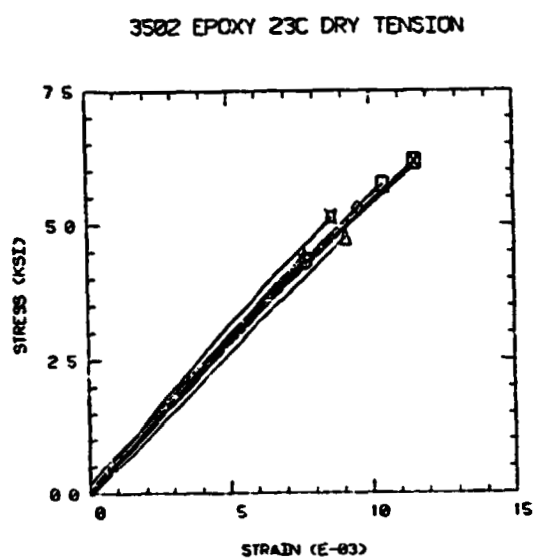


Figure C1. Tensile Stress-Strain Curves, 23°C, Dry.

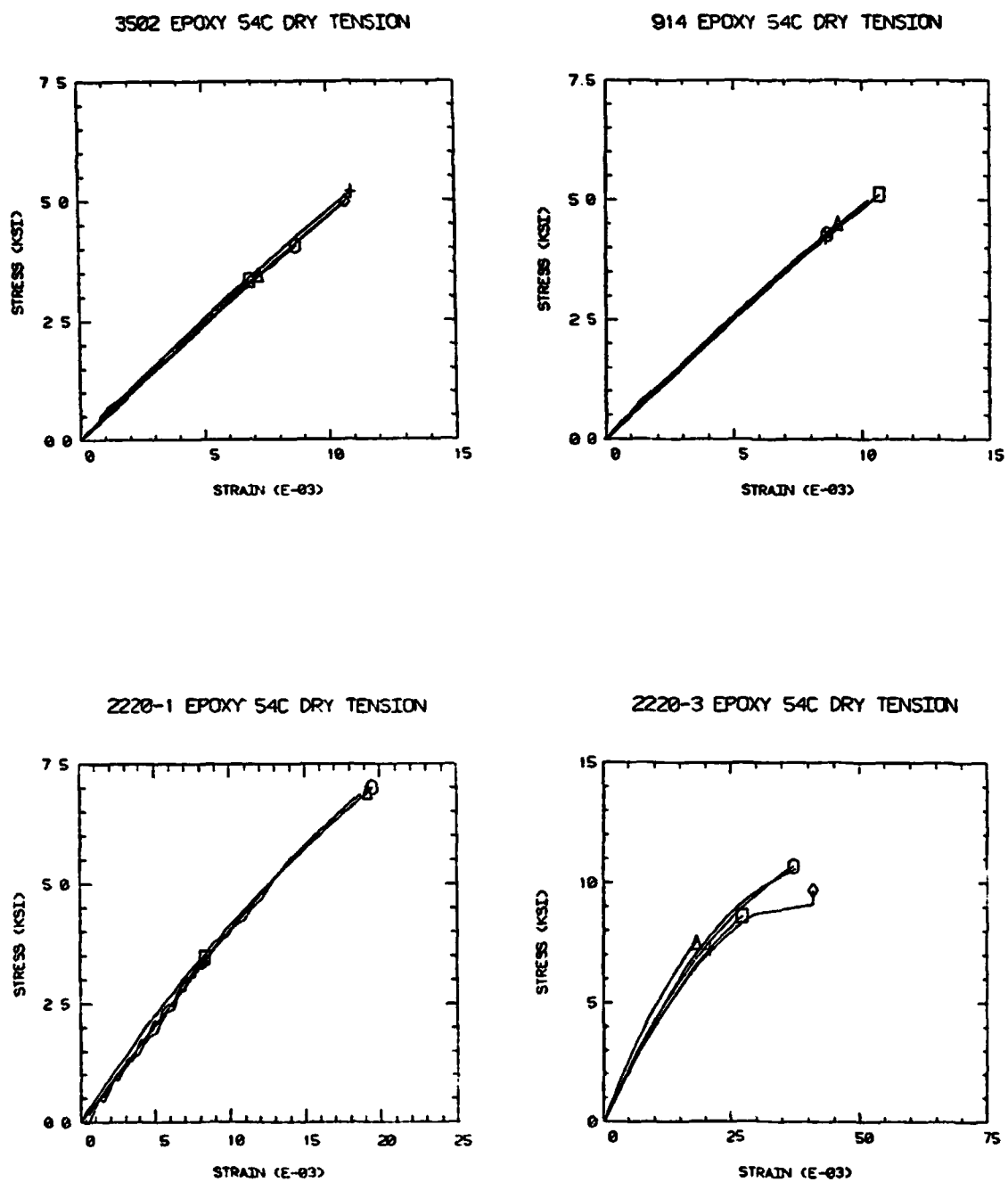


Figure C2. Tensile Stress-Strain Curves, 54°C, Dry.

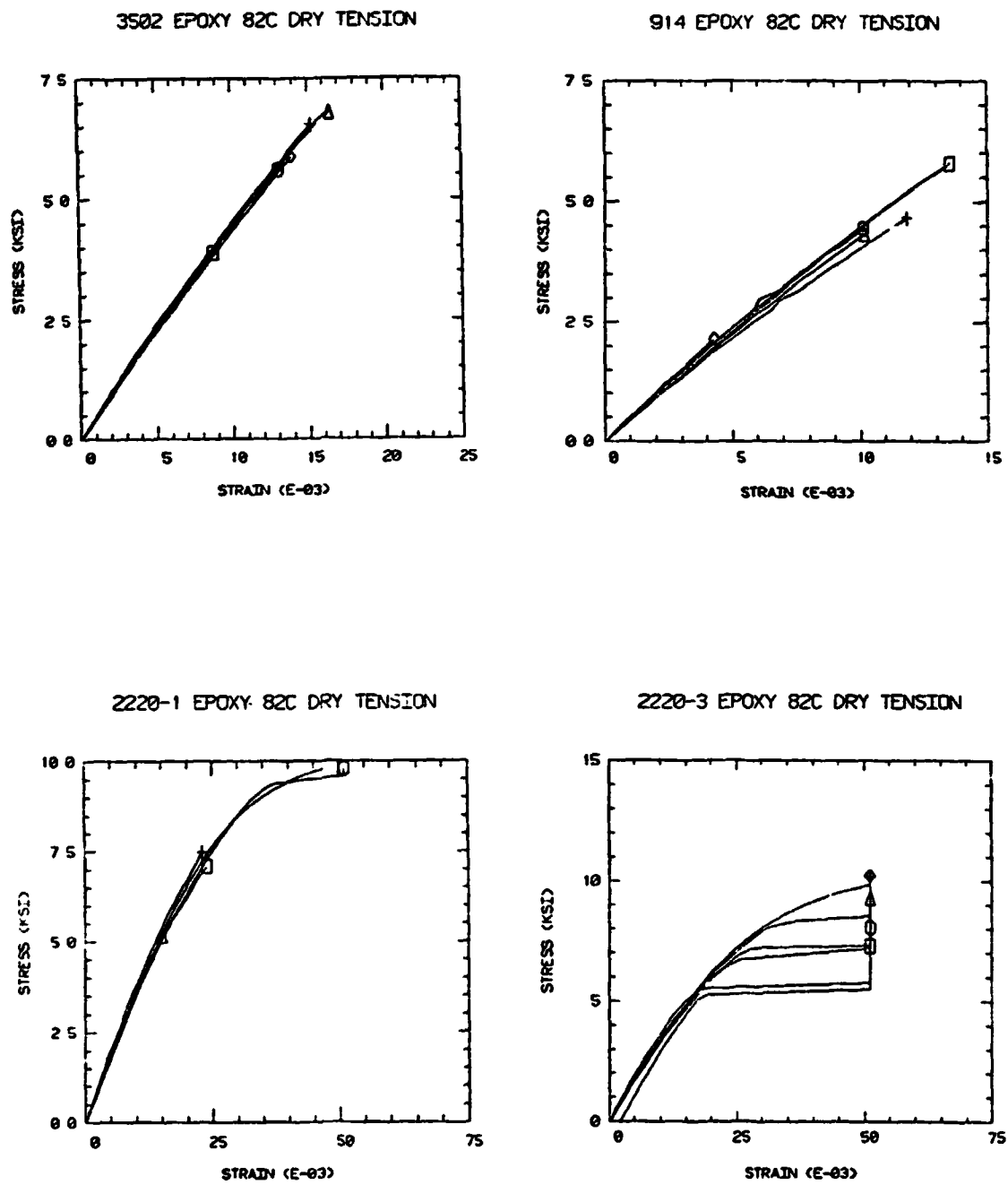


Figure C3. Tensile Stress-Strain Curves, 82°C, Dry.

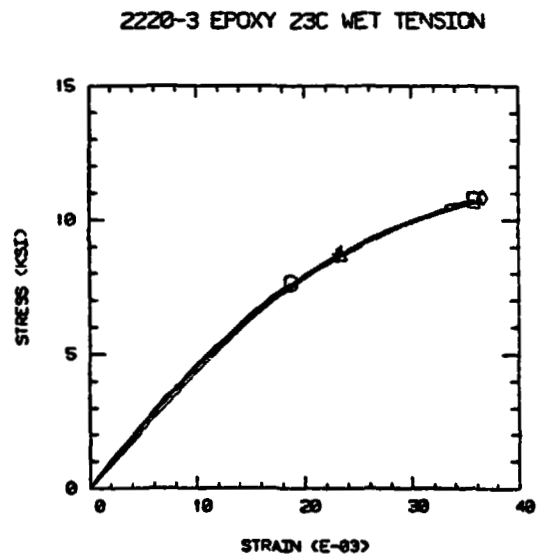
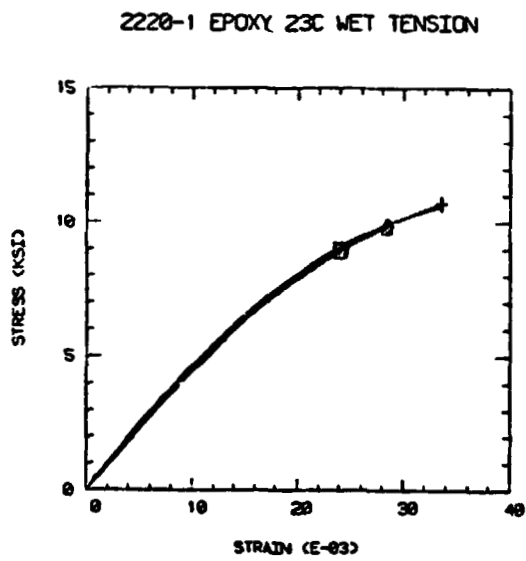
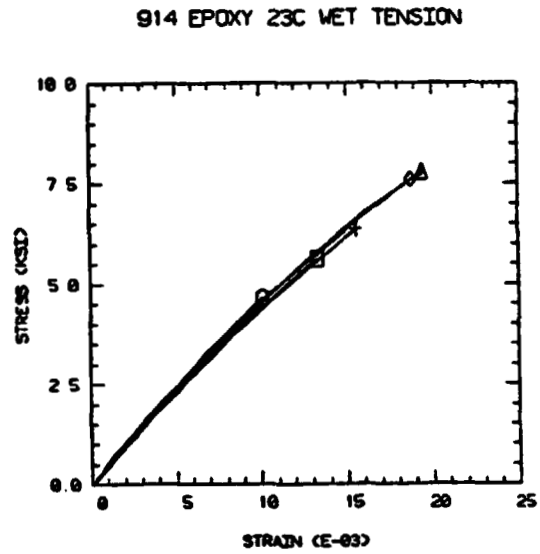
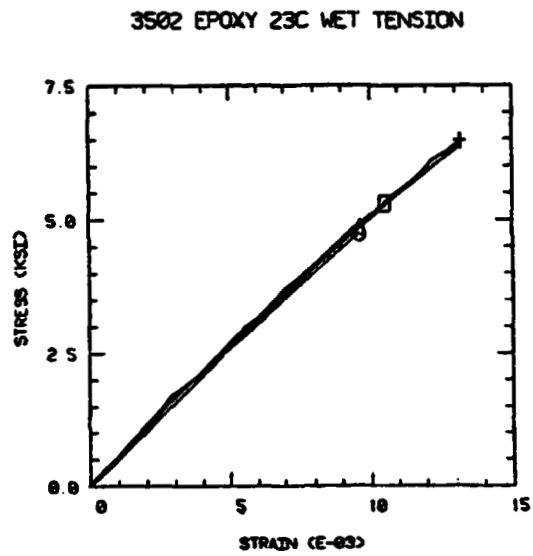


Figure C4. Tensile Stress-Strain Curves, 23°C, Wet.

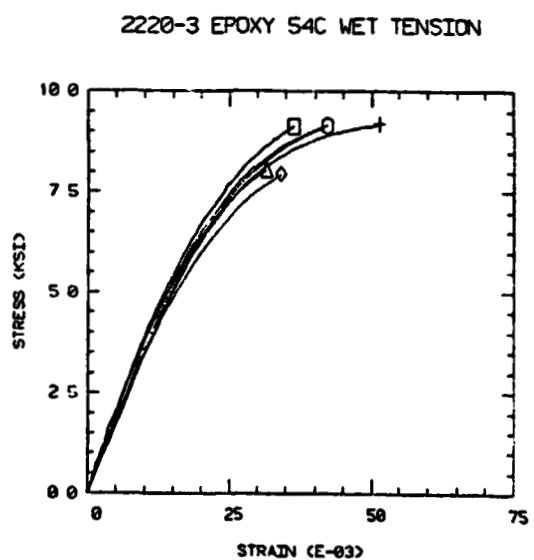
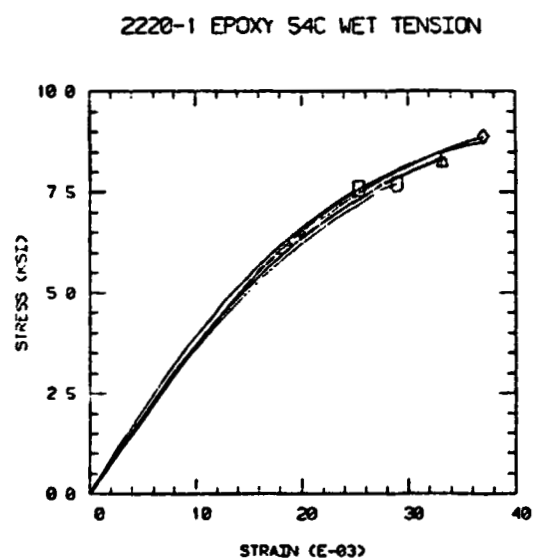
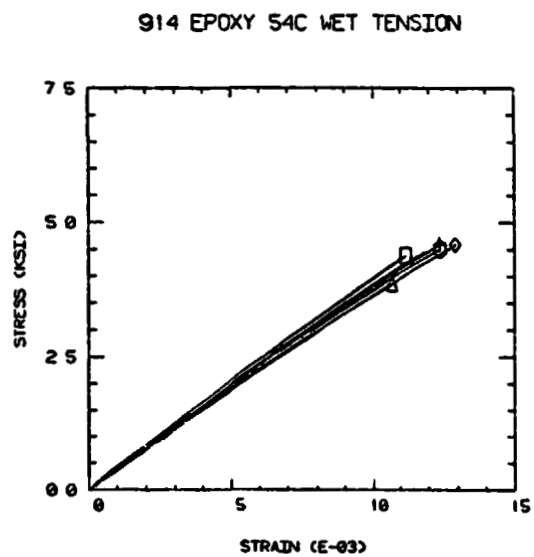
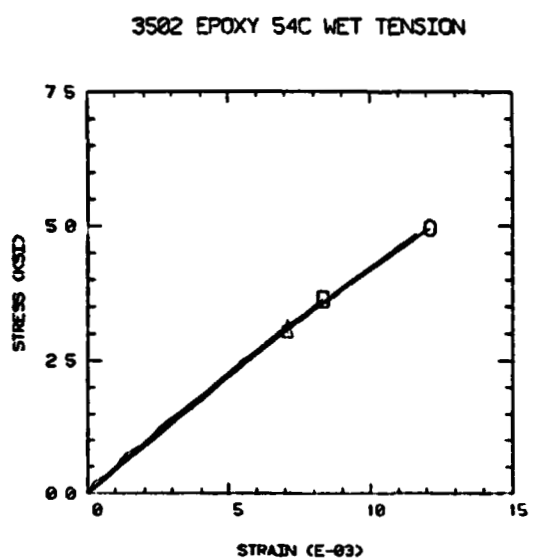


Figure C5. Tensile Stress-Strain Curves, 54°C, Wet.

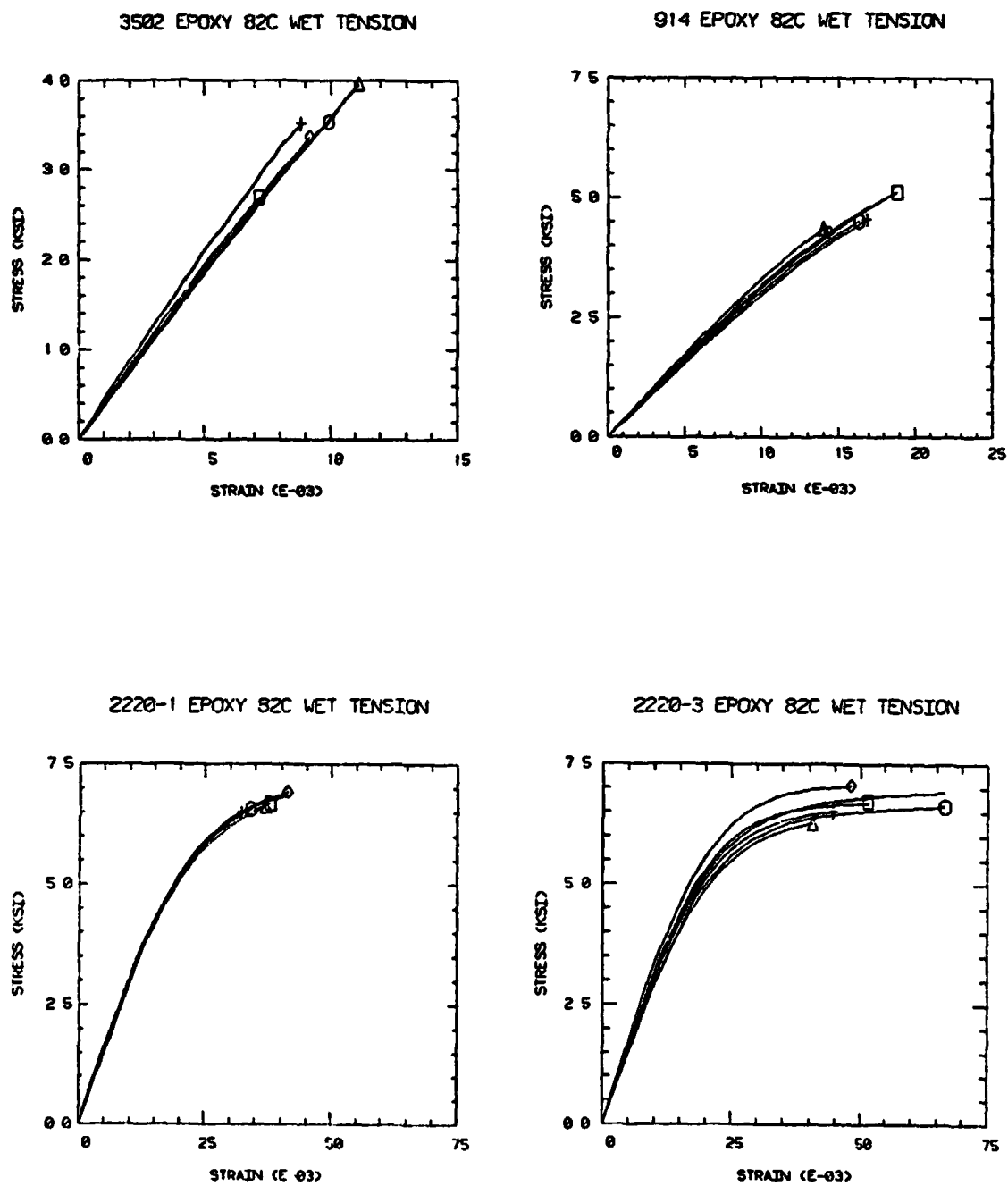


Figure C6. Tensile Stress-Strain Curves, 82°C, Wet.

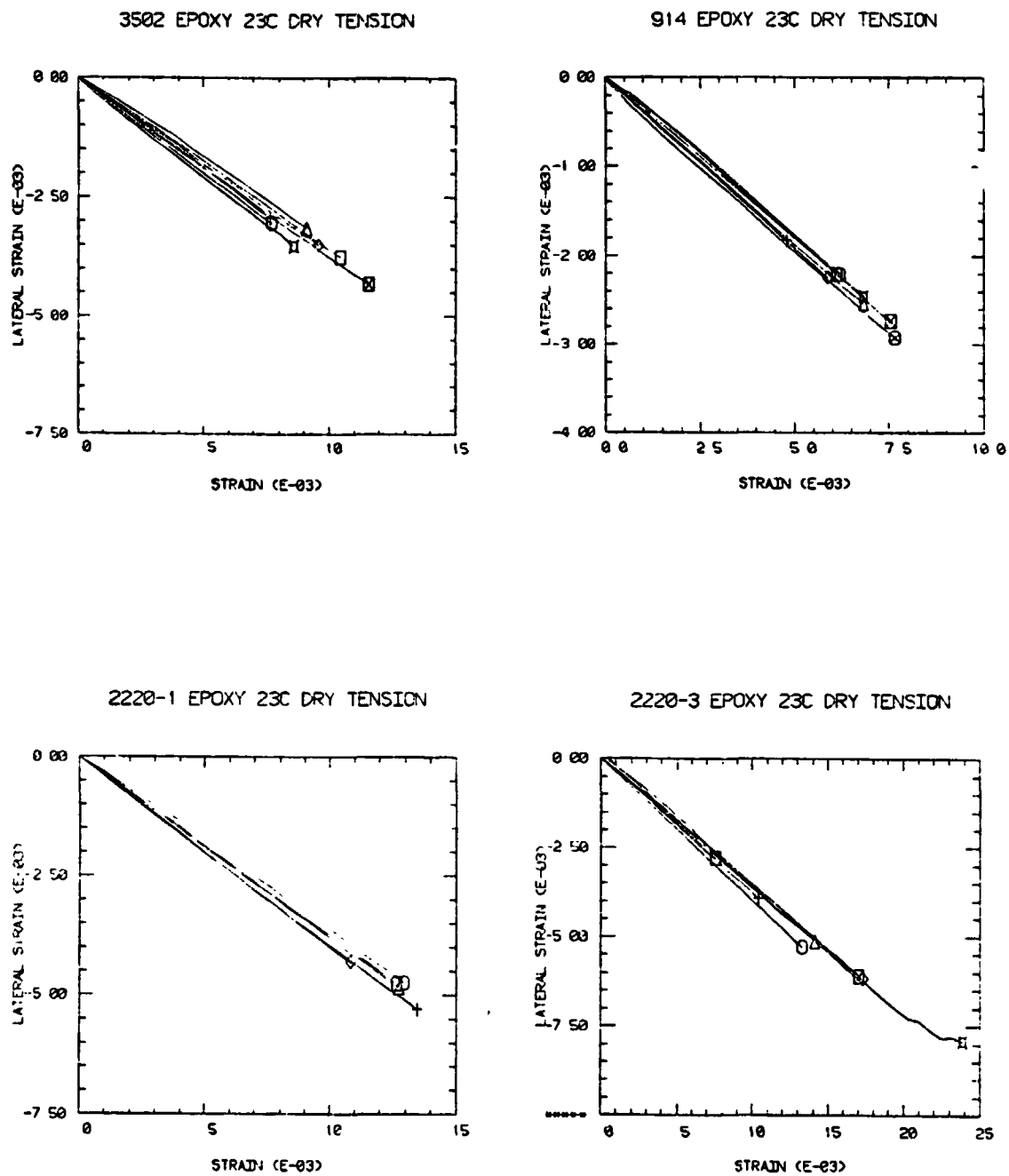


Figure C7. Poisson's Ratio Curves, 23°C, Dry.

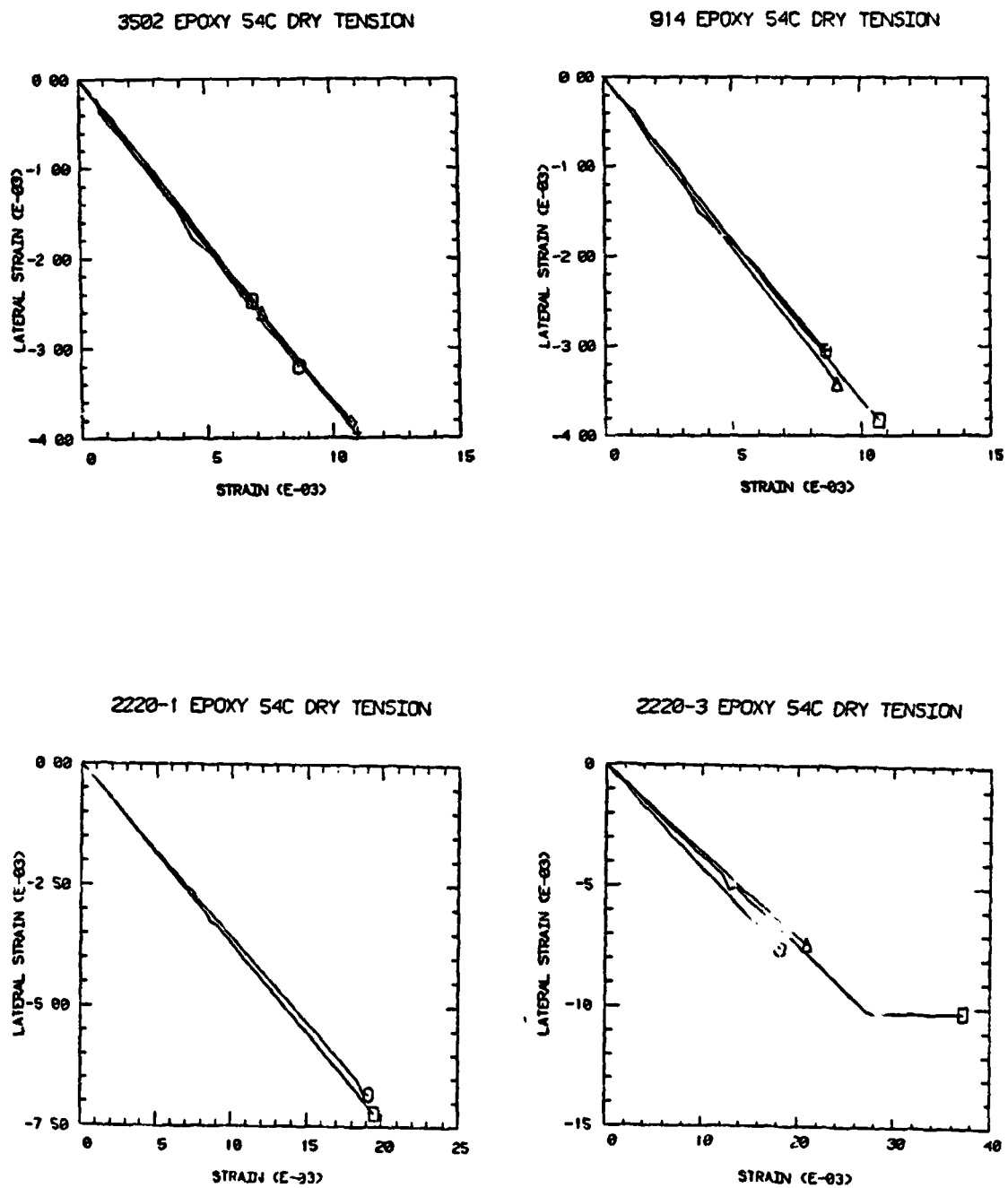


Figure C8. Poisson's Ratio Curves, 54°C, Dry.

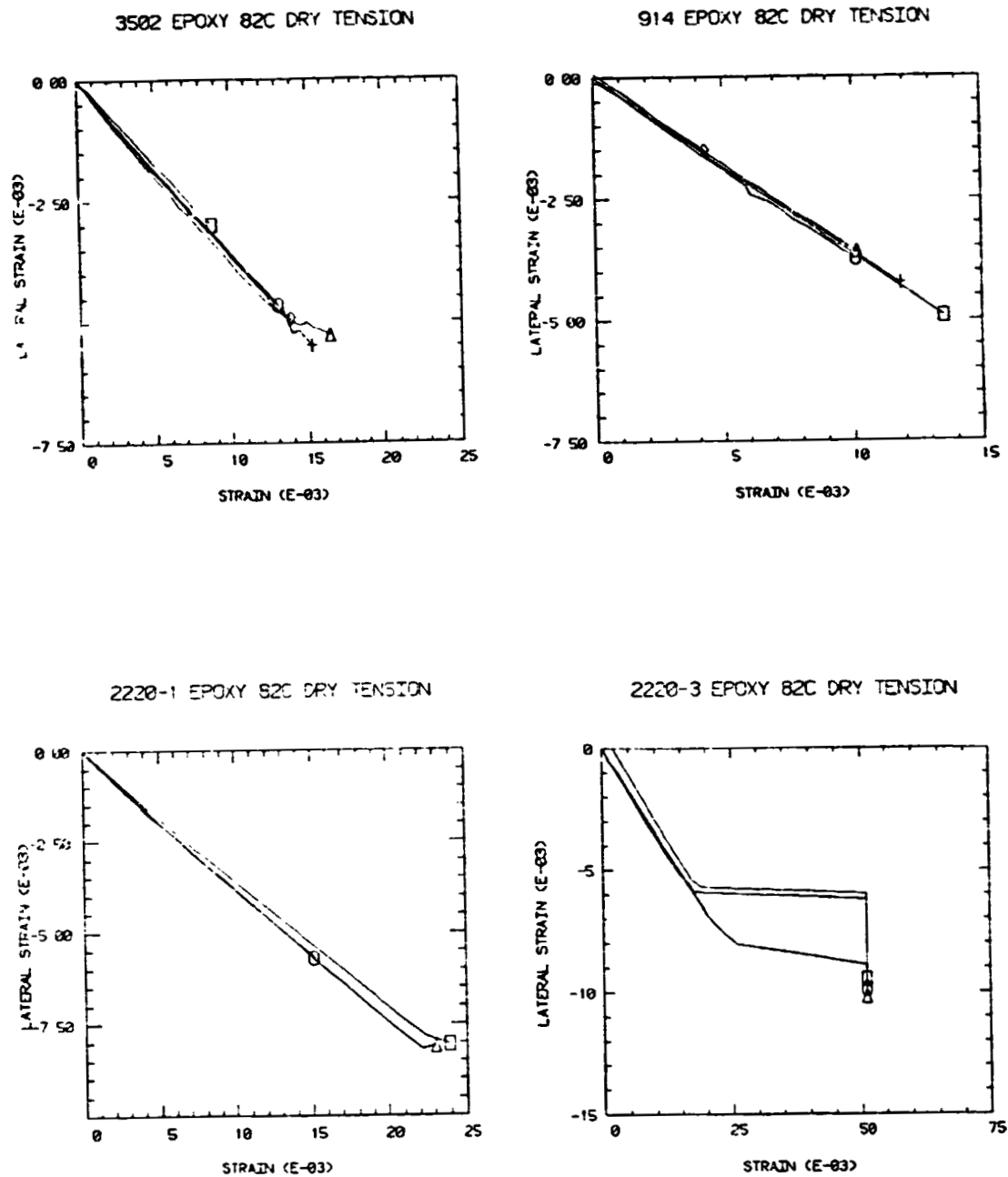


Figure C9. Poisson's Ratio Curves, 82°C, Dry.

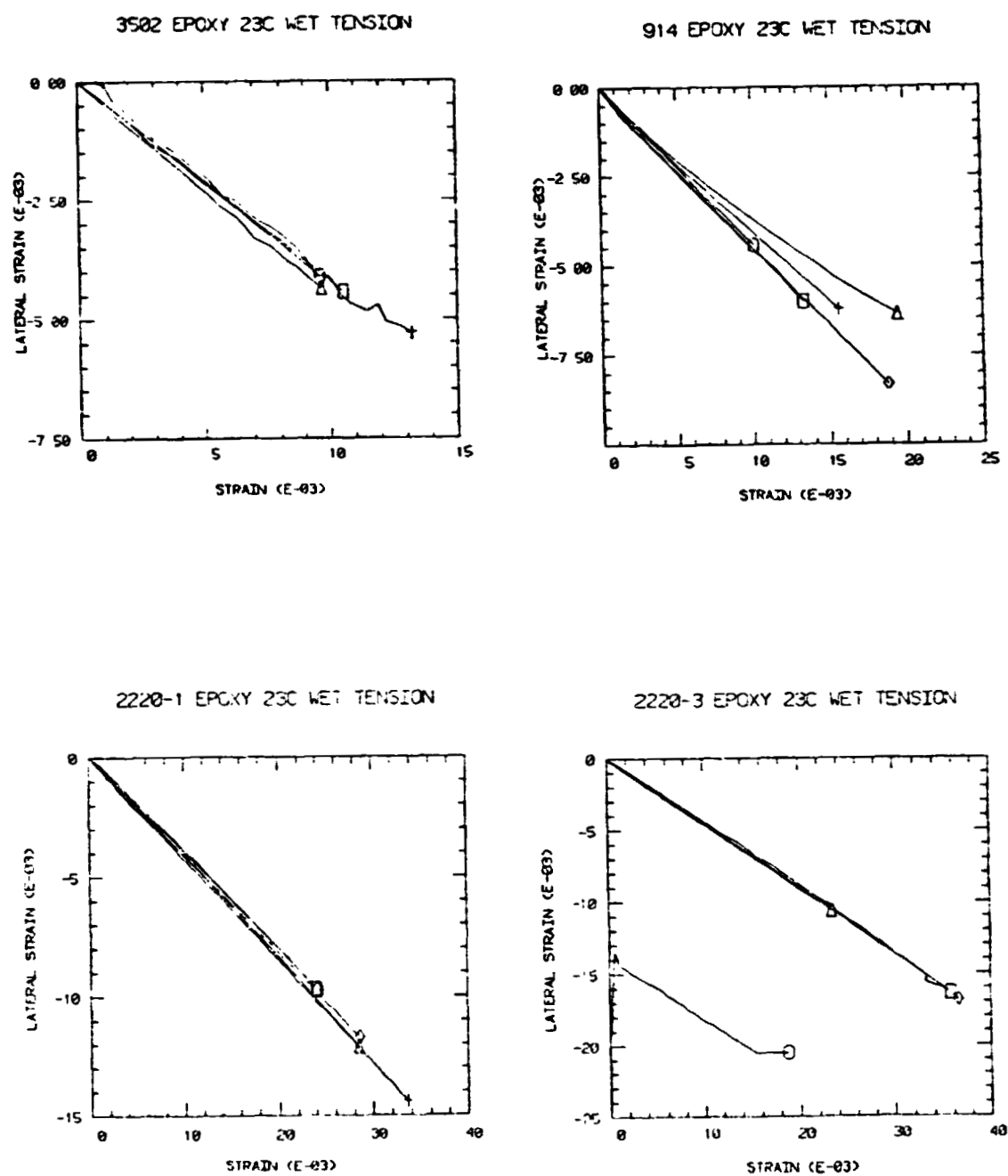


Figure C10. Poisson's Ratio Curves, 23°C, Wet.

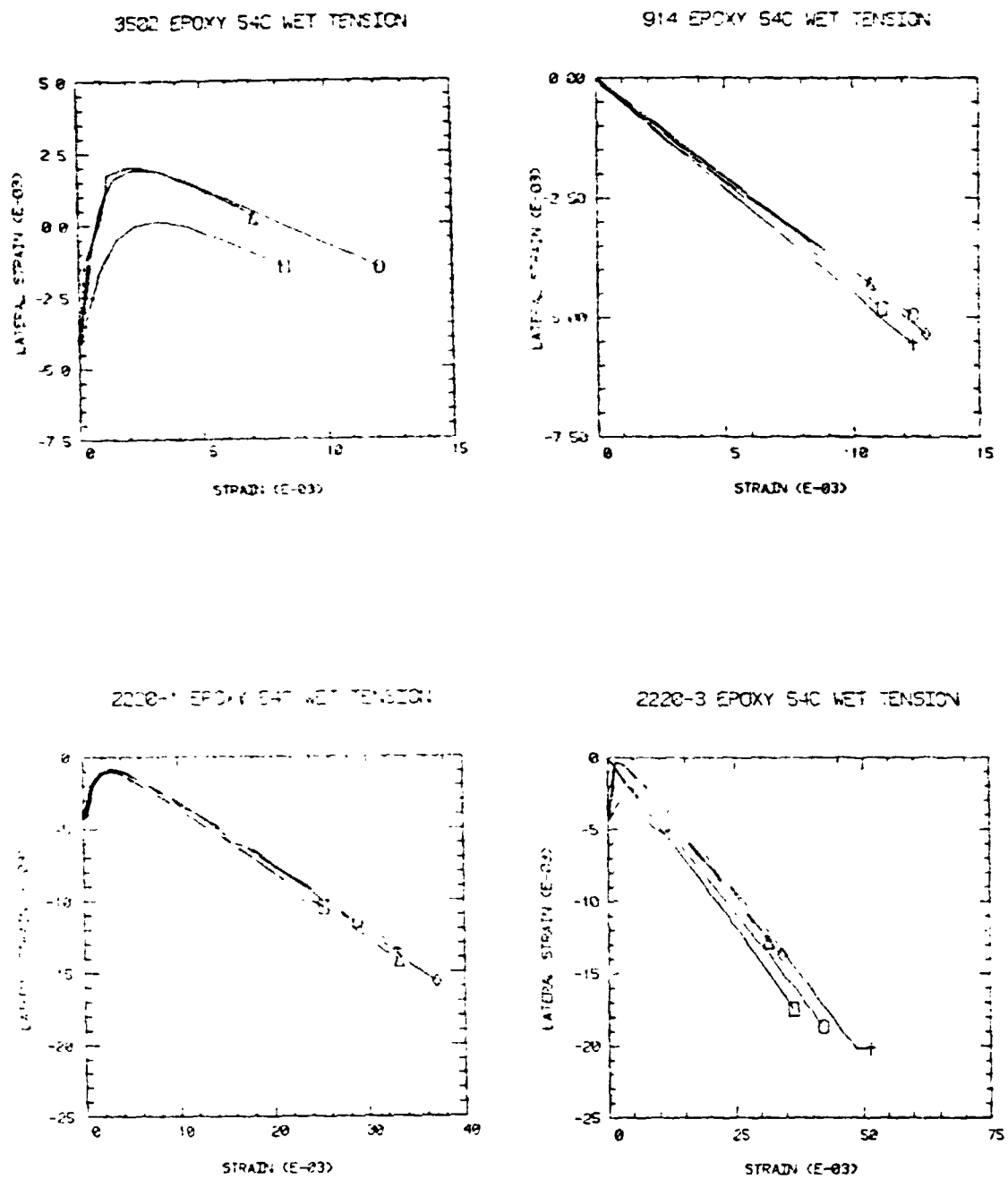


Figure C11. Poisson's Ratio Curves, 54°C, Wet.

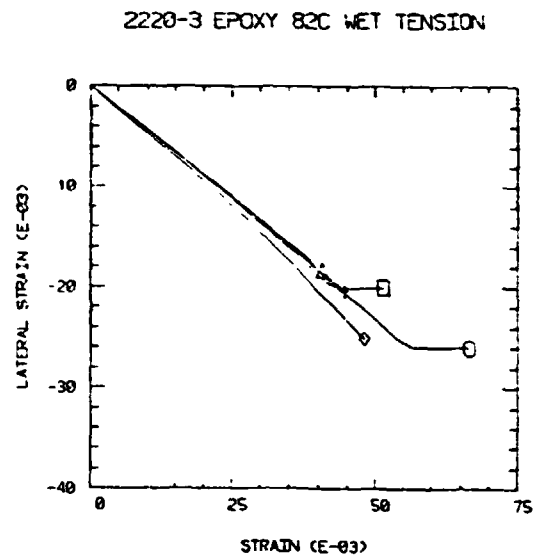
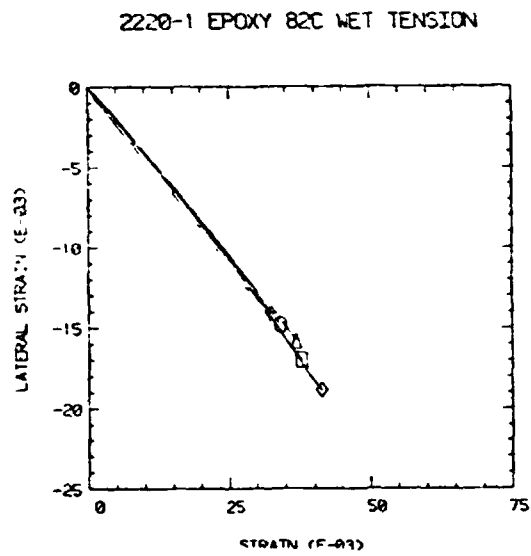
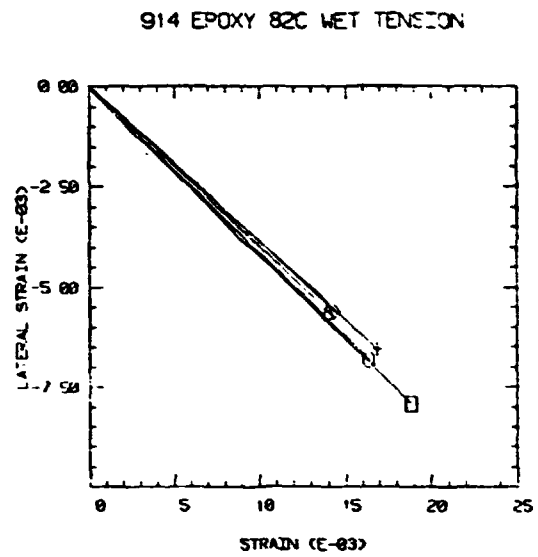
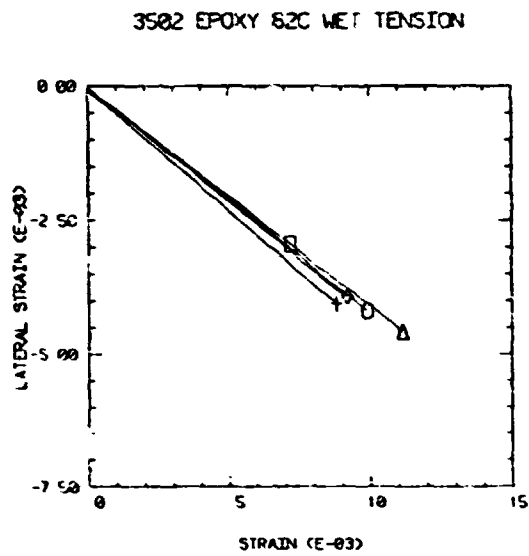


Figure C12. Poisson's Ratio Curves, 82°C, Wet.

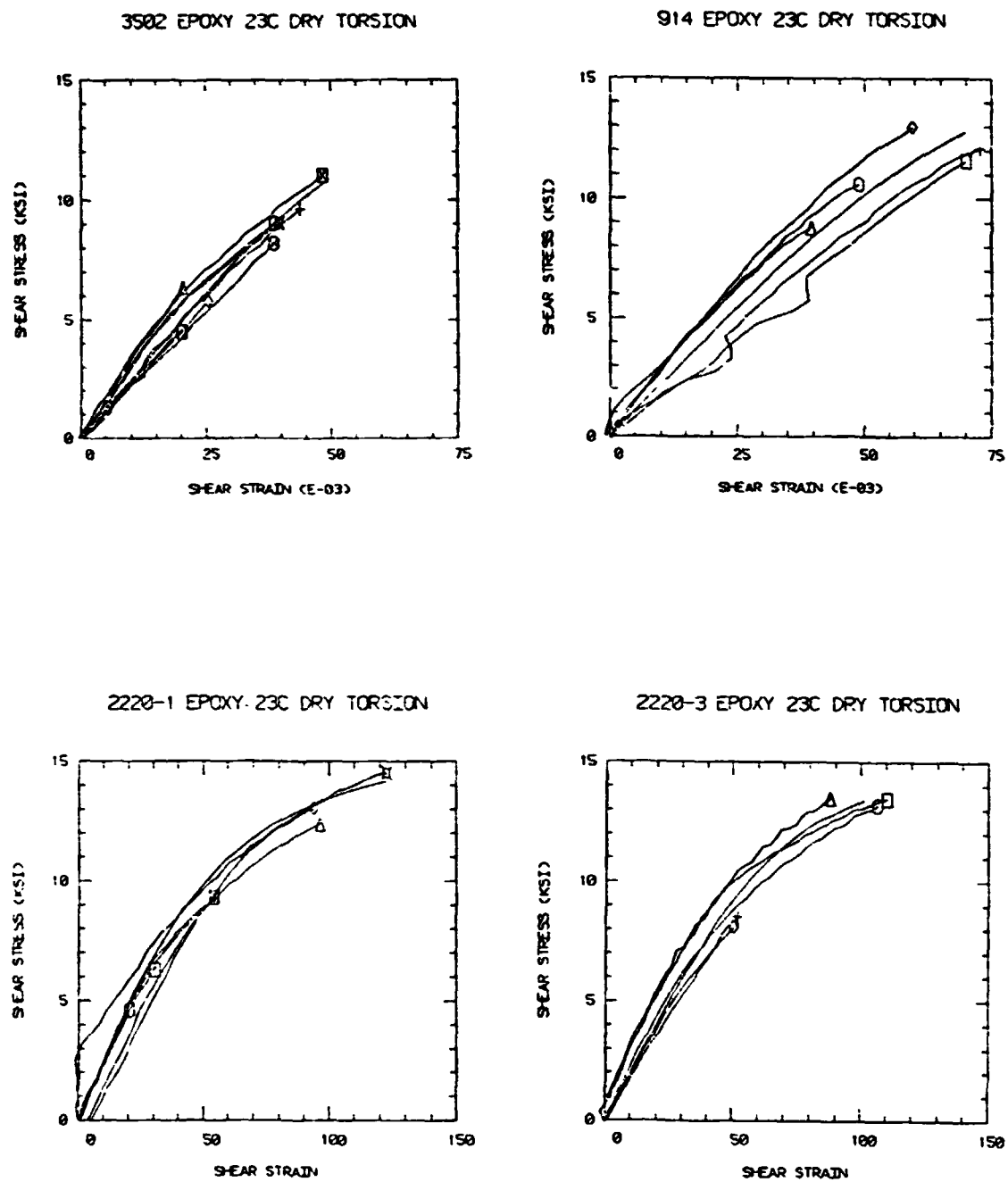


Figure C13. Shear Stress Strain Curves, 23°C, Dry.

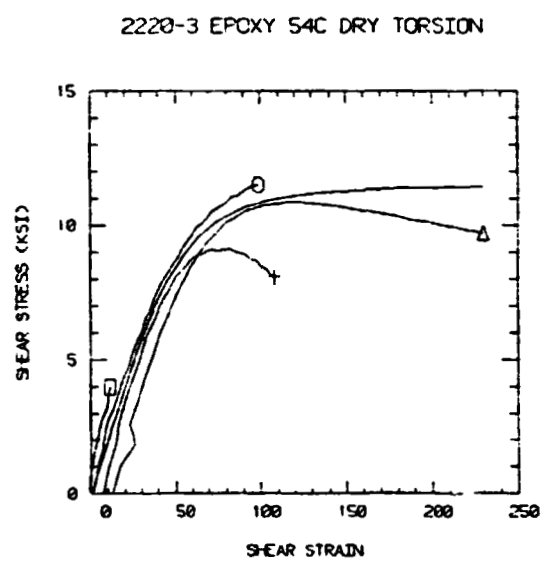
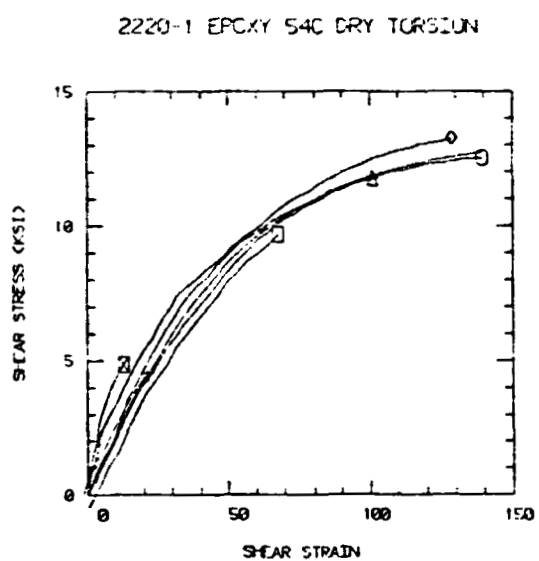
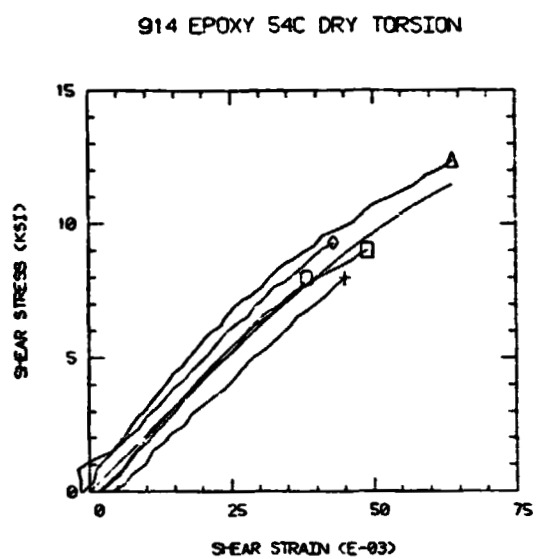
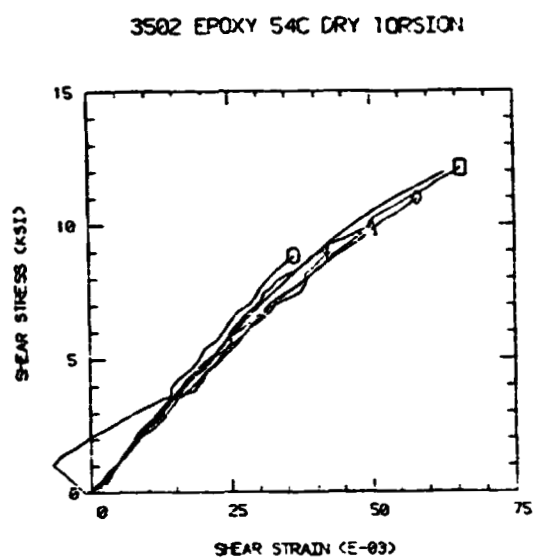


Figure C14. Shear Stress Strain Curves, 54°C, Dry.

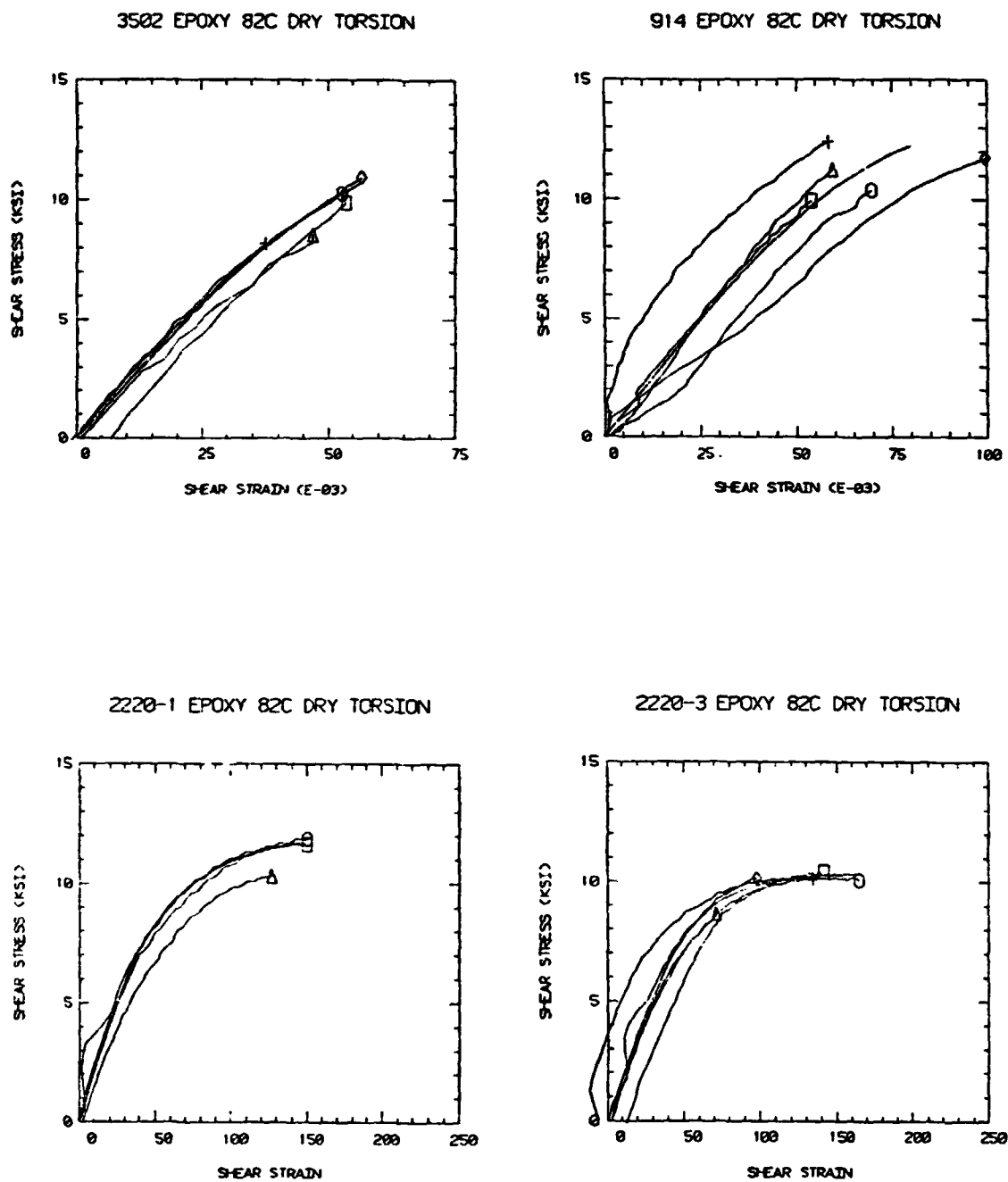


Figure C15. Shear Stress Strain Curves, 82°C, Dry.

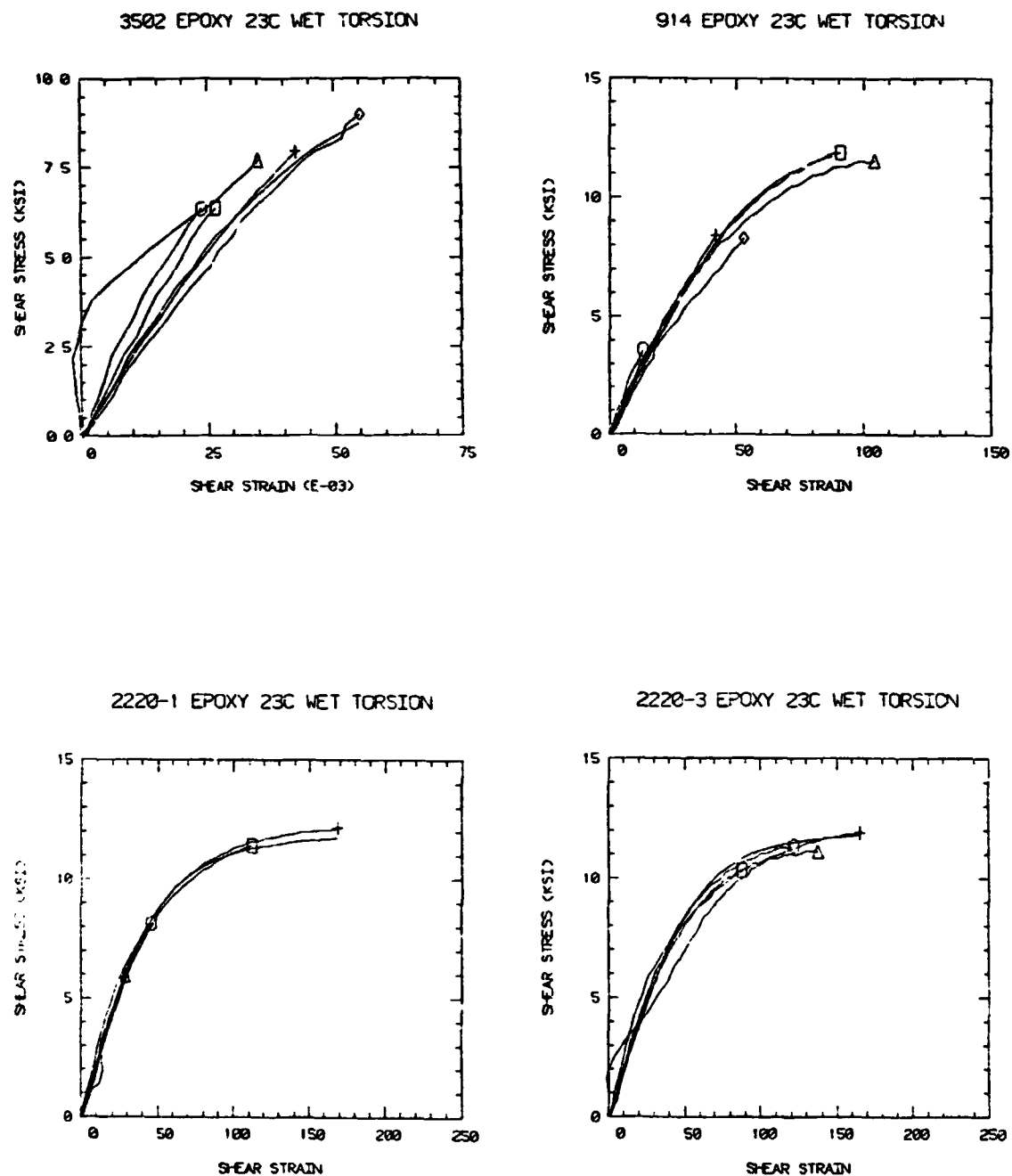


Figure C16. Shear Stress Strain Curves, 23°C, Wet.

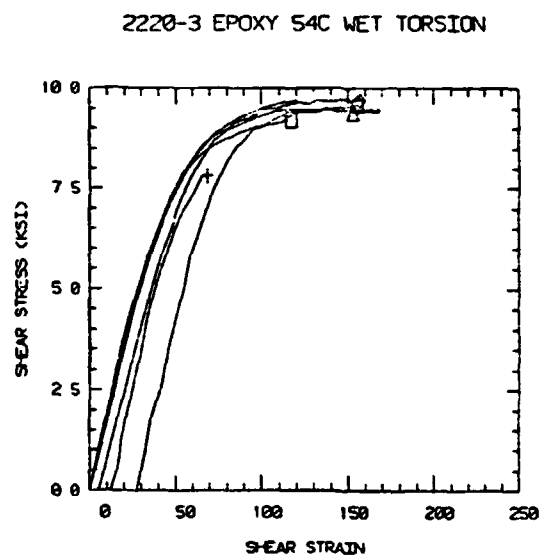
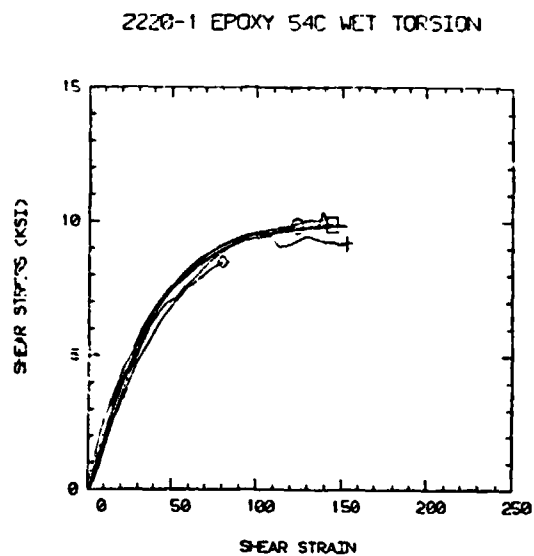
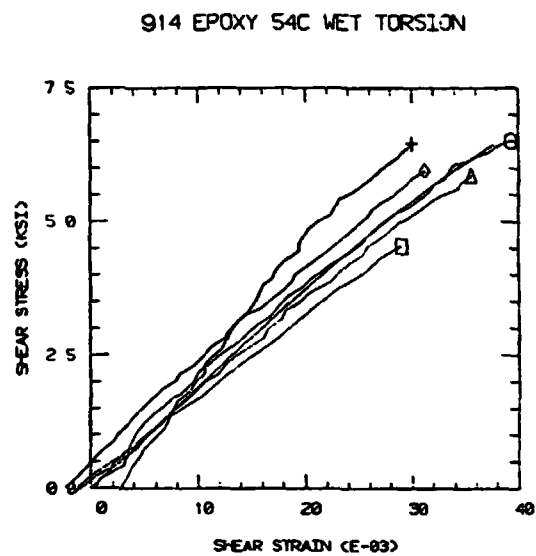
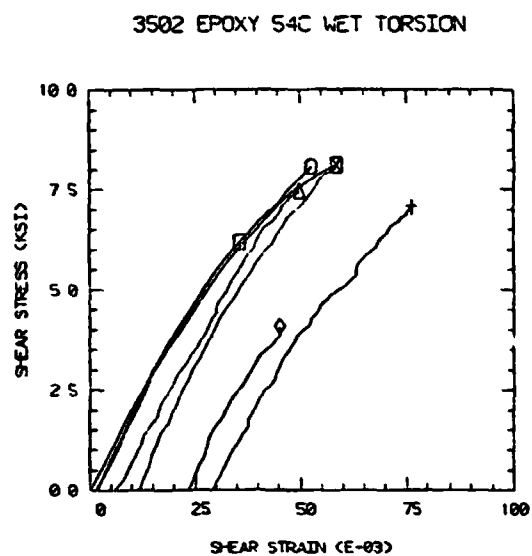


Figure C17. Shear Stress Strain Curves, 54°C, Wet.

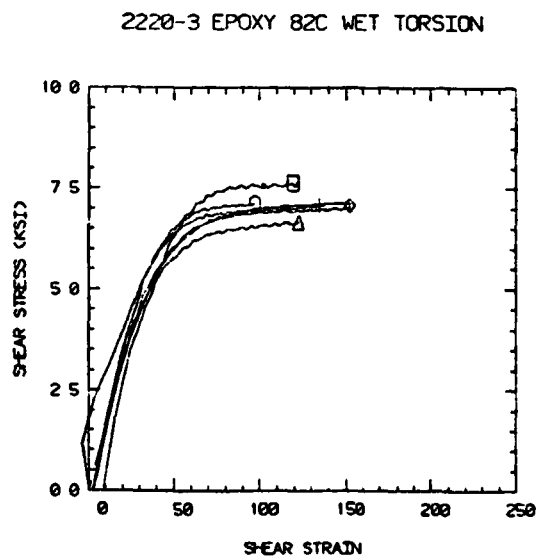
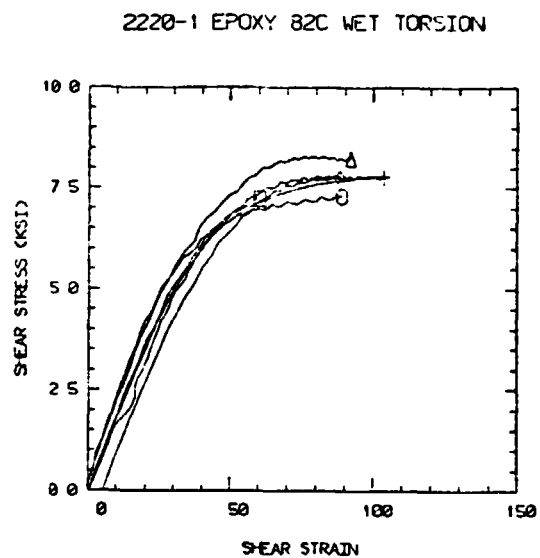
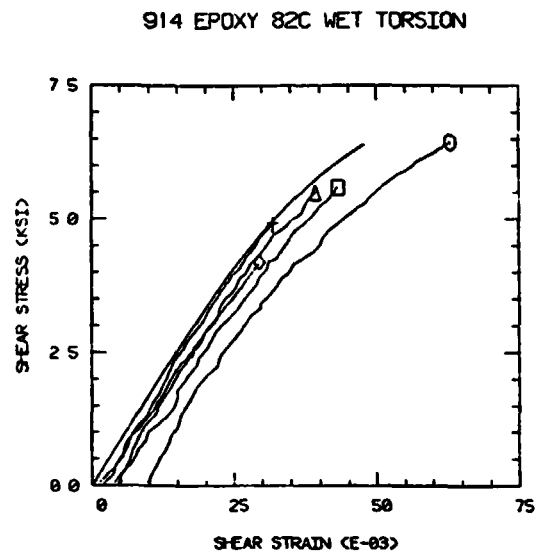
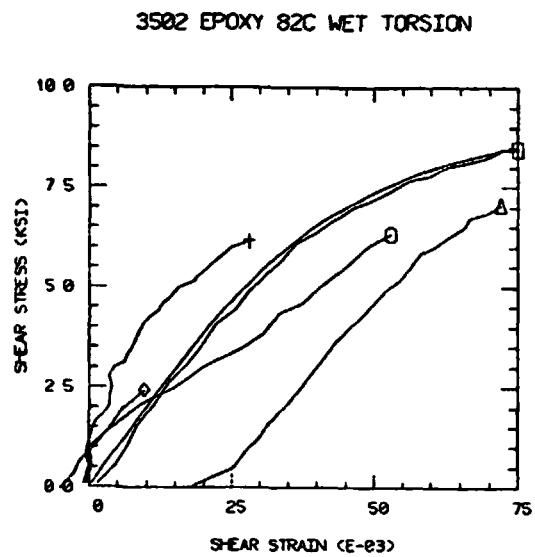


Figure C18. Shear Stress Strain Curves, 82°C, Wet.

APPENDIX D

ADDITIONAL SEM FRACTURE SURFACE PHOTOGRAPHS

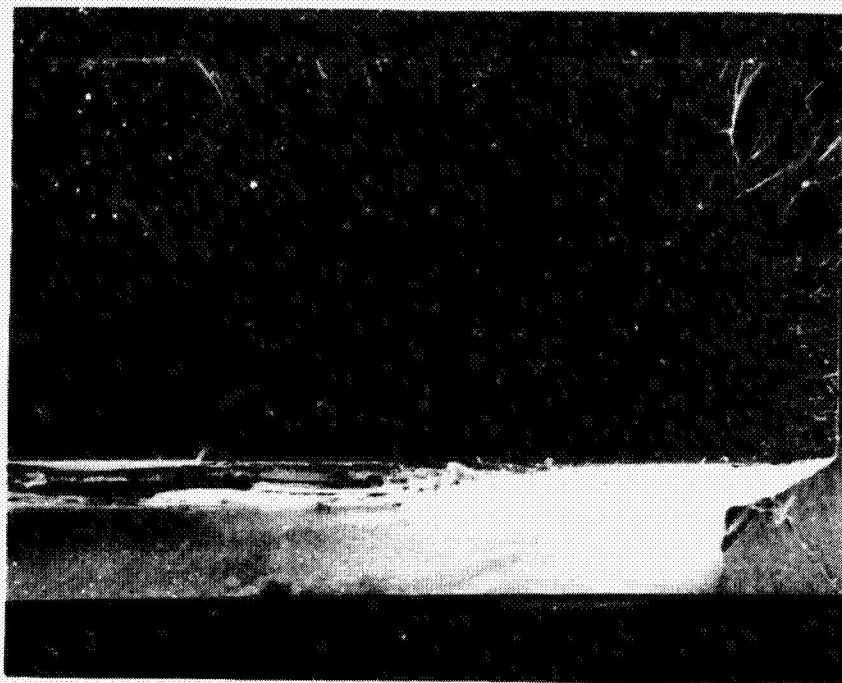


Figure D1. 3502 Neat Epoxy Tension, Specimen No. LTDA35, 23°C, Dry.

A large smooth region surrounded by a minimal coarse area is shown in this photograph. The triangular chip region typically observed in the gross failure observations is off the picture to the left. No explicit initiation site was discernible on the surface but is assumed to have been somewhere in the smooth region.

PRECEDING PAGE BLANK NOT FILMED

ORIGINAL PHOTOGRAPH
OF POOR QUALITY



Figure D2. 3502 Neat Epoxy Tension, Specimen No. LTDB34, 54°C, Dry.

A smaller smooth region is typical for this test condition than at room temperature (23°C), dry or wet. The rough area is similar in appearance, however.

ORIGINAL
OF POOR QUALITY

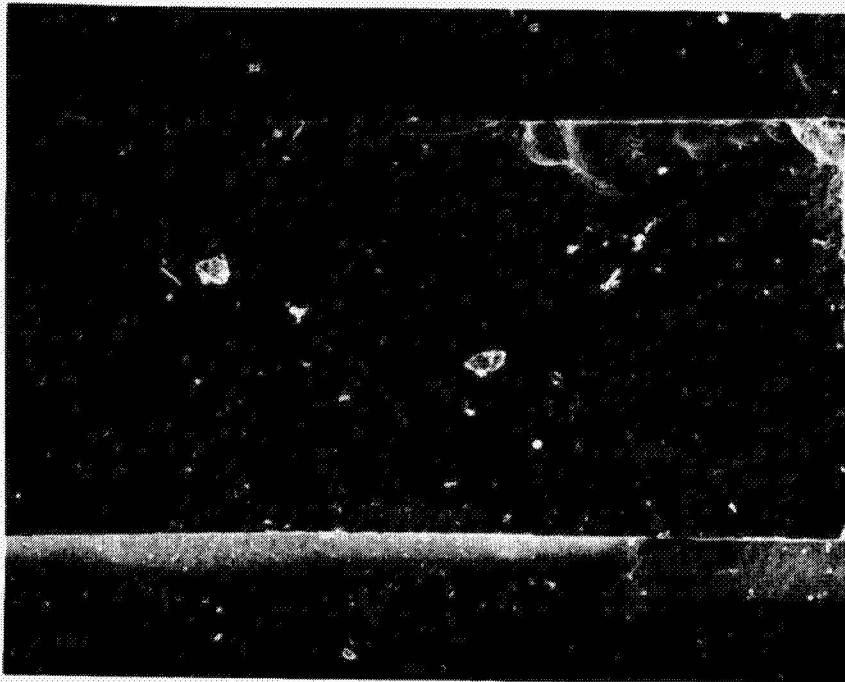


Figure D3. 3502 Neat Epoxy Tension, Specimen No. LTWB34, 54°C, Moisture-Saturated.

This failure is quite unique showing a very smooth glassy appearance over the entire surface. A relatively low strength was recorded for this specimen. The small clamshell marks at the top right edge of the specimen could possibly be where the failure initiated.

ORIGINAL
OF POOR QUALITY.



Figure D4. 3502 Neat Epoxy Tension, Specimen No. LTWB34, 54°C,
Moisture-Saturated.

This is a higher magnification view of a local region of Figure D3
showing the clamshell marks.

OF POOR QUALITY

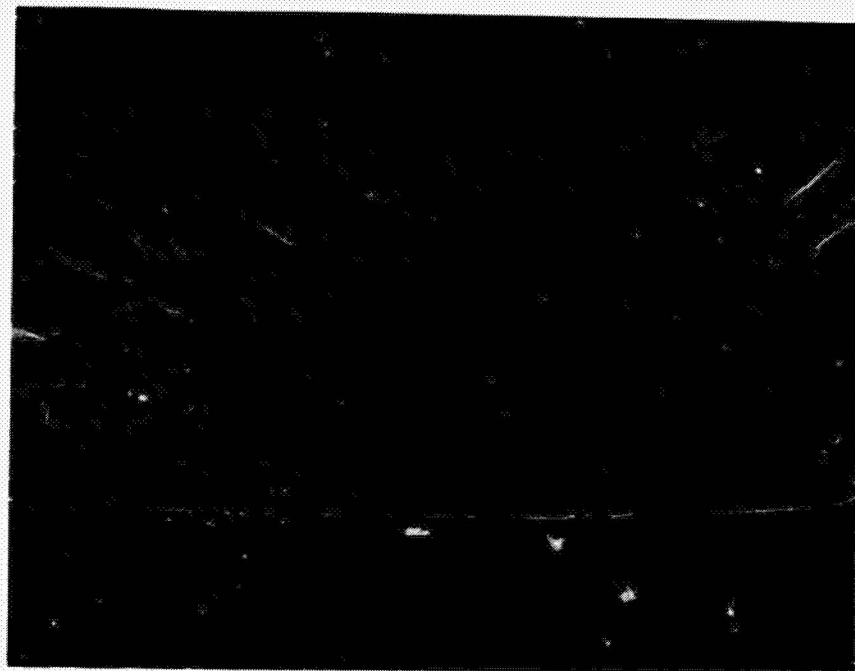


Figure D5. Fibredux 914 Neat Epoxy Tension, Specimen No. LFDA23, 23°C, Dry.

There are no distinguishing characteristics in this typical photograph of 914 epoxy to differentiate it from the 3502 epoxy fractures observed at the same test conditions.



Figure D6. Fibredux 914 Neat Epoxy Tension, Specimen No. LTDB2X, 54°C, Dry.

No discernible initiation site can be seen in the smooth area of this failed specimen.

ORIGINAL
OF PHOTOGRAPH

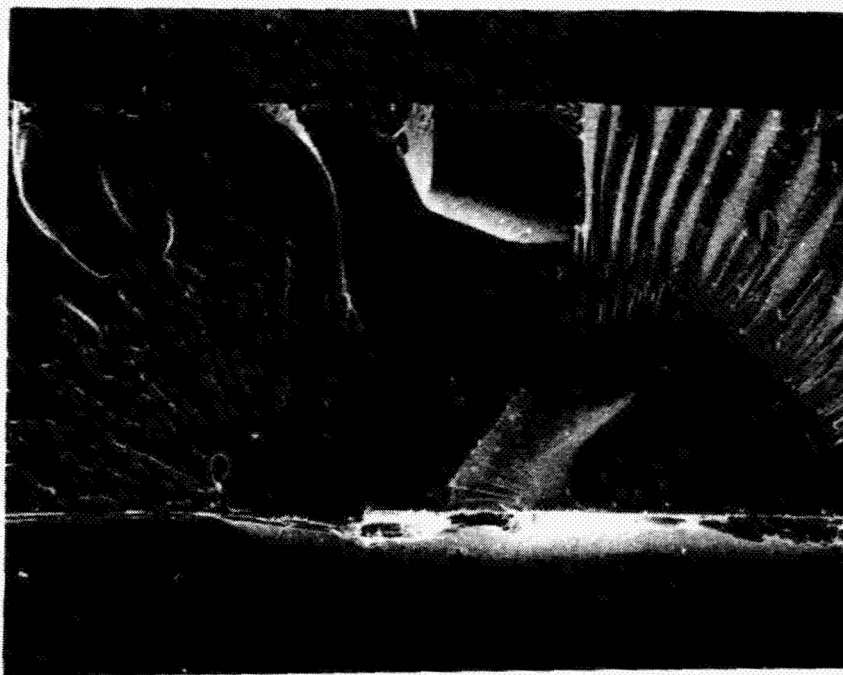


Figure D7. 2220-1 Neat Epoxy Tension, Specimen No. LTDA02, 23°C, Dry.

The smooth region of suspected failure initiation is located in the lower, slightly left of center part of this photograph. No discernible site is evident within the smooth region. The large region in the right half of the photograph is the surface where the characteristic chip was expelled at failure.

ORIGIN OF
OF POOR QUALITY

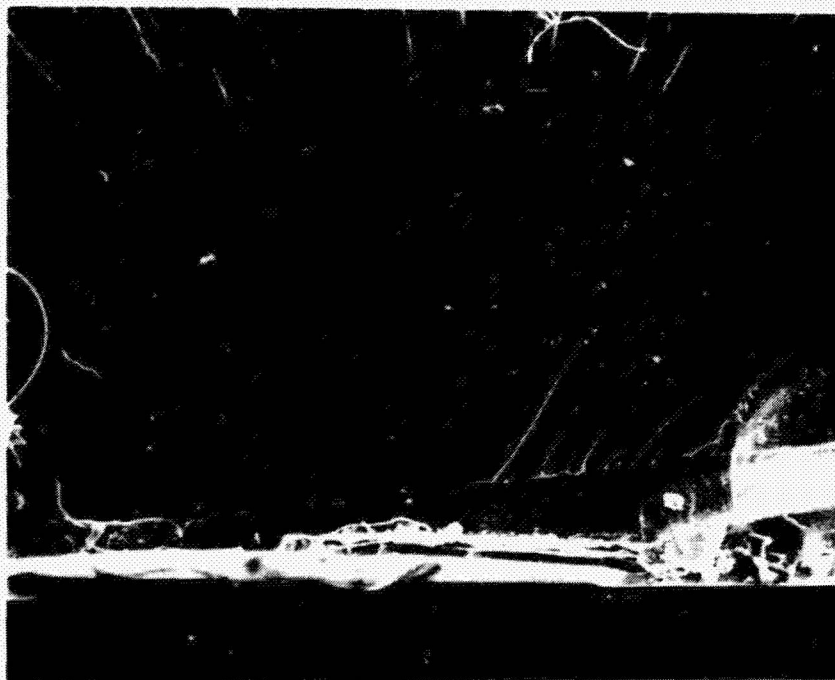


Figure D8. 2220-1 Neat Epoxy Tension, Specimen No. LTDA02, 23°C, Dry.

The very small smooth region at the lower edge of the specimen, surrounded by the transition region, is shown in this close-up of Figure D7.

CRACK
OF FLOW

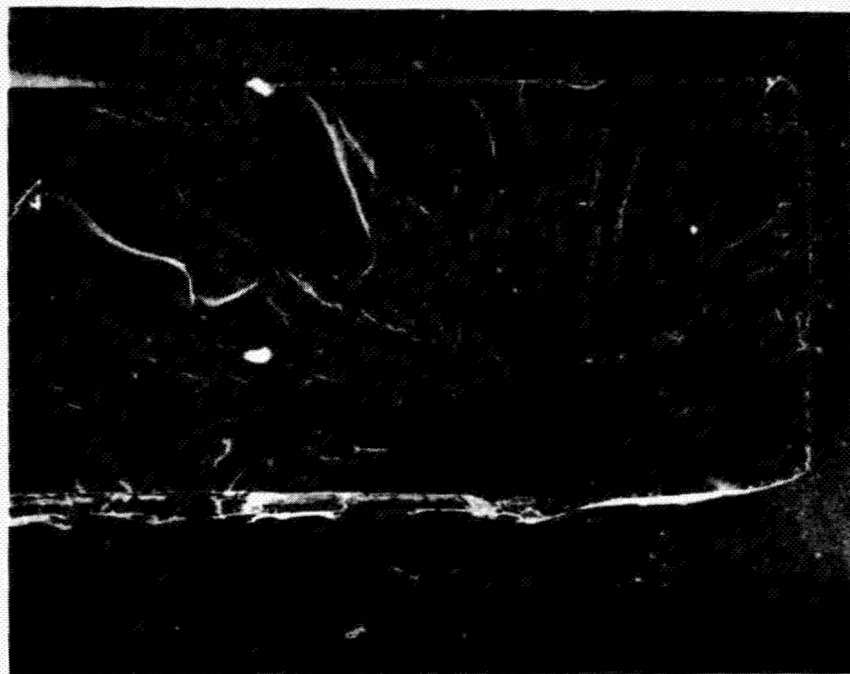


Figure D9. 2220-1 Neat Epoxy Tension, Specimen No. LTDB02, 54°C, Dry.

No specific failure site is apparent for this failed specimen.



Figure D10. 2220-1 Neat Epoxy Tension, Specimen No. LTWC03, 82°C, Moisture-Saturated.

Very little, if any, difference was seen between the wet and dry failures at 82°C for the 2220-1 epoxy system.

ORIGINAL
OF POOR QUALITY

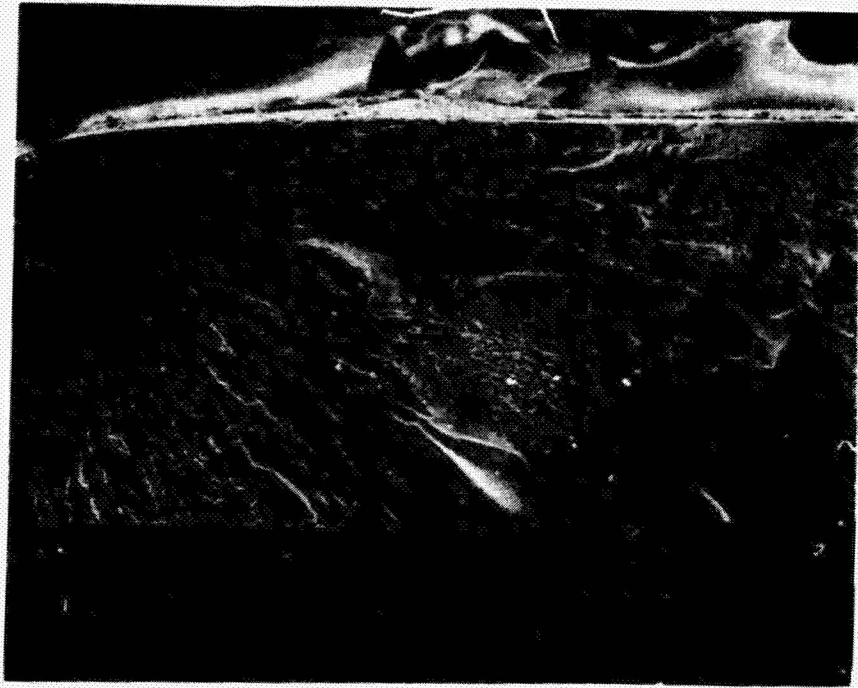


Figure D11. 2220-3 Neat Epoxy Tension, Specimen No. LTDA12, 23°C, Dry.

The fracture initiation site is located in the upper left corner of the specimen. It is surrounded by the transition and rough areas with the chip area on the right-hand side and partially out of the field of view of the photograph.

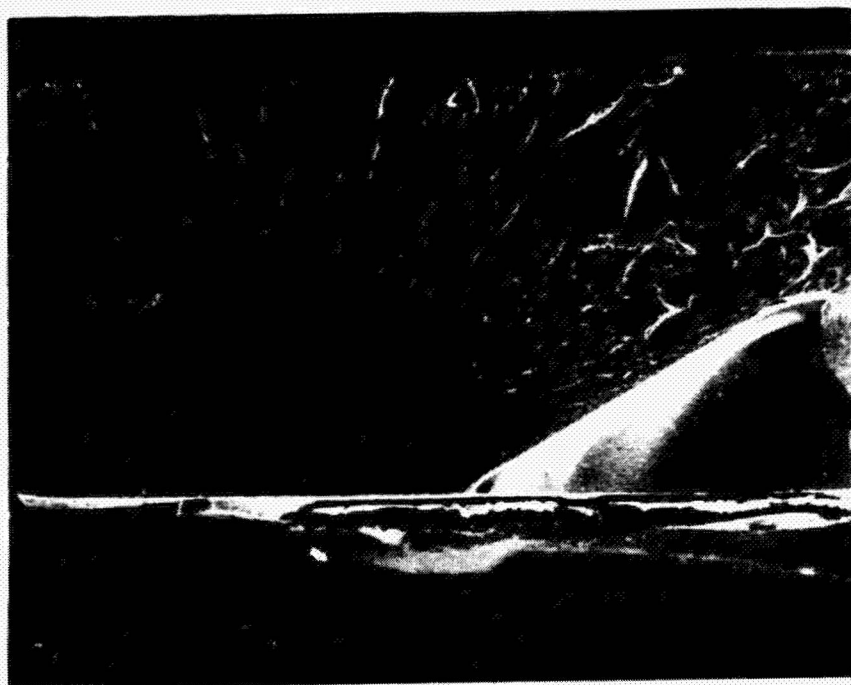


Figure D12. 2220-3 Neat Epoxy Tension, Specimen No. LTDA13, 23°C, Dry.

This specimen has a large chip missing in the lower right-hand corner, with the failure initiation site located to its left.

CLASSIFIED BY
OF POOR QUALITY

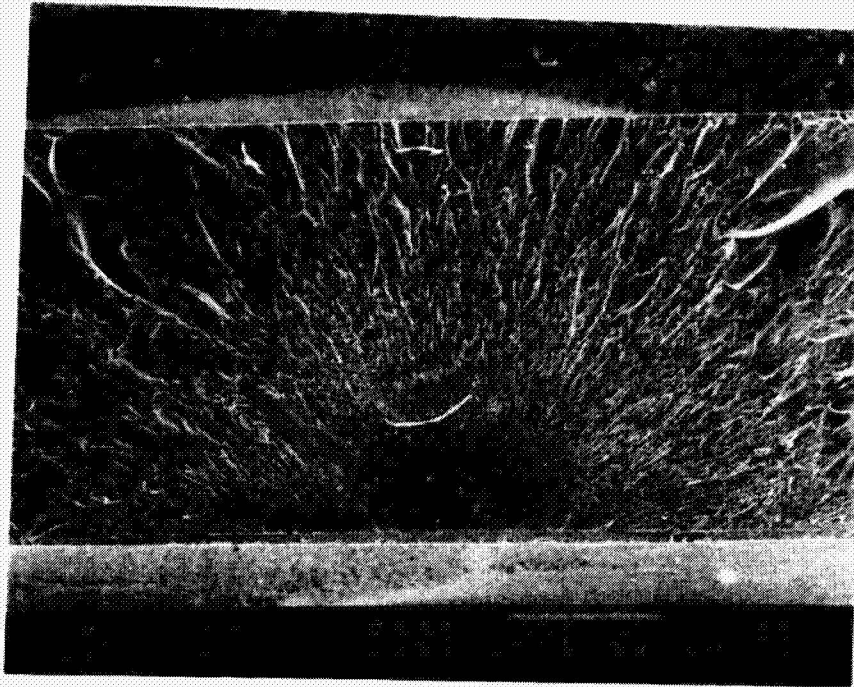


Figure D13. 2220-3 Neat Epoxy Tension, Specimen No. LTDA15, 23°C, Dry.

OF FORTH



Figure D14. 2220-3 Neat Epoxy Tension, Specimen No. LTDA16, 23°C, Dry.

ONE OF FOUR QUARTERS

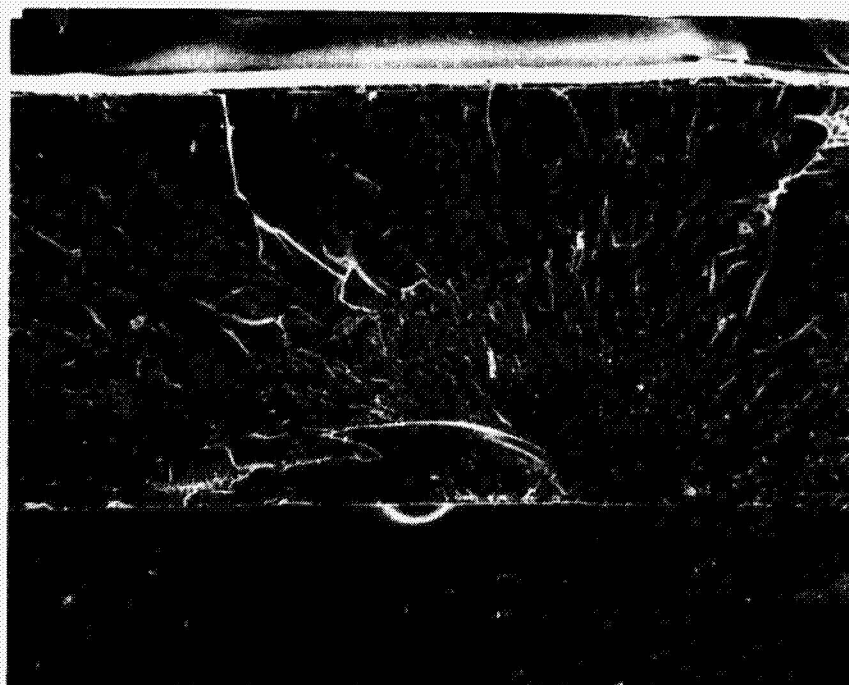


Figure D15. 2220-3 Neat Epoxy Tension, Specimen No. LTDA17, 23°C, Dry.

OF POLYMER

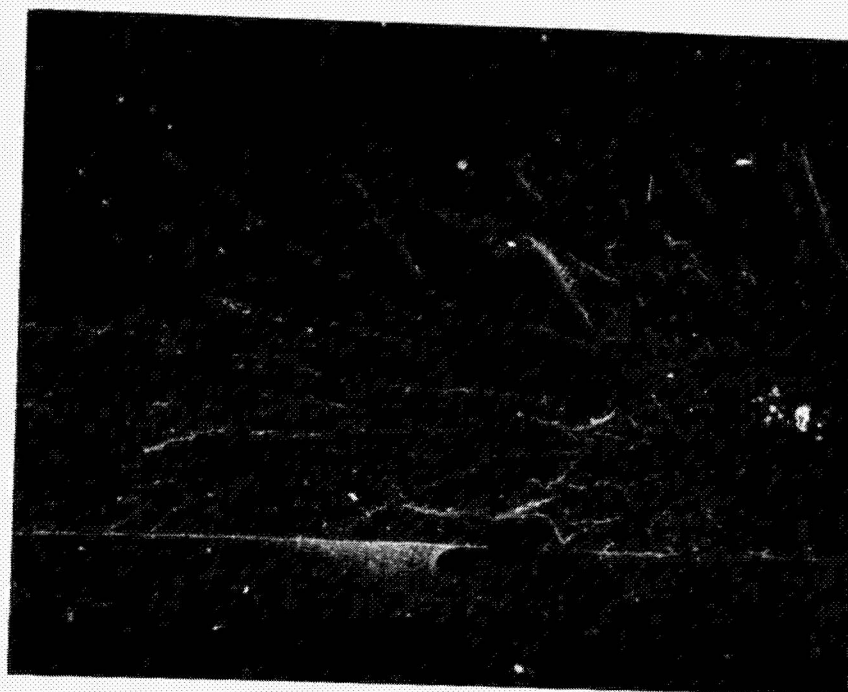


Figure D16. 2220-3 Neat Epoxy Tension Specimen No. LTWB13, 54°C, Moisture-Saturated.

This specimen appears very similar to the dry failed specimen at 54°C.

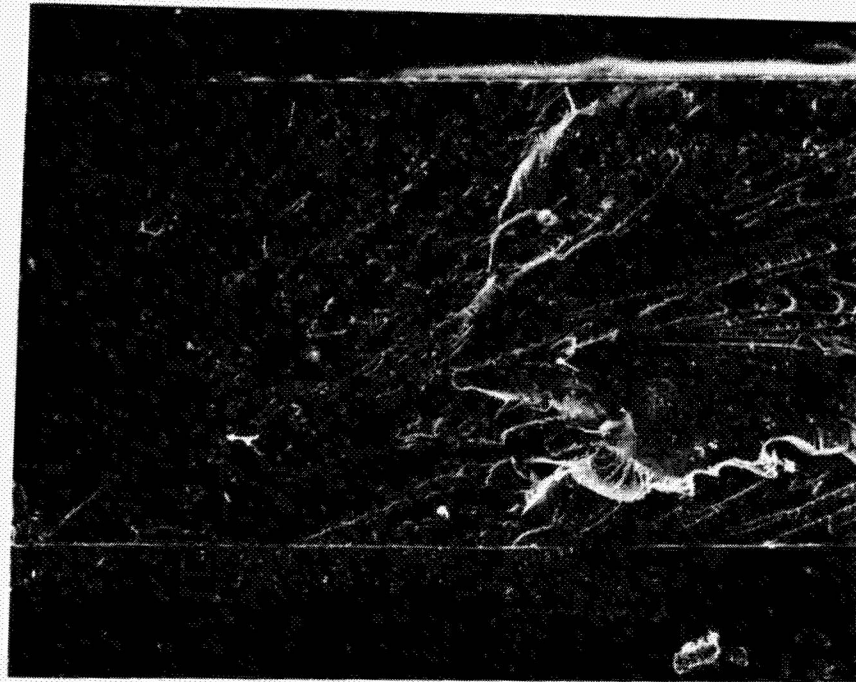


Figure D17. 2220-3 Neat Epoxy Tension, Specimen No. LTWC11, 82°C, Moisture-Saturated.

This specimen failure appears quite similar to the dry 82°C specimen failures except for the small internal void within the smooth region on the left.

GROUP
OF FIBERS

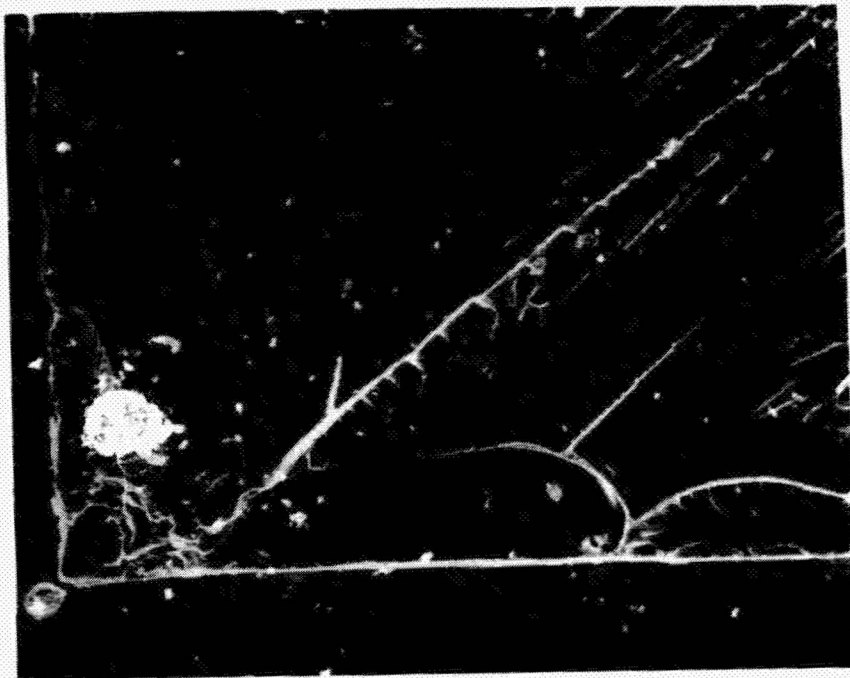


Figure D18. 222G-3 Neat Epoxy Tension, Specimen No. LTWC11, 82°C, Moisture-Saturated.

This closeup of Figure D17 shows the void area directly in the corner next to a missing chip area.

CON
OF POOR

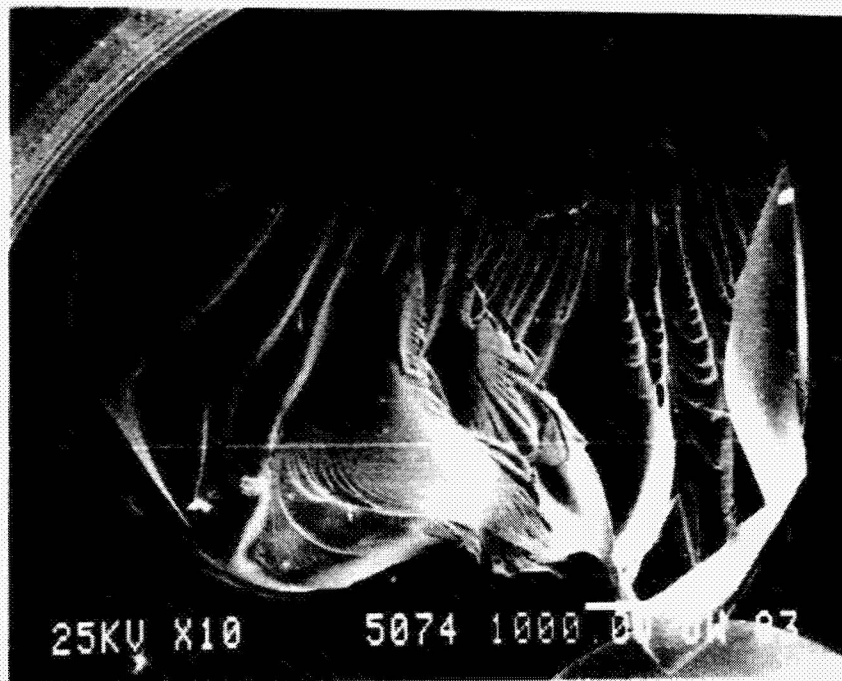


Figure D19. 3502 Neat Epoxy Torsional Shear, 54°C, Moisture-Saturated.

This failure is quite similar to those observed in the 54°C, dry specimens.

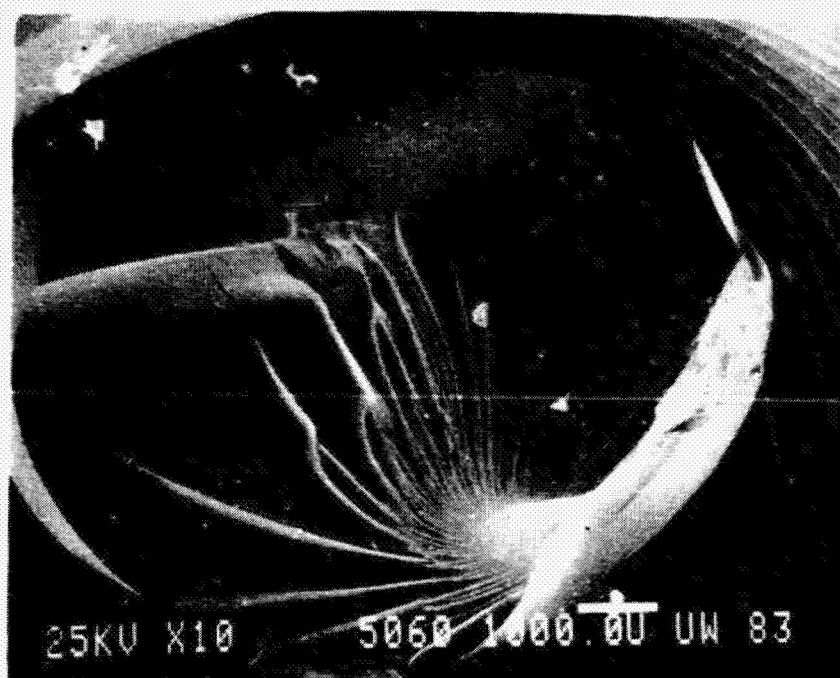


Figure D20. 3502 Neat Epoxy Torsion Shear, 82°C, Dry.

This failure has a much smoother appearance away from the failure initiation site and transition zone in the lower right-hand corner of the photograph..

ORIGINAL
OF PCL



Figure D21. 3502 Neat Epoxy Torsional Shear, 82°C, Moisture-Saturated.

A smooth area extending around the perimeter of the specimen, which is the most highly stressed region in a torsion test, is shown. The transition region extends inward from this surface region.

View
OF PO...

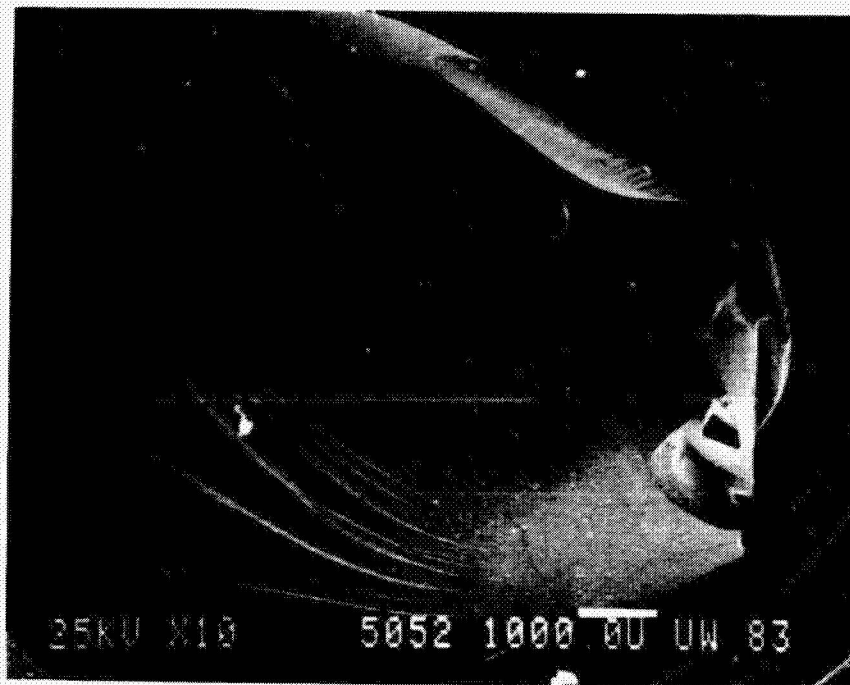


Figure D22. Fibredux 914 Neat Epoxy Torsional Shear, 23°C, Dry.

This failure surface indicates many missing pieces.

ORIGINAL
OF PHOTOGRAPH



Figure D23. Fibredux 914 Neat Epoxy Torsional Shear, 54°C, Dry.

ORIGIN OF
OF POCA



Figure D24. Fibredux 914 Neat Epoxy Torsional Shear 54°C,
Moisture-Saturated.

Post failure cracking due to dry-out is seen in the upper
right-hand corner of this cross section.

ORIGINAL
OF POOR QUALITY

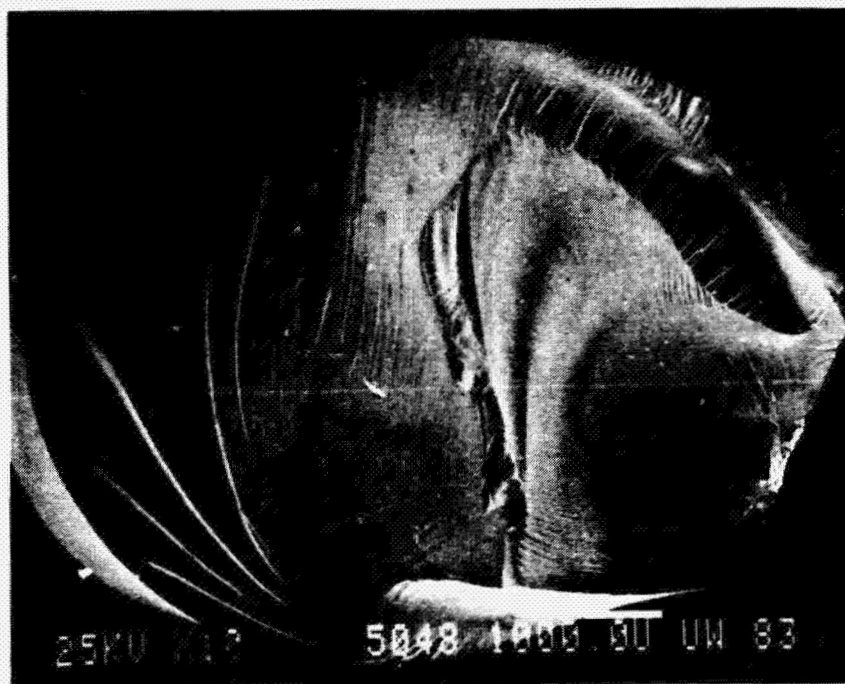


Figure D25. Fibredux 914 Neat Epoxy Torsional Shear, 82°C, Dry.

More evidence of the characteristic transition region and the striations associated with it is seen.

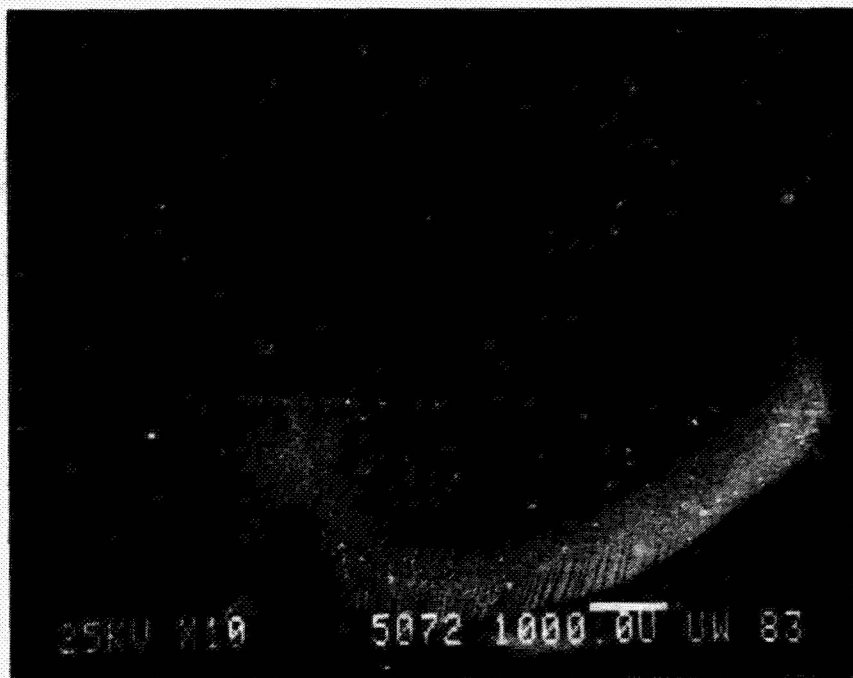


Figure D26. Fibredux 914 Neat Epoxy Torsional Shear, 82°C, Moisture-Saturated.

A very large post-failure crack is evident through this specimen due to high stress gradients induced during drying out after failure. Note also the loose pieces of resin extending over the crack in the upper left.

ORIGINAL PAGE IS
OF POOR QUALITY

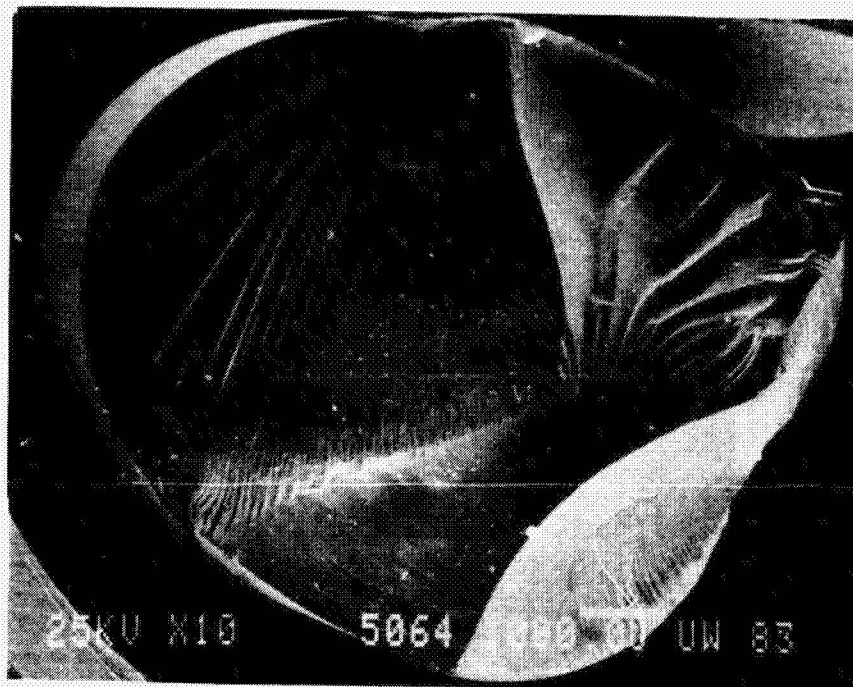


Figure D27. 2220-1 Neat Epoxy Torsional Shear, 23°C,
Moisture-Saturated.

Failure probably initiated at the upper center surface of this specimen.

ORIGINAL PHOTOGRAPH
OF POOR QUALITY

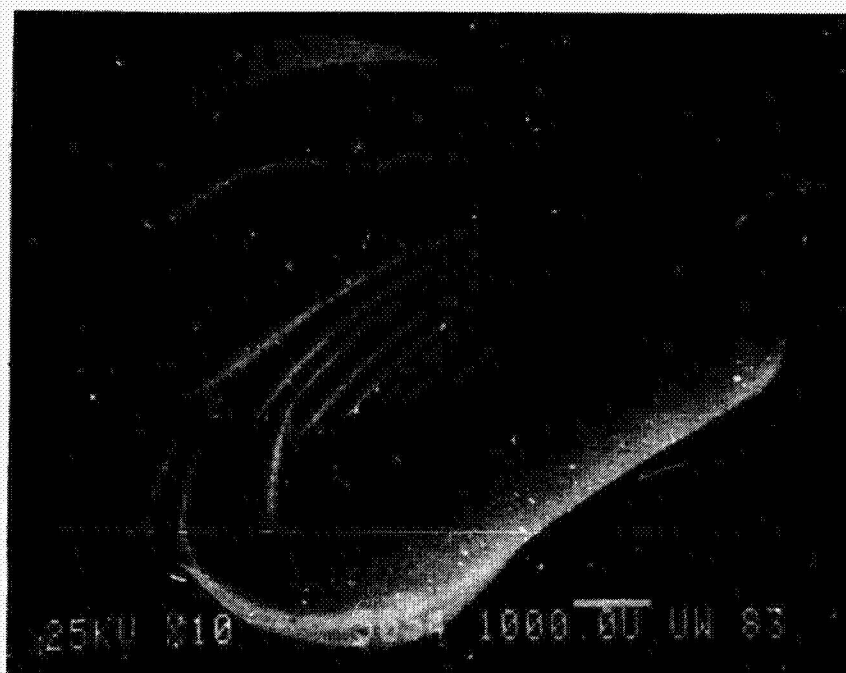


Figure D28. 2220-1 Neat Epoxy Torsional Shear, 54°C, Dry.

Failure probably began at the upper right-hand surface of this specimen.

ORIGINAL PHOTOGRAPH
OF POOR QUALITY

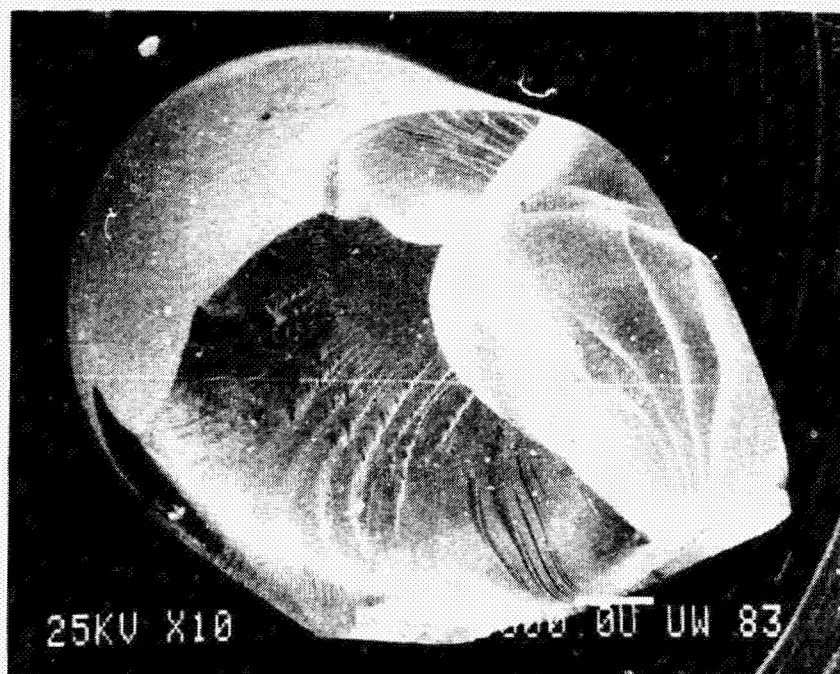


Figure D29. 2220-1 Neat Epoxy Torsional Shear, 82°C, Dry.

The location of the initial failure site was lost with the chip that broke loose from the upper right side of this specimen during fracture.

CRACK
OF EPOXY



Figure D30. 2220-3 Neat Epoxy Torsional Shear, 23°C, Dry.

ORIGINAL PHOTOGRAPH
OF POOR QUALITY

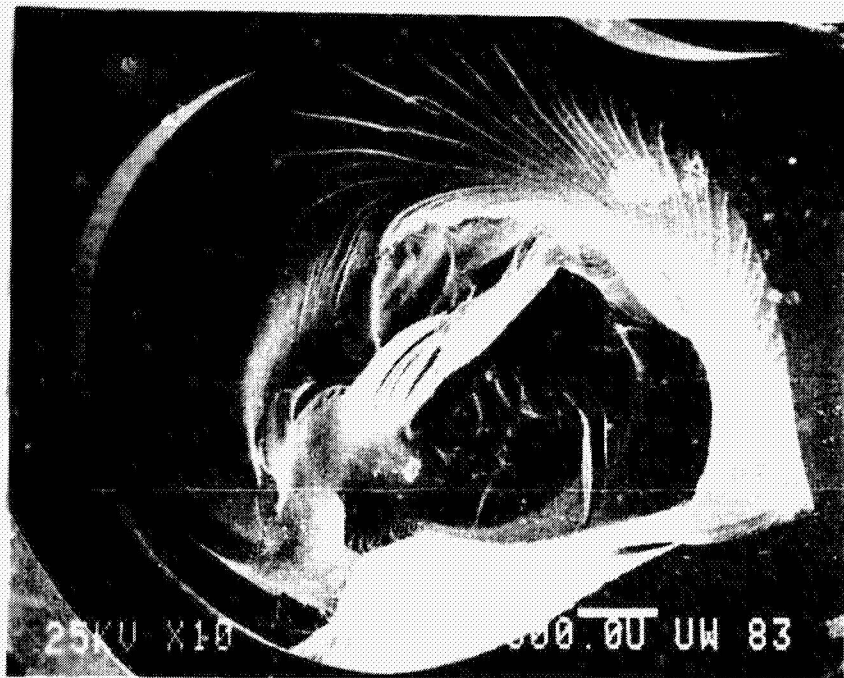


Figure D31. 2220-3 Neat Epoxy Torsional Shear, 23°C,
Moisture-Saturated.

One of the
Or POOR Quality



Figure D32. 2220-3 Neat Epoxy Torsional Shear, 82°C,
Moisture-Saturated.

The many cracks present in this specimen will be noted. It is not known if they occurred at failure, or after due to moisture-induced dry-out stresses. The latter is probable.

ORIGIN OF
OF POOR QUALITY



Figure D33. 2220-3 Neat Epoxy Fracture Toughness, 23°C, Dry.

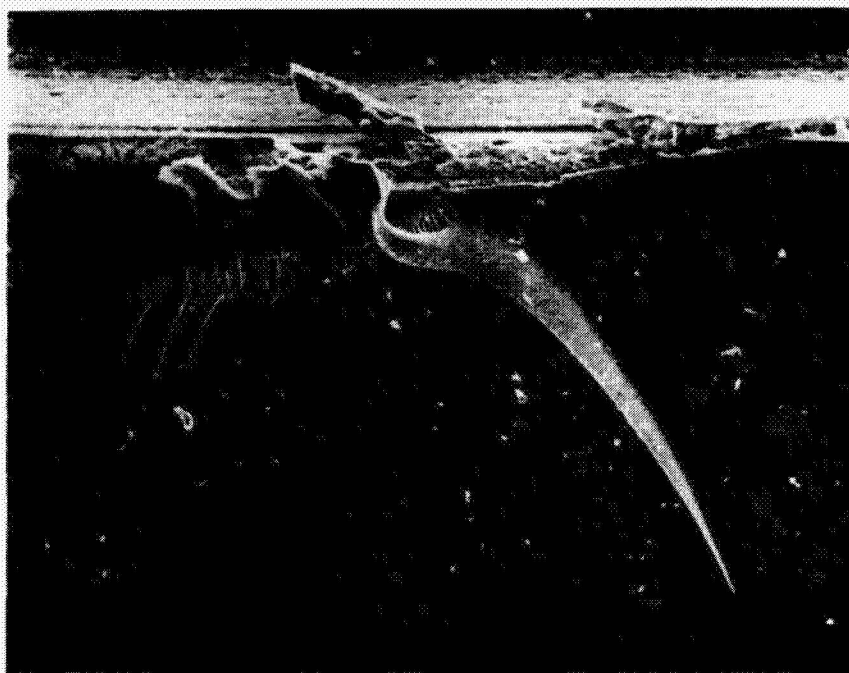


Figure D34. 2220-3 Neat Epoxy Fracture Toughness, 82°C, Dry.

The notch area is located at the top of this photograph.

APPENDIX E

PLOTS OF INTERNAL STRESS STATES IN AS4 GRAPHITE FIBER-REINFORCED

UNDIRECTIONAL COMPOSITES INCORPORATING

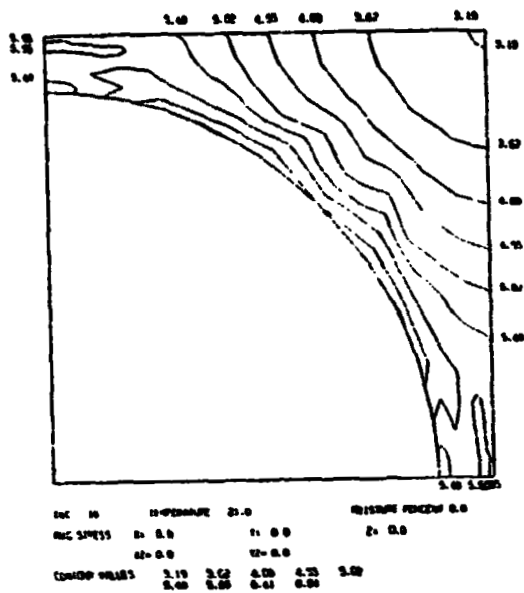
3502, 2220-3, and 914 MATRIX MATERIALS

APPENDIX E1

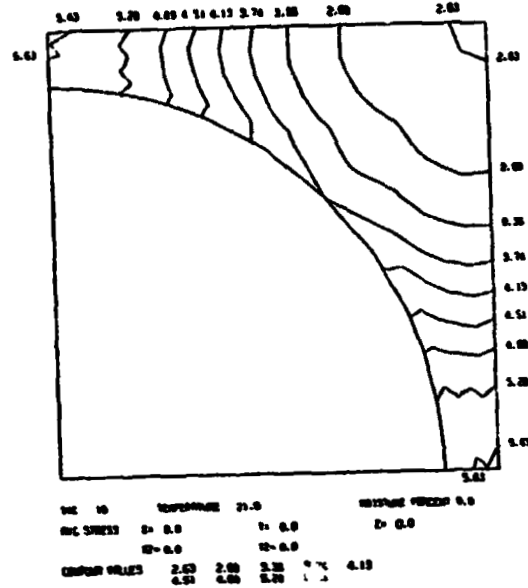
AS4/3502 GRAPHITE/EPOXY UNIDIRECTIONAL COMPOSITE

PRECEDING PAGE BLANK NOT FILMED

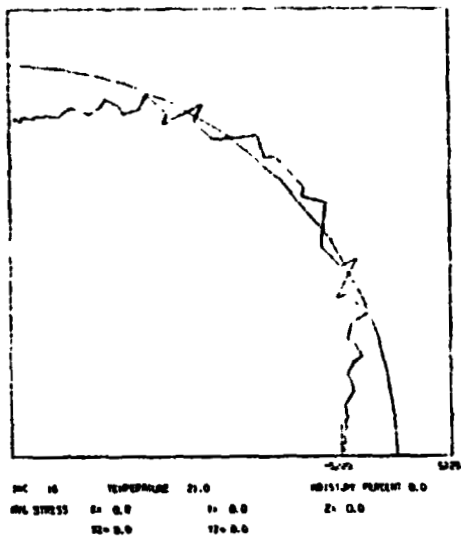
ORIGIN OF POOR QUALITY



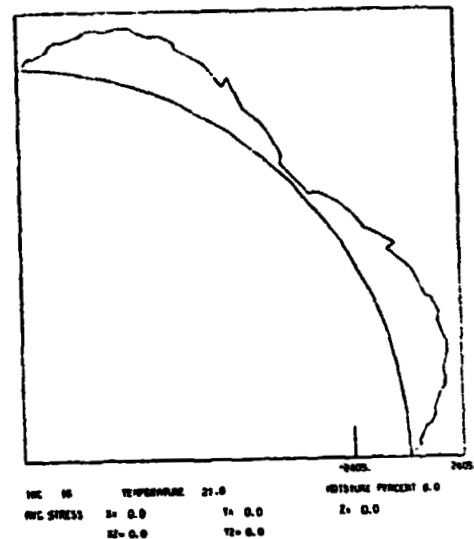
e) Intermediate Principal Stress (ksi)



f) Maximum Shear Stress (ksi)

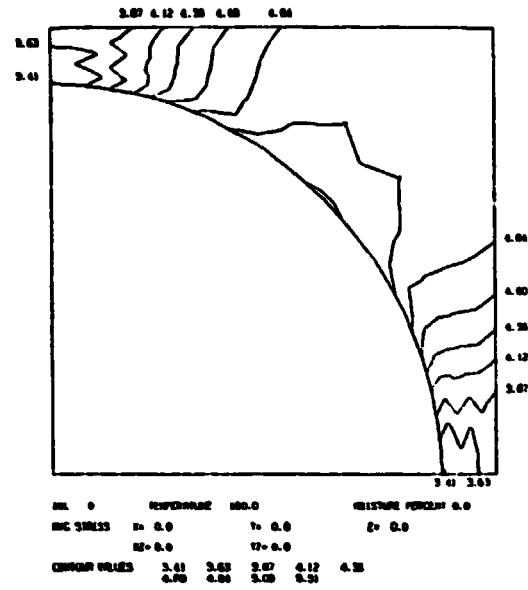
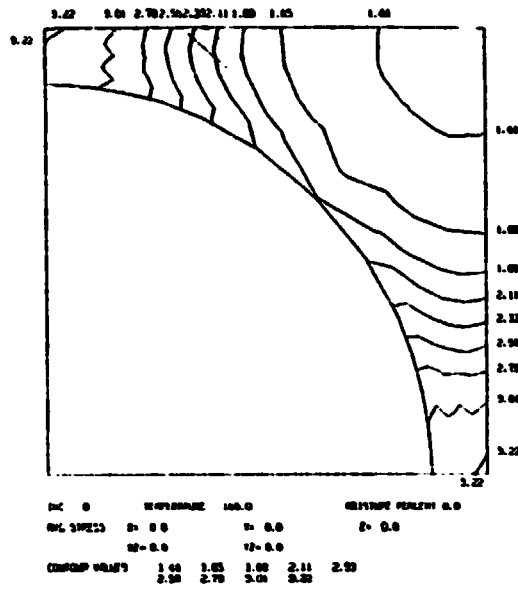


g) Interface Normal Stress (psi)

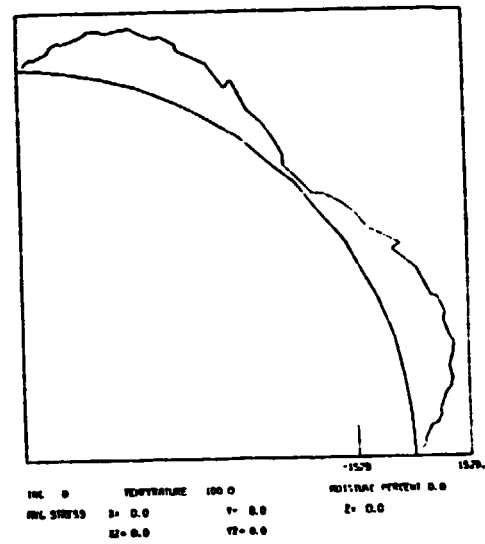
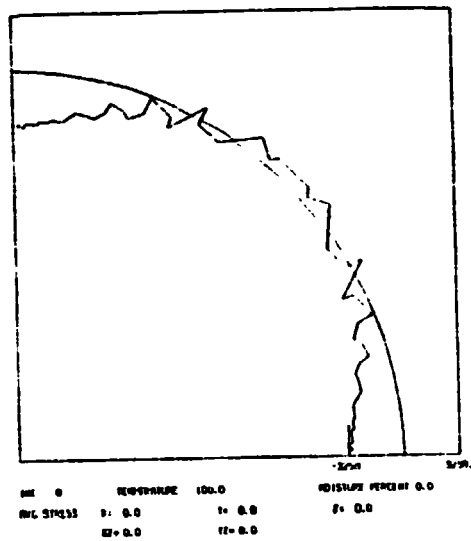


h) Interface Shear Stress (psi)

Figure E1 (continued). AS4/3502 Graphite/Epoxy Unidirectional Composite, Room Temperature, Dry (RTD); No Mechanical Loading.

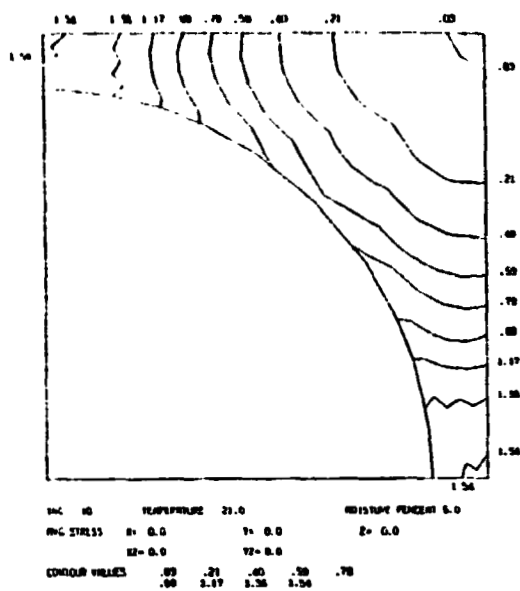


a) Octahedral Shear Stress (ksi) b) Maximum Principal Stress (ksi)

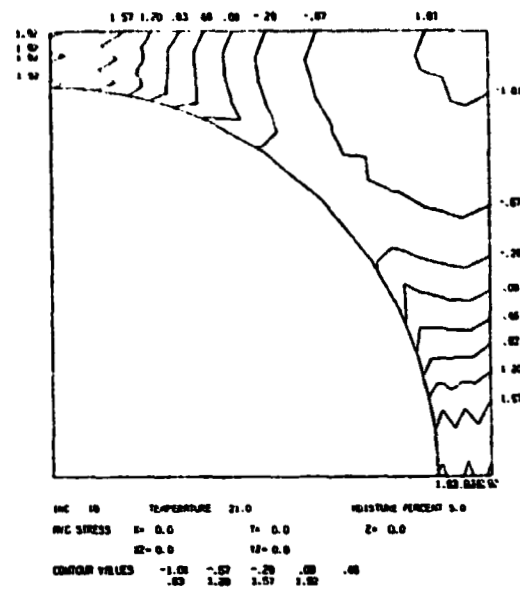


c) Interface Normal Stress (psi) d) Interface Shear Stress (psi)

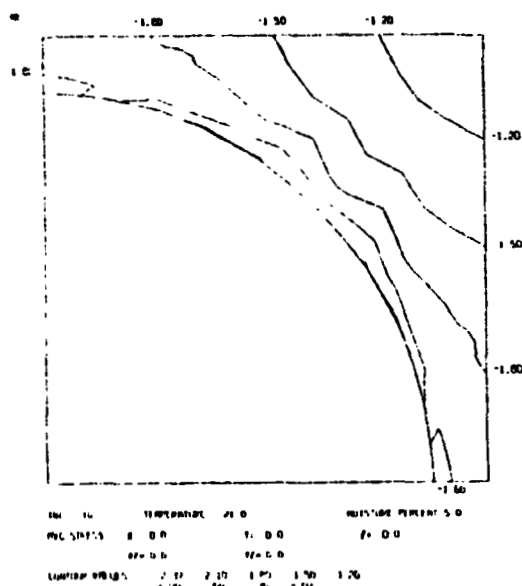
Figure E2. AS4/3502 Graphite/Epoxy Unidirectional Composite, 100°C, Dry (ETD); No Mechanical Loading.



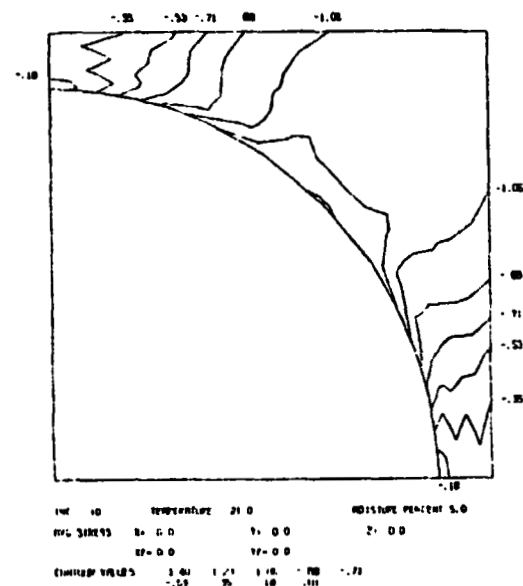
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

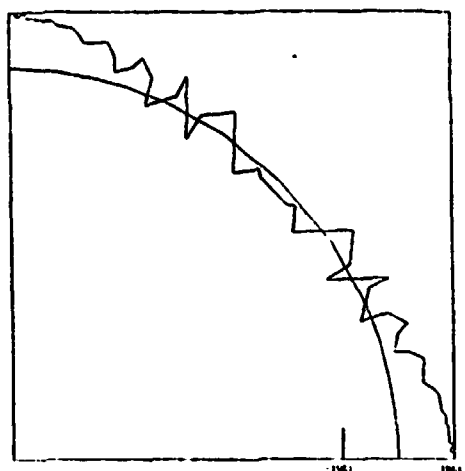


c) Minimum Principal Stress (ksi)

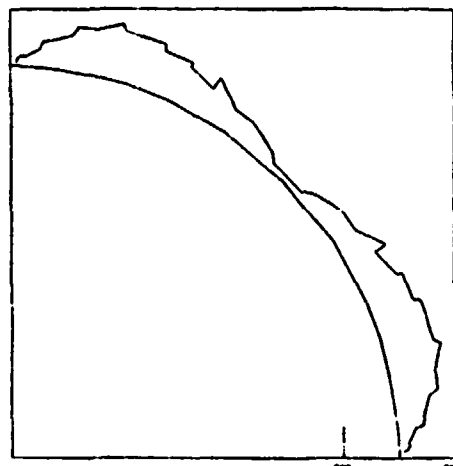


d) Intermediate Principal Stress (ksi)

Figure E3. AS4/3502 Graphite/Epoxy Unidirectional Composite, Room Temperature, 5.0 Percent Moisture (RTW); No Mechanical Loading.



INC 10 TEMPERATURE 21.0 MOISTURE PERCENT 5.0
 AVG STRESS 0.0 0.0 0.0
 SZ= 0.0 TX= 0.0

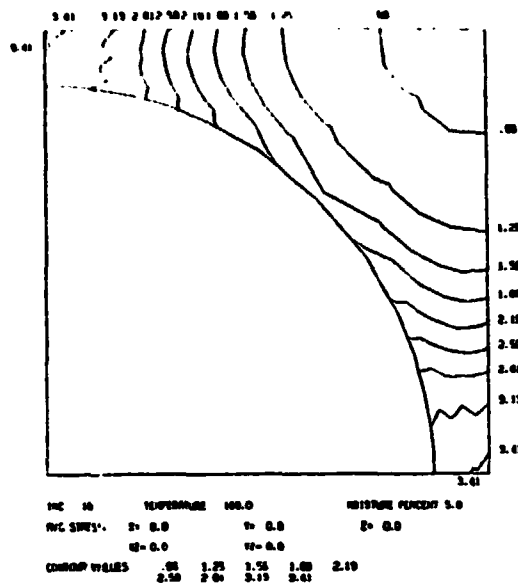


INC 10 TEMPERATURE 21.0 MOISTURE PERCENT 5.0
 AVG STRESS 0.0 0.0 0.0
 SZ= 0.0 TX= 0.0

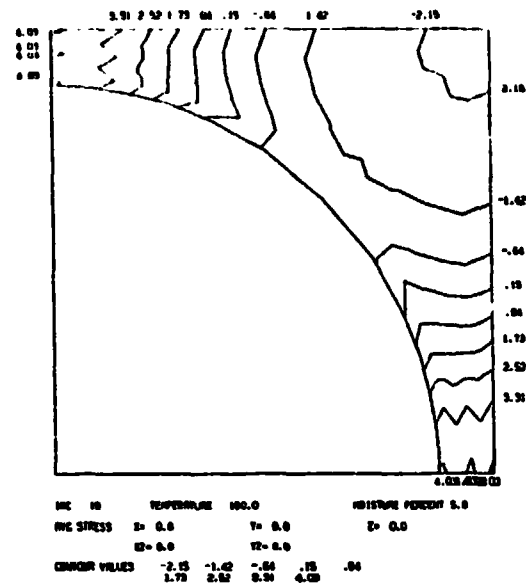
e) Interface Normal Stress (psi)

f) Interface Shear Stress (psi)

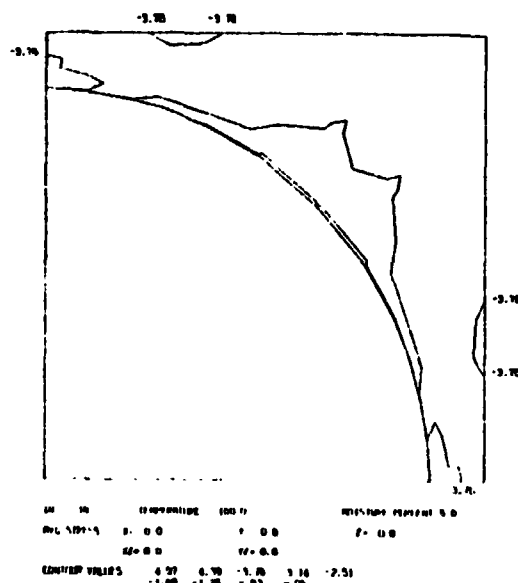
Figure E3 (continued). AS4/3502 Graphite/Epoxy Unidirectional Composite,
 Room Temperature, 5.0 Percent Moisture (RTW); No Mechanical
 Loading.



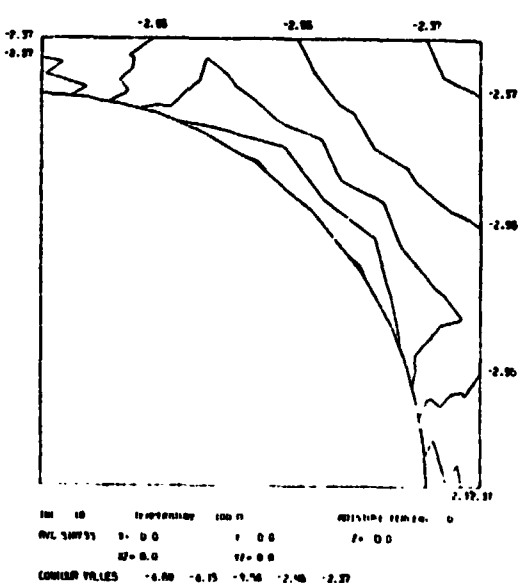
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

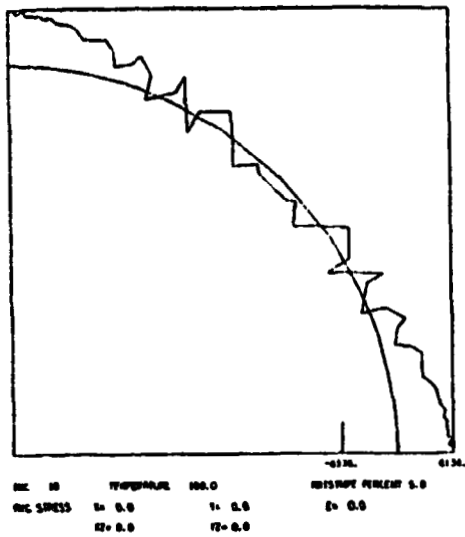


c) Minimum Principal Stress (ksi)

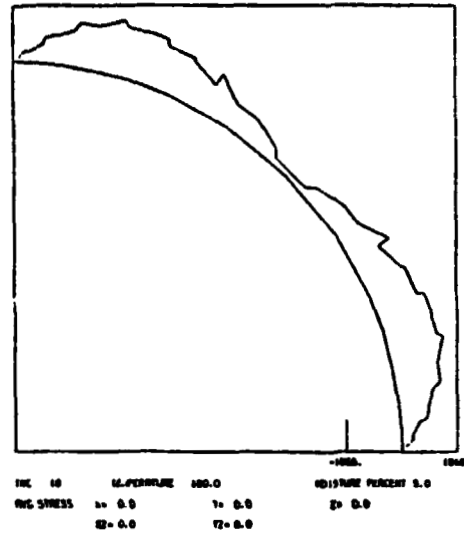


d) Intermediate Principal Stress (ksi)

Figure E4. AS4/3502 Graphite/Epoxy Unidirectional Composite, 100°C, 5.0 Percent Moisture (ETW); No Mechanical Loading.

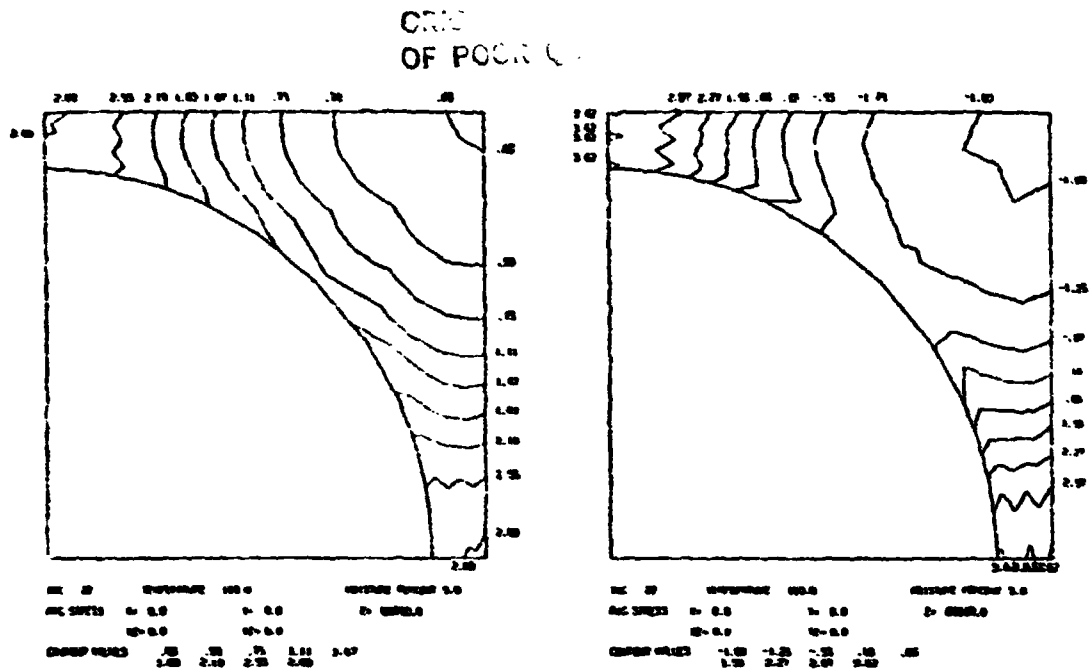


e) Interface Normal Stress (psi)

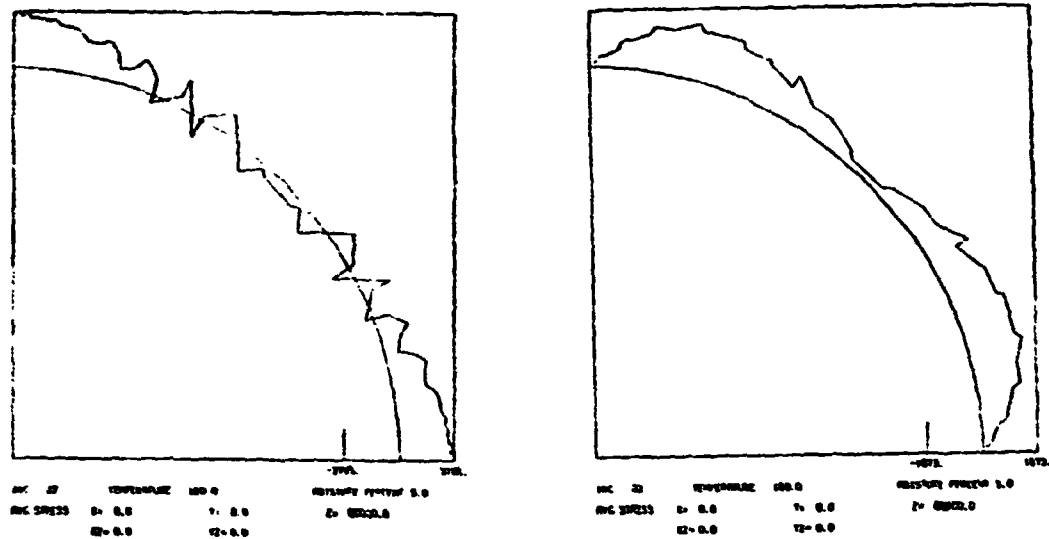


f) Interface Shear Stress (psi)

Figure E4 (continued). AS4/3502 Graphite/Epoxy Unidirectional Composite, 100°C, 5.0 Percent Moisture (ETW); No Mechanical Loading.

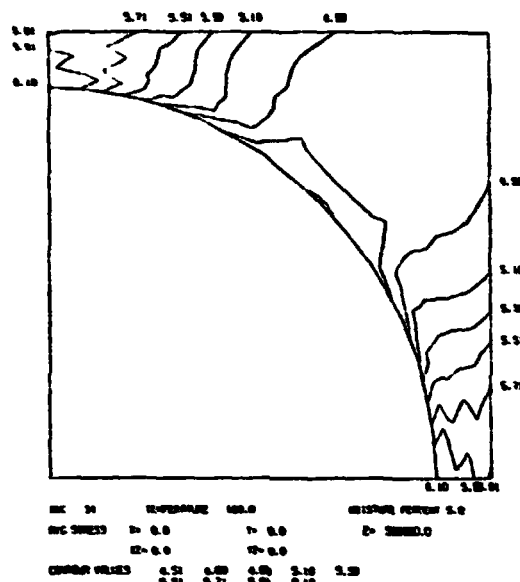
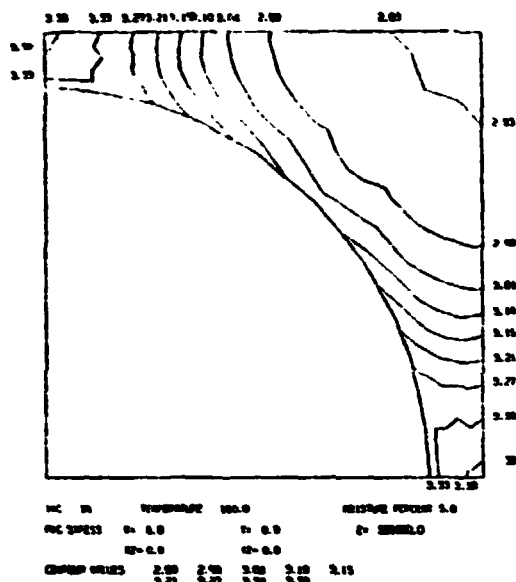


a) Octahedral Shear Stress (ksi) b) Maximum Principal Stress (ksi)



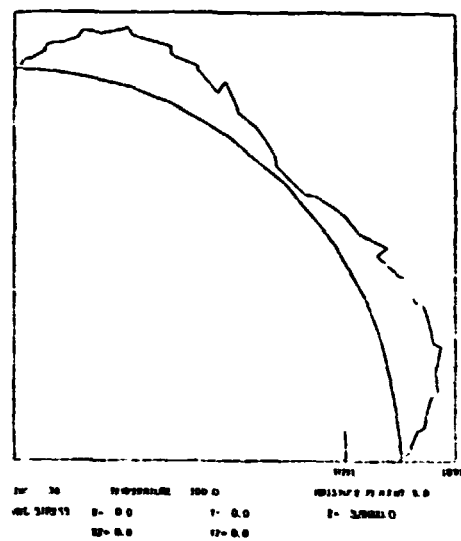
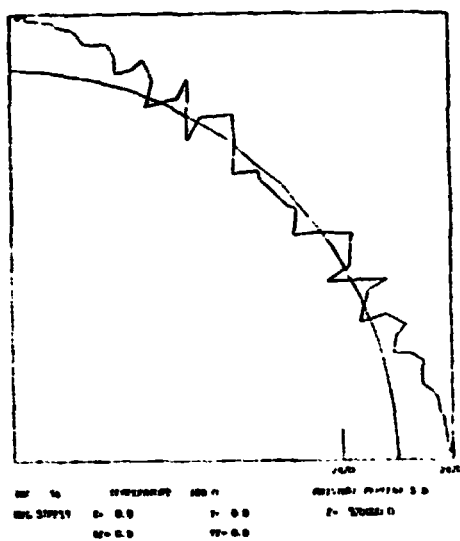
c) Interface Normal Stress (psi) d) Interface Shear Stress (psi)

Figure E8. AS4/3502 Graphite/Epoxy Unidirectional Composite, 100°C, 5.0 Percent Moisture (ETW); 0.55 GPa (80 ksi) Longitudinal Tensile Applied Stress.



a) Octahedral Shear Stress (ksi)

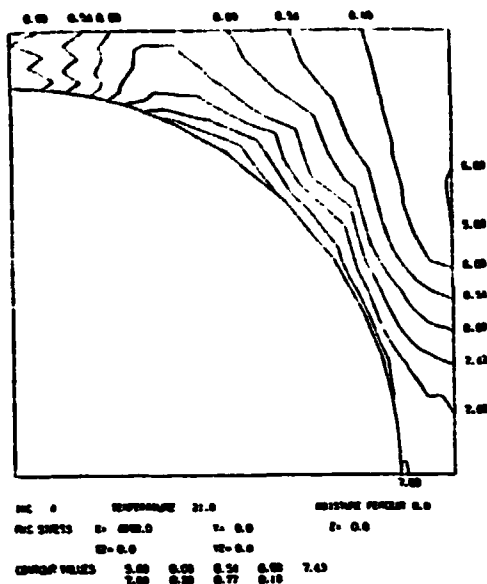
b) Maximum Principal Stress (ksi)



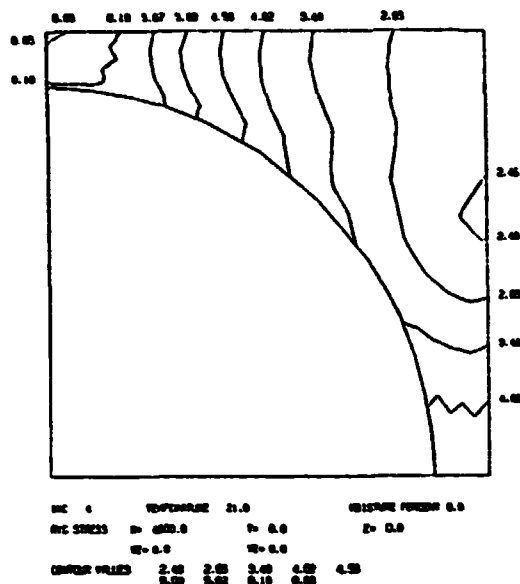
c) Interface Normal Stress (psi)

d) Interface Shear Stress (psi)

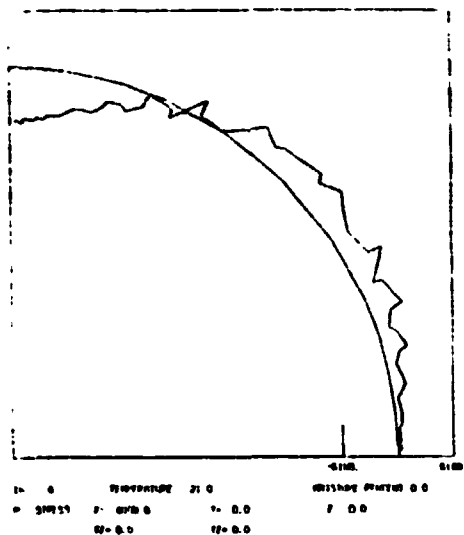
Figure E9. AS4/3502 Graphite/Epoxy Unidirectional Composite, 100°C, 5.0 Percent Moisture (ETW); 2.20 GPa (320 ksi) Longitudinal Tensile Applied Stress.



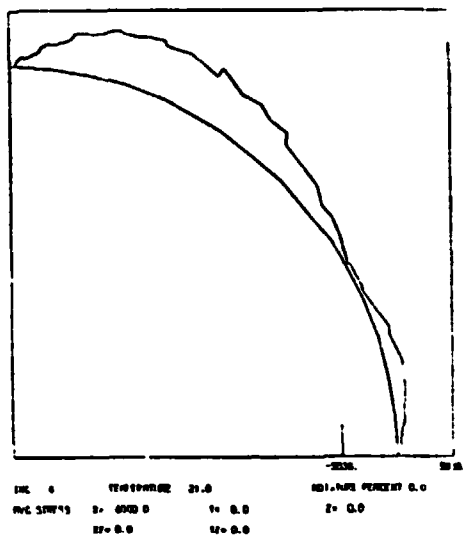
e) Intermediate Principal Stress (ksi)



f) Maximum Shear Stress (ksi)

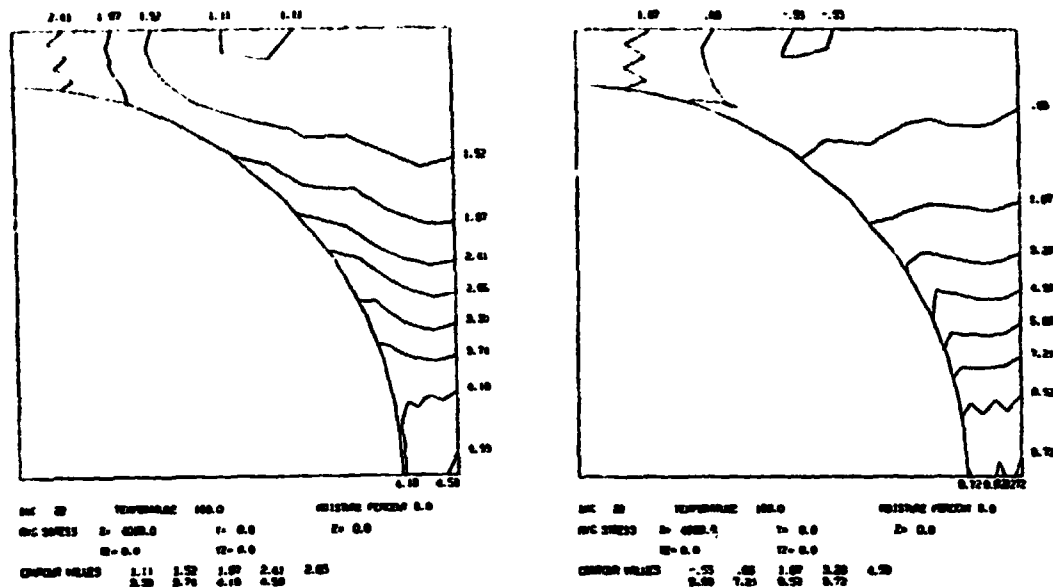


g) Interface Normal Shear (psi)

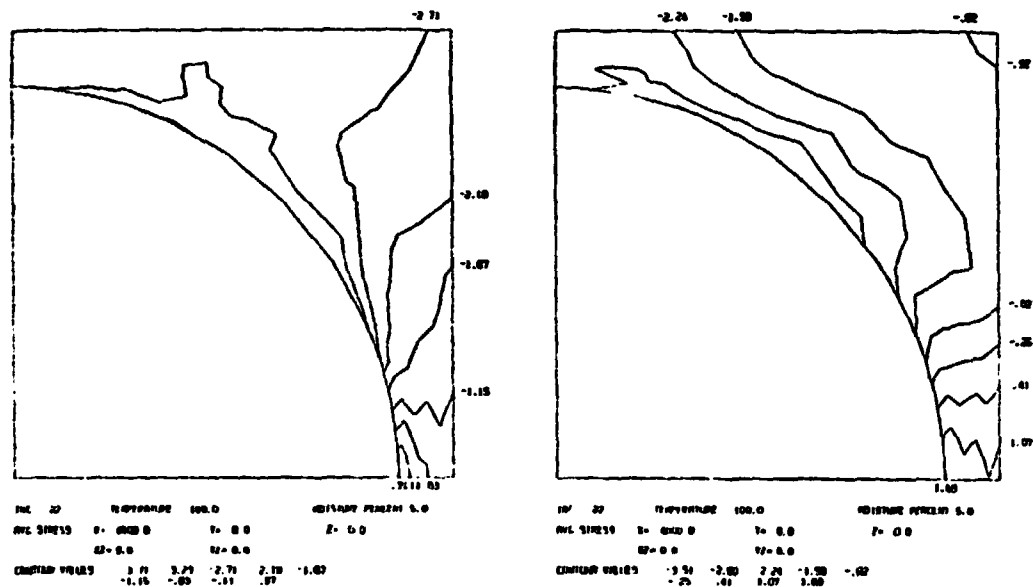


h) Interface Shear Stress (psi)

Figure E10 (continued). AS4/3502 Graphite/Epoxy Unidirectional Composite, Room Temperature, Dry (RTD); 27.6 MPa (4 ksi) Transverse Tensile Applied Stress.

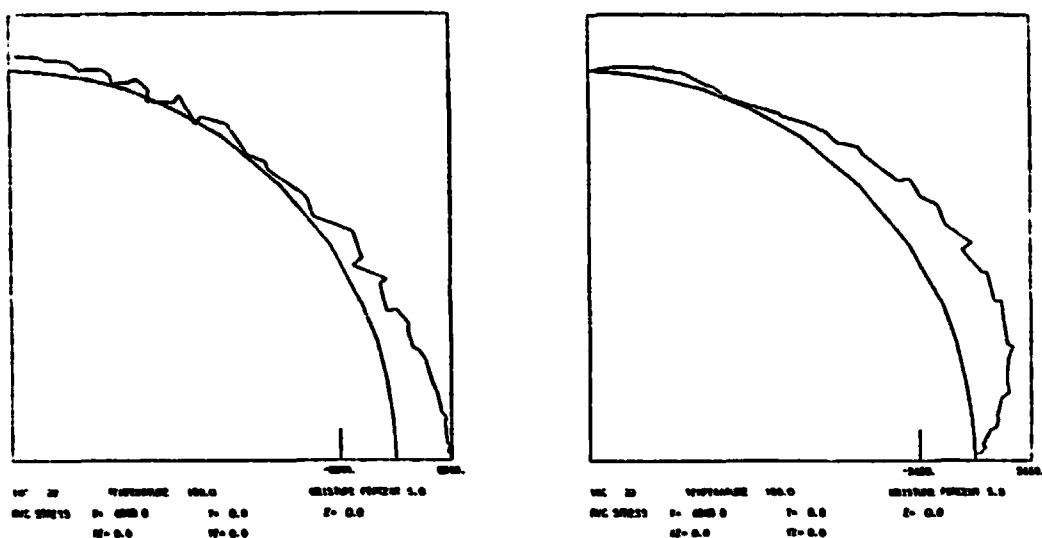


a) Octahedral Shear Stress (ksi) b) Maximum Principal Stress (ksi)



c) Minimum Principal Stress (ksi) d) Intermediate Principal Stress (ksi)

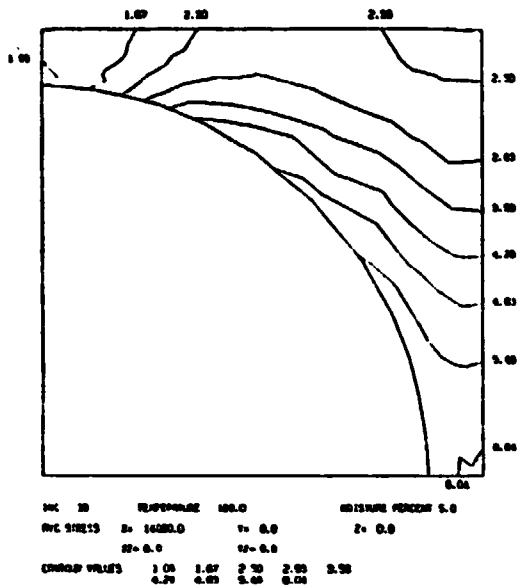
Figure E13. AS4/3502 Graphite/Epoxy Unidirectional Composite, 100°C, 5.0 Percent Moisture (ETW); 27.6 MPa (4 ksi) Transverse Tensile Applied Stress.



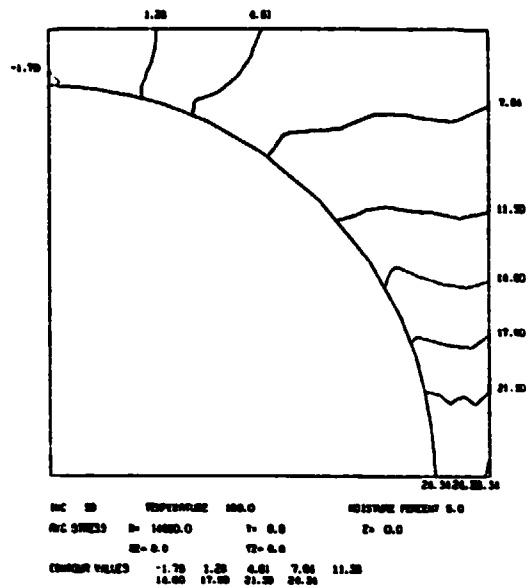
e) Interface Normal Stress (psi)

f) Interface Shear Stress (psi)

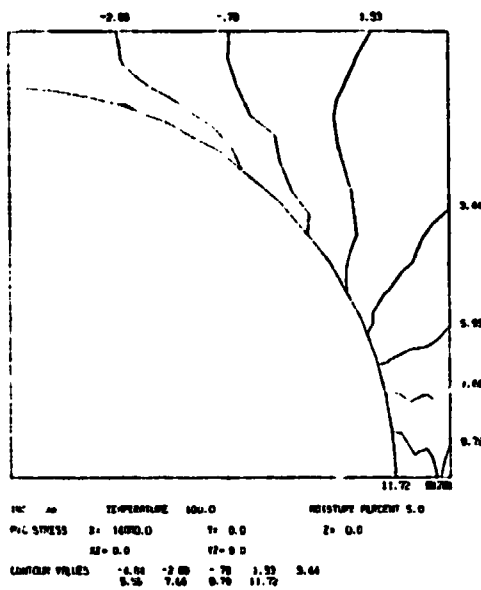
Figure E13 (continued). AS4/3502 Graphite/Epoxy Unidirectional Composite, 100°C, 5.0 Percent Moisture (ETW); 27.6 MPa (4 ksi) Transverse Tensile Applied Stress.



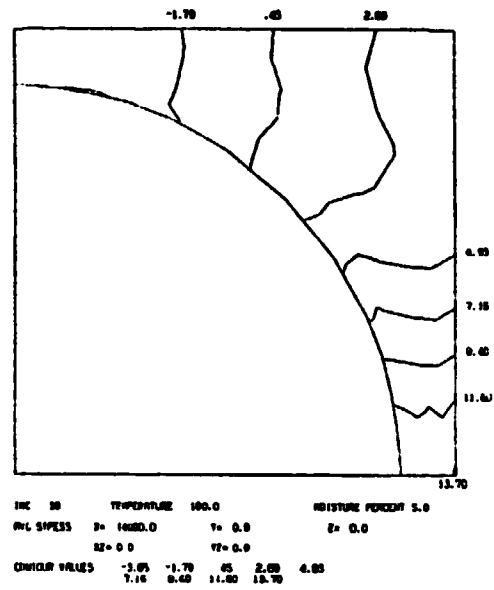
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)



c) Minimum Principal Stress (ksi)

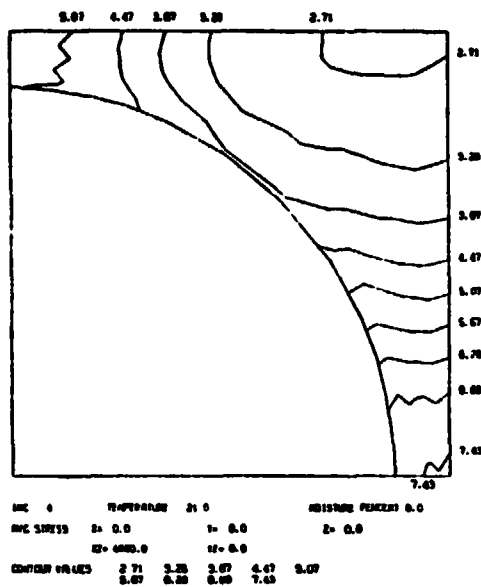


d) Intermediate Principal Stress (ksi)

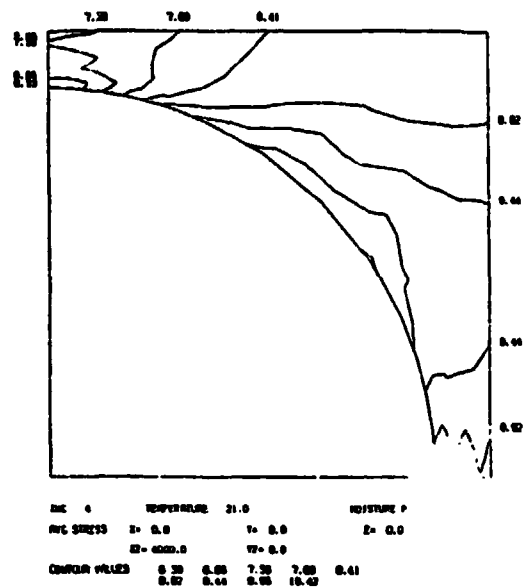
Figure E14. AS4/3502 Graphite/Epoxy Unidirectional Composite, 100°C, 5.0 Percent Moisture (ETW); 97 MPa (14 ksi) Transverse Tensile Applied Stress.



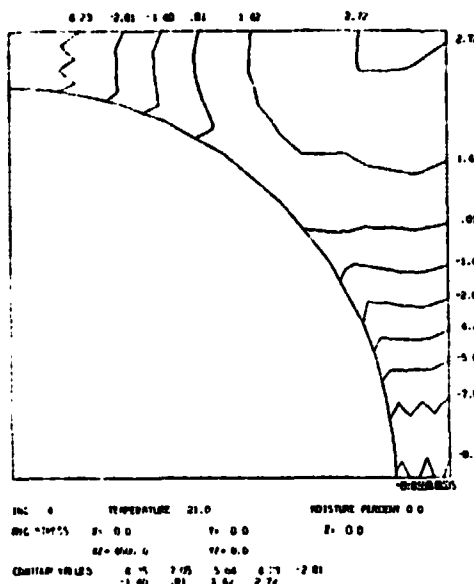
120



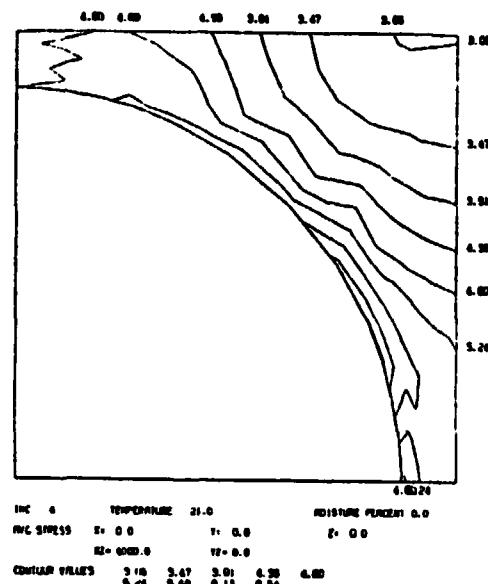
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

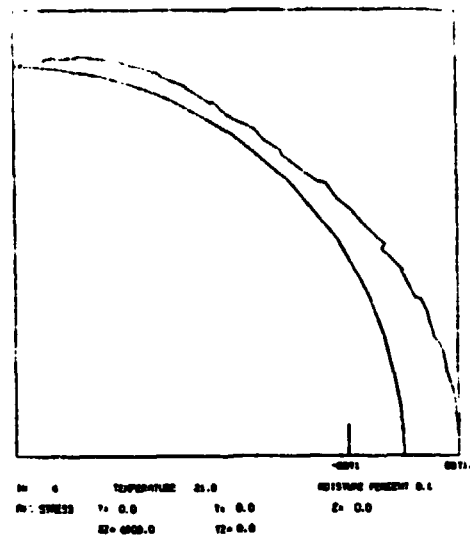
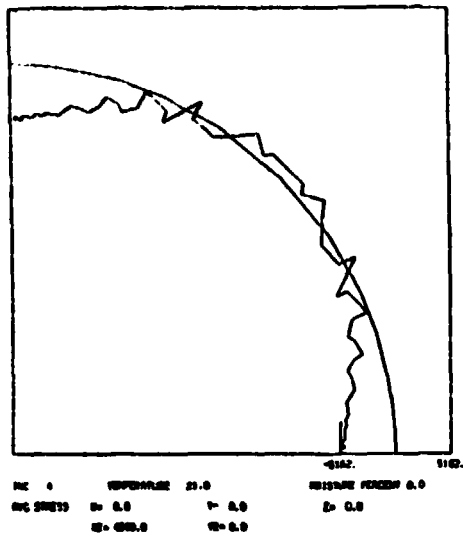


c) Minimum Principal Stress (ksi)



d) Intermediate Principal Stress (ksi)

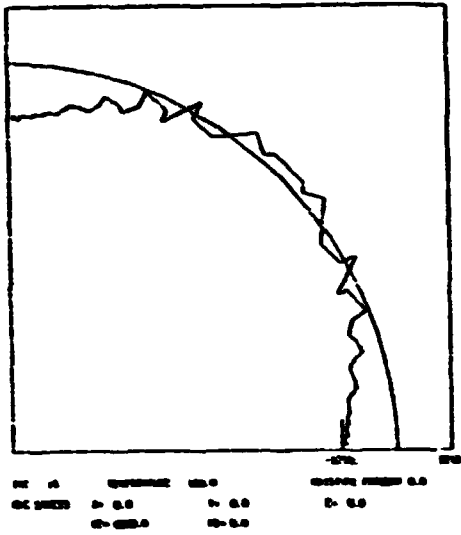
Figure E15. AS4/3502 Graphite/Epoxy Unidirectional Composite, Room Temperature, Dry (RTD); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.



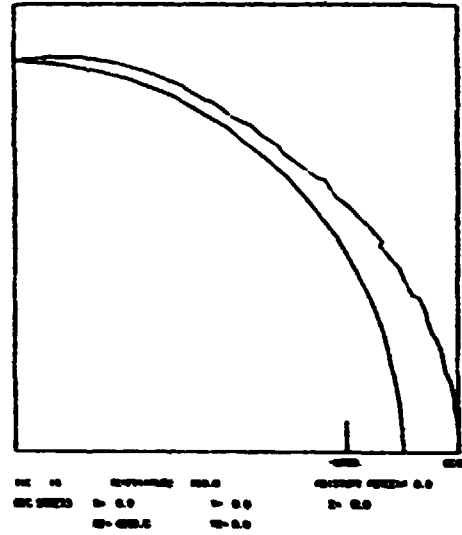
e) Interface Normal Stress (psi)

f) Interface Shear Stress (psi)

Figure E15 (continued). AS4/3502 Graphite/Epoxy Unidirectional Composite, Room Temperature, Dry (RTD); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.

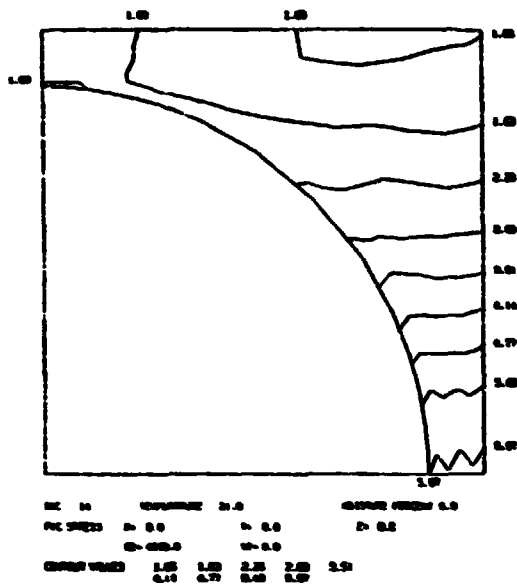


e) Interface Normal Stress (psi)

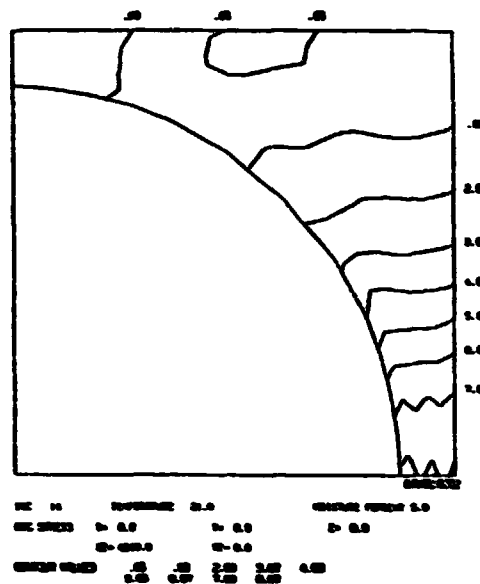


f) Interface Shear Stress (psi)

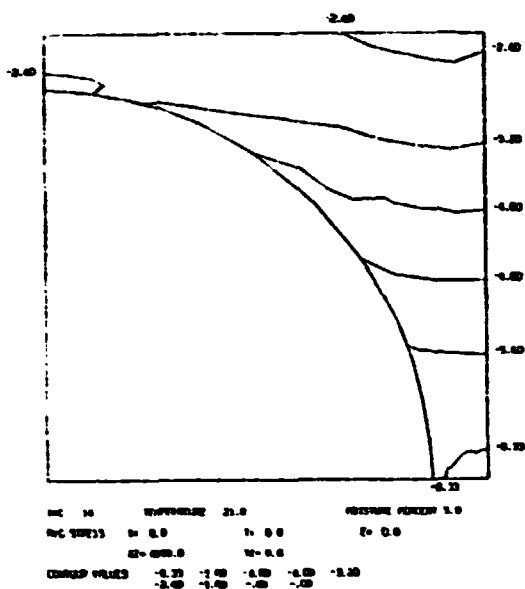
Figure E16 (continued). AS4/3502 Graphite/Epoxy Unidirectional Composite, 100 °C, DRY (ETD); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.



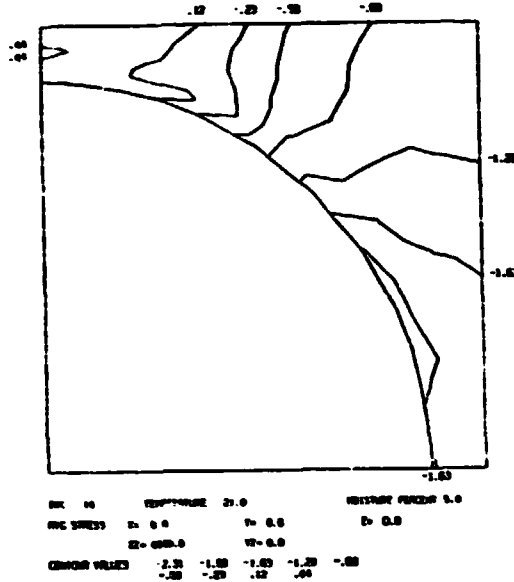
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

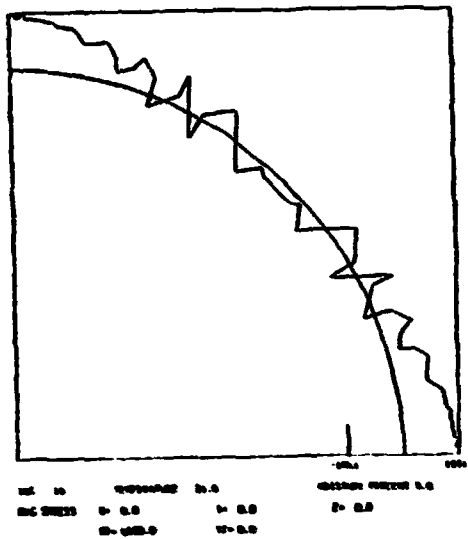


c) Minimum Principal Stress (ksi)

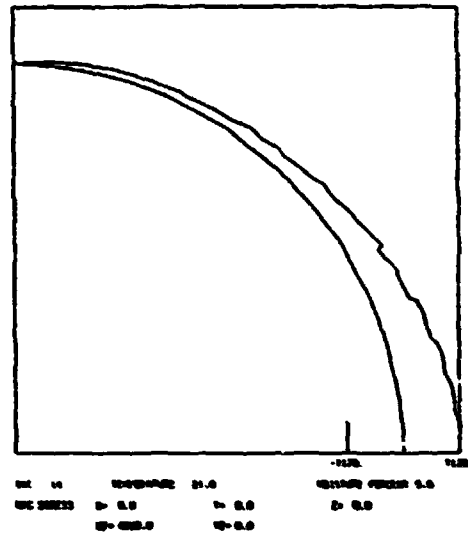


d) Intermediate Principal Stress (ksi)

Figure E17. AS4/3502 Graphite/Epoxy Unidirectional Composite, 21 °C, Wet (RTW); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.

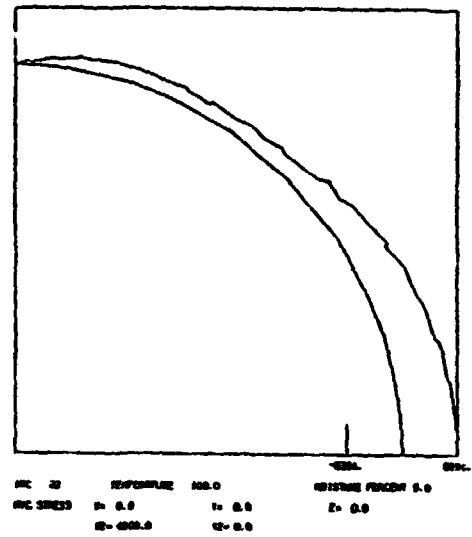
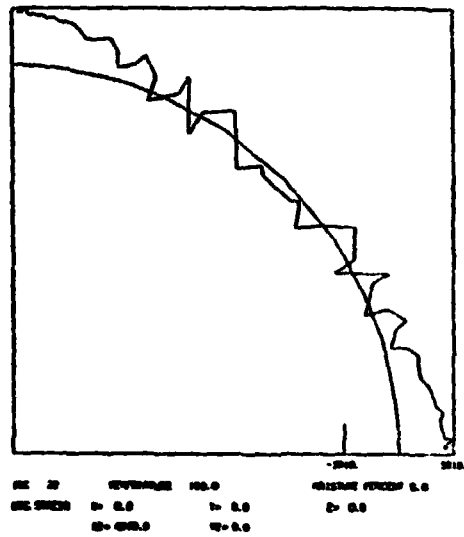


e) Interface Normal Stress (psi)



f) Interface Shear Stress (psi)

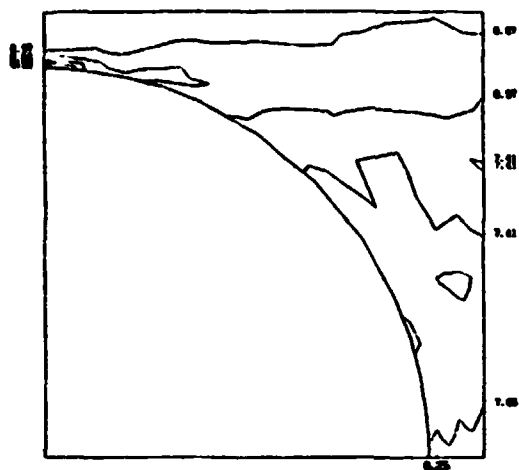
Figure E17 (continued). AS4/3502 Graphite/Epoxy Unidirectional Composite, 21^o C, Wet (RTW); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.



e) Interface Normal Stress (psi)

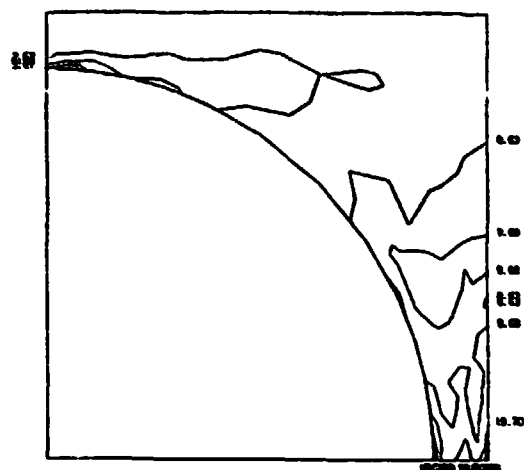
f) Interface Shear Stress (psi)

Figure E18 (continued). AS4/3502 Graphite/Epoxy Unidirectional Composite, 100°C, 5.0 Percent Moisture (ETW); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.



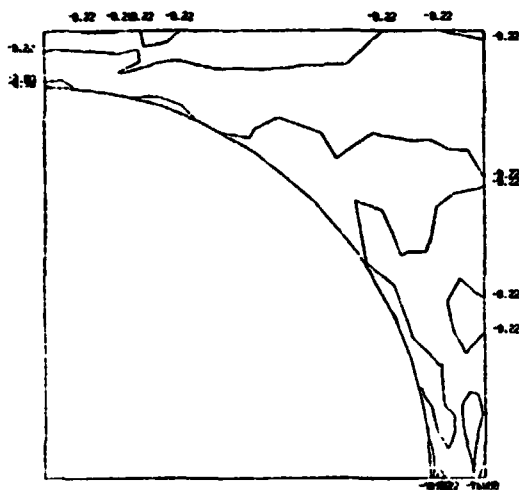
INC 30 TEMPERATURE 100.0 MOISTURE PERCENT 5.0
 AVG STRESS S= 0.0 T= 0.0 Z= 0.0
 ST= 0000.0 WT= 0.0
 COORDINATE VALUES 4.02 5.22 5.02 0.10 0.53
 0.07 7.61 7.05 0.25

a) Octahedral Shear Stress (ksi)



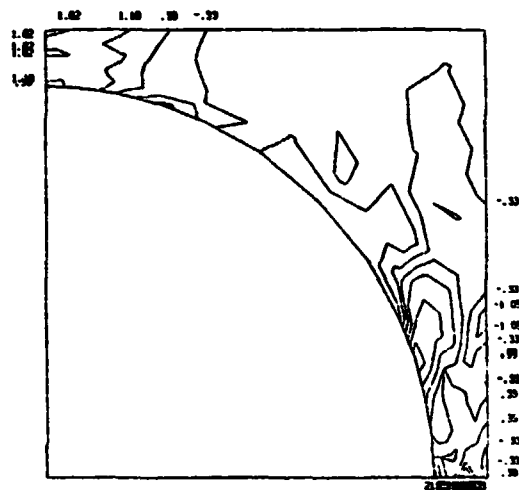
INC 30 TEMPERATURE 100.0 MOISTURE PERCENT 5.0
 AVG STRESS S= 0.0 T= 0.0 Z= 0.0
 ST= 0000.0 WT= 0.0
 COORDINATE VALUES 4.97 5.47 0.57 7.57 0.07
 0.00 10.70 11.00 12.75

b) Maximum Principal Stress (ksi)



INC 30 TEMPERATURE 100.0 MOISTURE PERCENT 5.0
 AVG STRESS S= 0.0 T= 0.0 Z= 0.0
 ST= 0000.0 WT= 0.0
 COORDINATE VALUES -10.61 -0.22 -7.00 -0.50 -0.27
 -0.05 0.00 1.32 -1.11

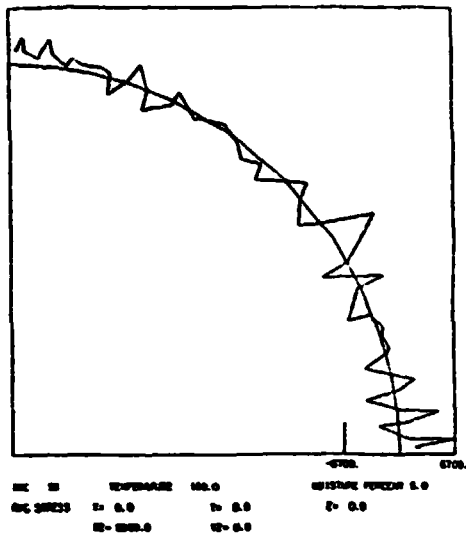
c) Minimum Principal Stress (ksi)



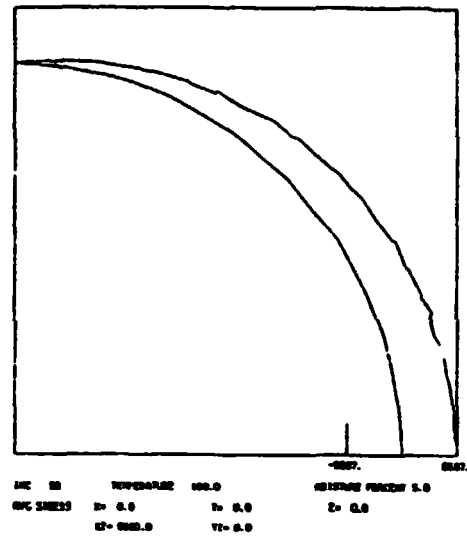
INC 30 TEMPERATURE 100.0 MOISTURE PERCENT 5.0
 AVG STRESS S= 0.0 T= 0.0 Z= 0.0
 ST= 0000.0 WT= 0.0
 COORDINATE VALUES -2.62 -1.70 -1.05 -1.50 -1.30
 2.10 1.07 2.50 0.10

d) Intermediate Principal Stress (ksi)

Figure E19. AS4/3502 Graphite/Epoxy Unidirectional Composite, 100°C, 5.0 Percent Moisture (ETW); 97 MPa (14 ksi) Longitudinal Shear Applied Stress.



e) Interface Normal Stress (psi)

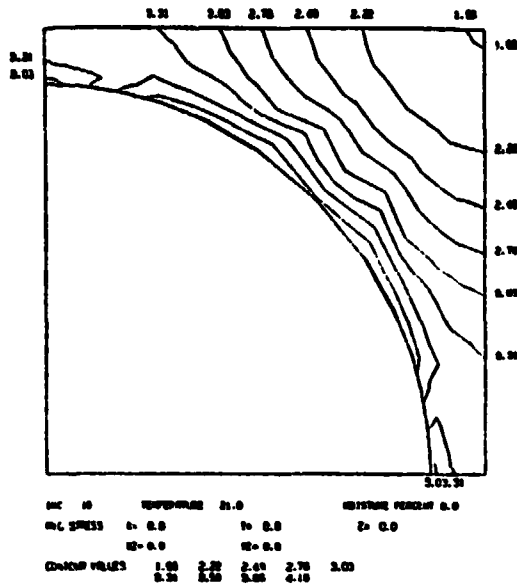


f) Interface Shear Stress (psi)

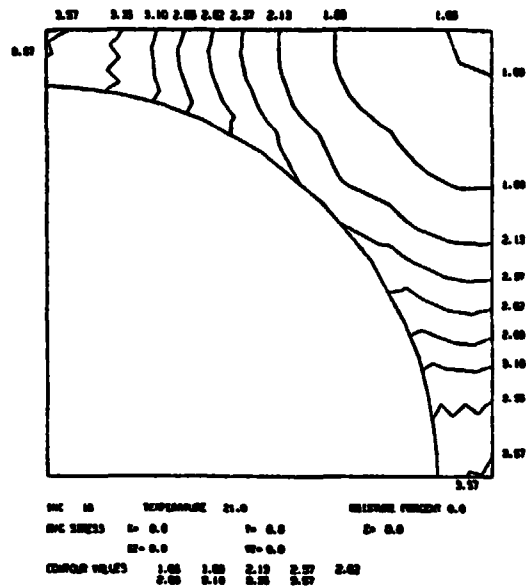
Figure E19 (continued). AS4/3502 Graphite/Epoxy Unidirectional Composite, 100°C, 5.0 Percent Moisture (ETW); 97 MPa (14 ksi) Longitudinal Shear Applied Stress.

APPENDIX E2

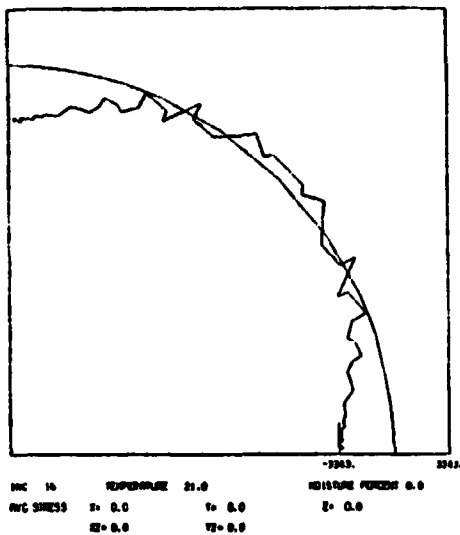
AS4/2220-3 GRAPHITE/EPOXY UNIDIRECTIONAL COMPOSITE



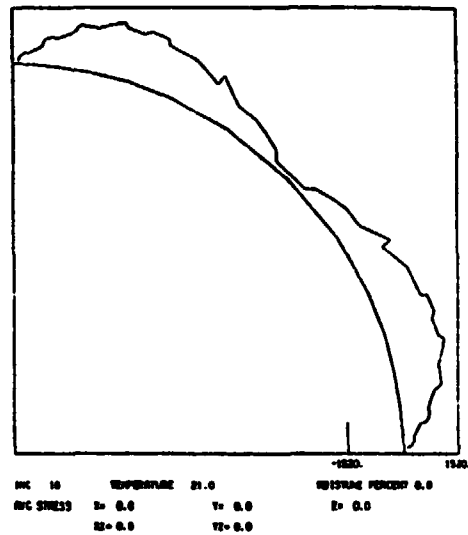
e) Intermediate Principal Stress (ksi)



f) Maximum Shear Stress (ksi)

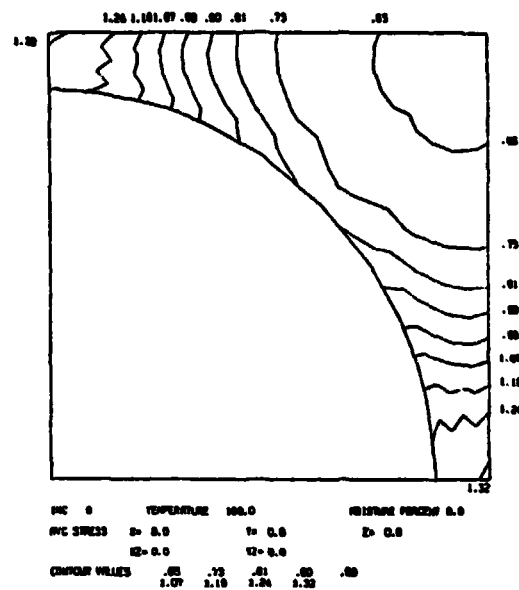
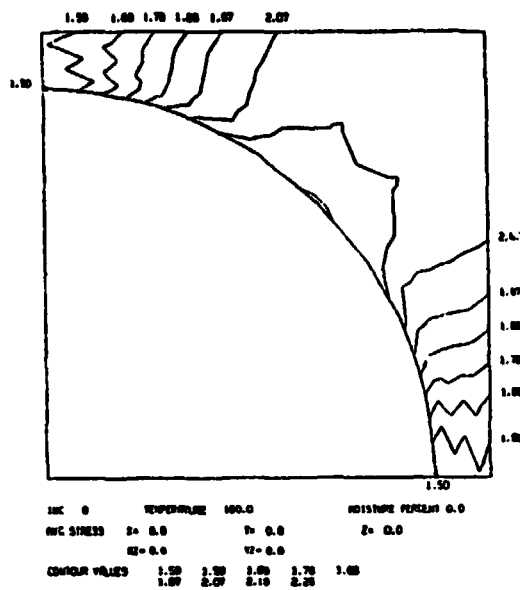


g) Interface Normal Stress (psi)

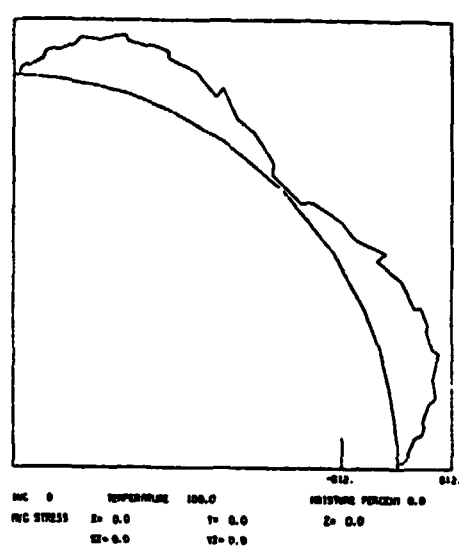
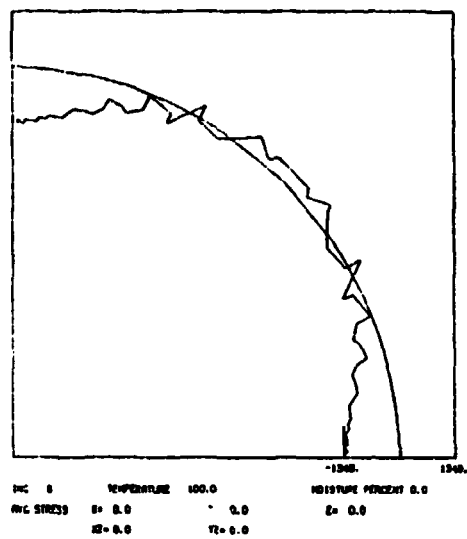


h) Interface Shear Stress (psi)

Figure E20 (continued). AS4/2220-3 Graphite/Epoxy Unidirectional Composite, Room Temperature, Dry (RTD); No Mechanical Loading.

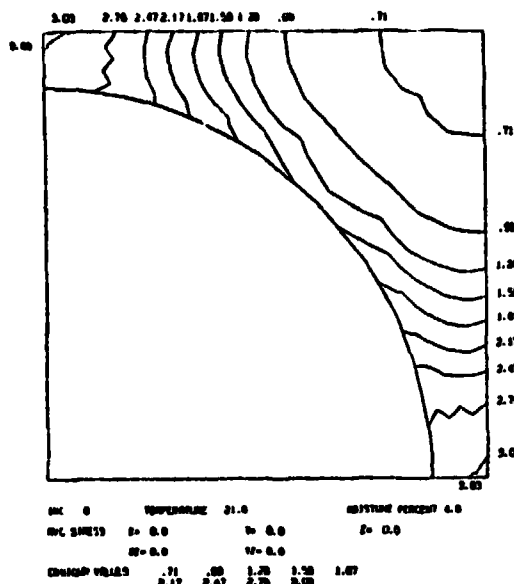


a) Octahedral Shear Stress (ksi) b) Maximum Principal Stress (ksi)

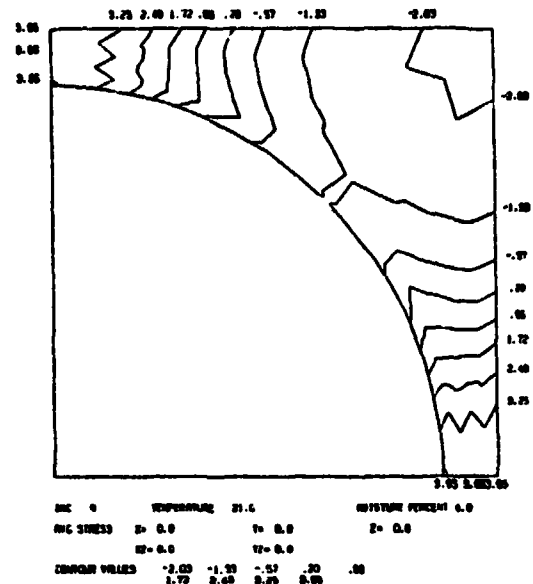


c) Interface Normal Stress (psi) d) Interface Shear Stress (psi)

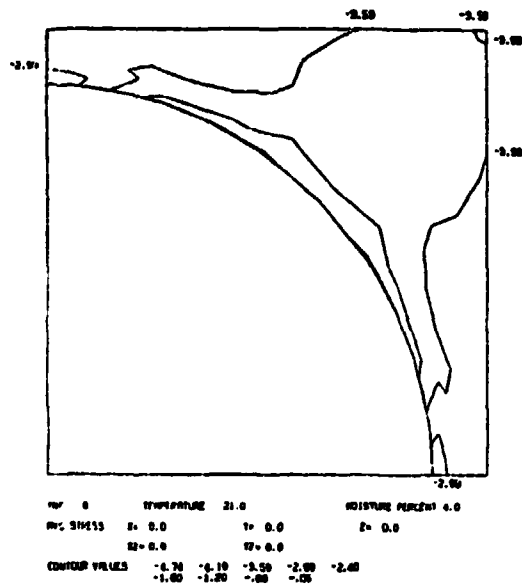
Figure E21. AS4/2220-3 Graphite/Epoxy Unidirectional Composite, 100°C, Dry (ETD); No Mechanical Loading.



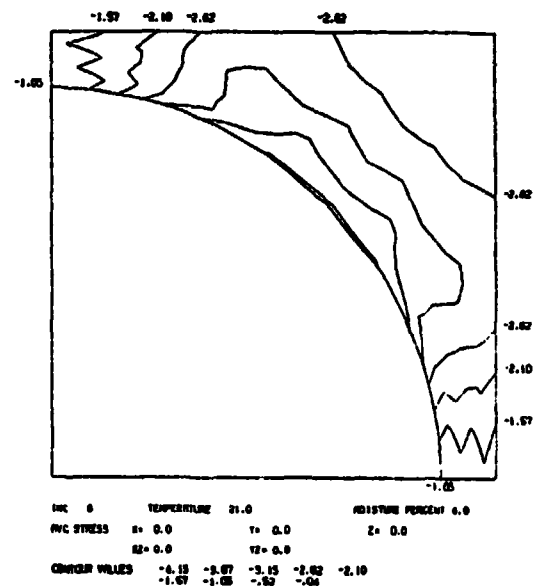
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

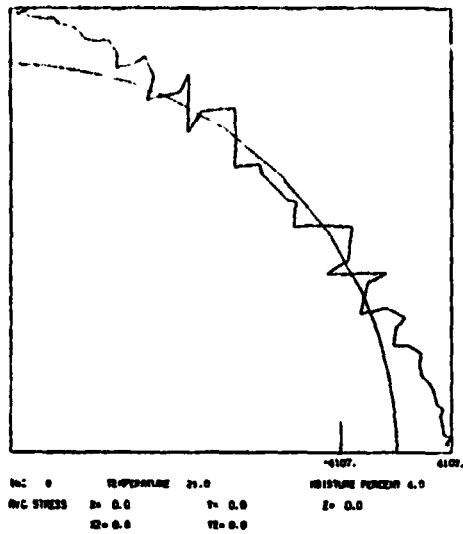


c) Maximum Principal Stress (ksi)

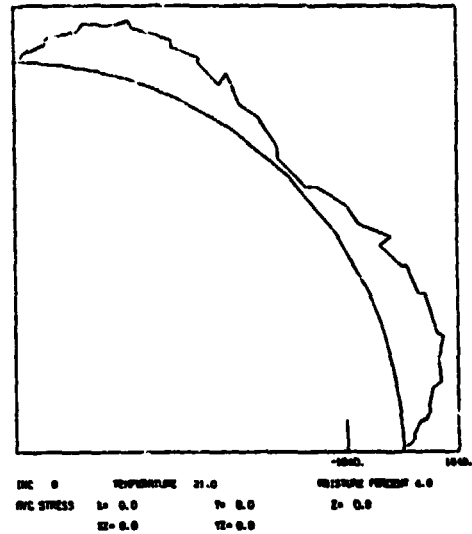


d) Intermediate Principal Stress (ksi)

Figure E22. AS4/2220-3 Graphite/Epoxy Unidirectional Composite, Room Temperature, 4.0 Percent Moisture (RTW); No Mechanical Loading.

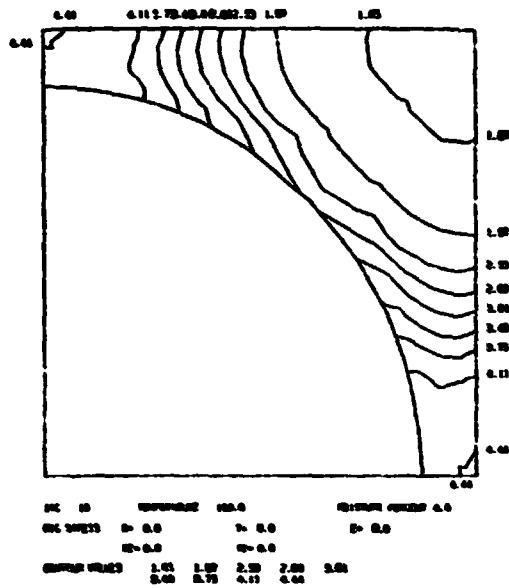


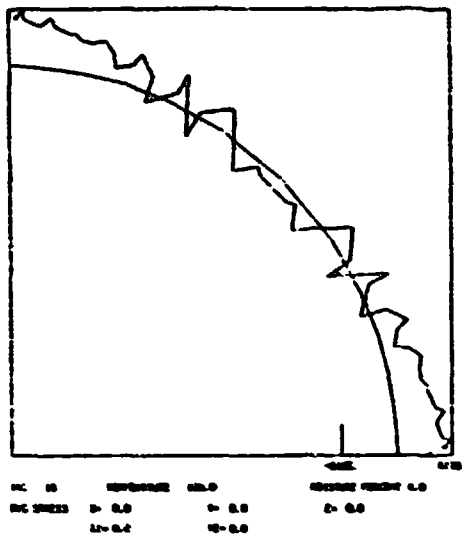
e) Interface Normal Stress (psi)



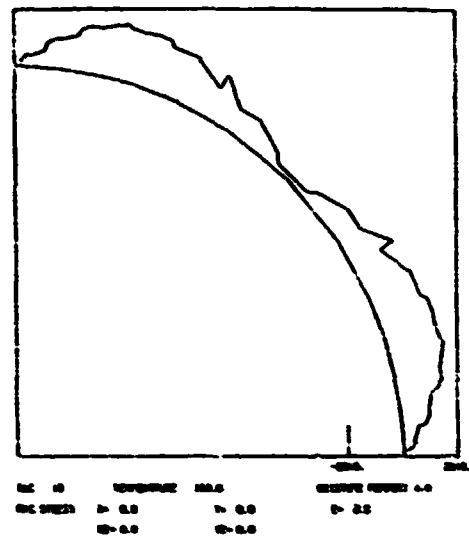
f) Interface Shear Stress (psi)

Figure E22 (continued). AS4/2220-3 Graphite/Epoxy Unidirectional Composite, Room Temperature, 4.0 Percent Moisture (RTW); No Mechanical Loading.





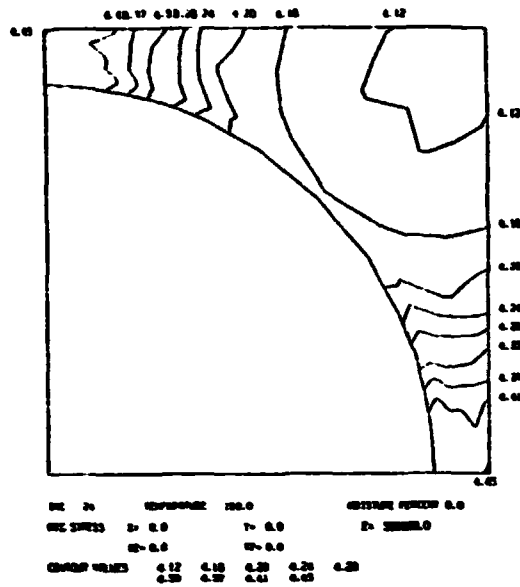
e) Interface Normal Stress (psi)



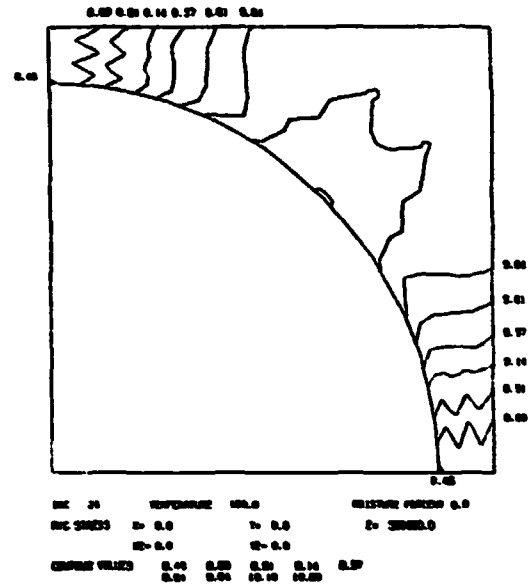
f) Interface Shear Stress (psi)

Figure E23 (continued). AS4/2220-3 Graphite/Epoxy Unidirectional Composite, 100°C, 4.0 Percent Moisture (ETW); No Mechanical Loading.

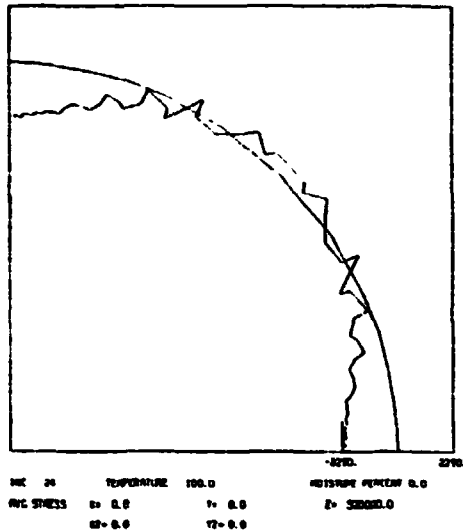
ORIGINAL PLACES
OF POOR QUALITY



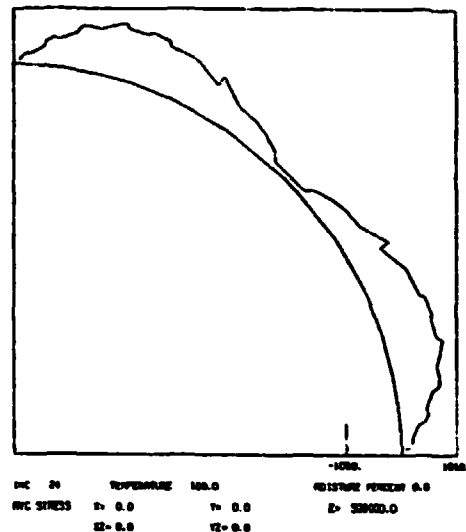
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

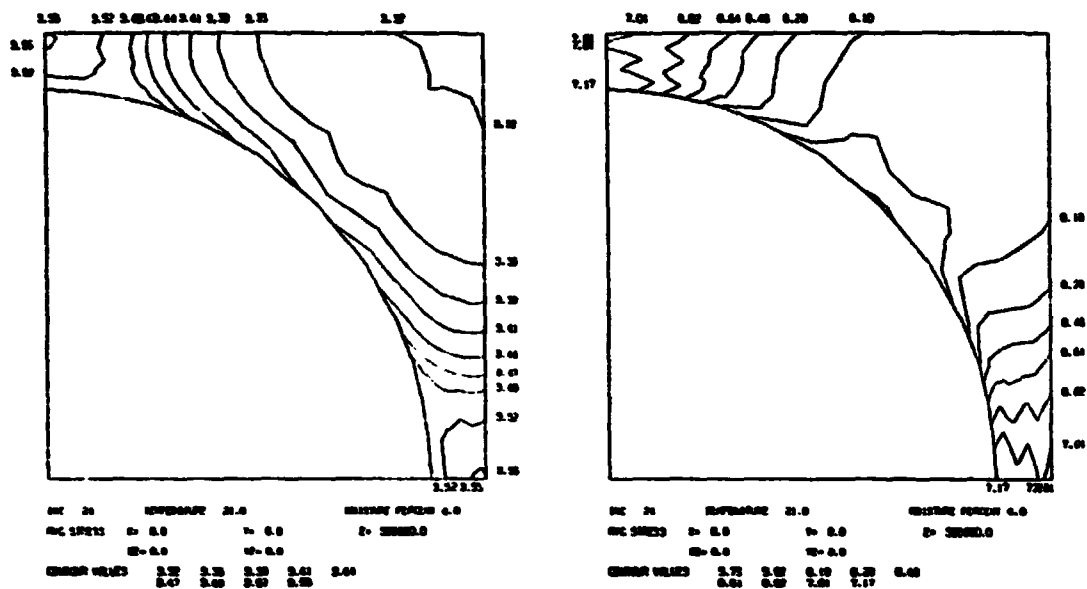


c) Interface Normal Stress (psi)

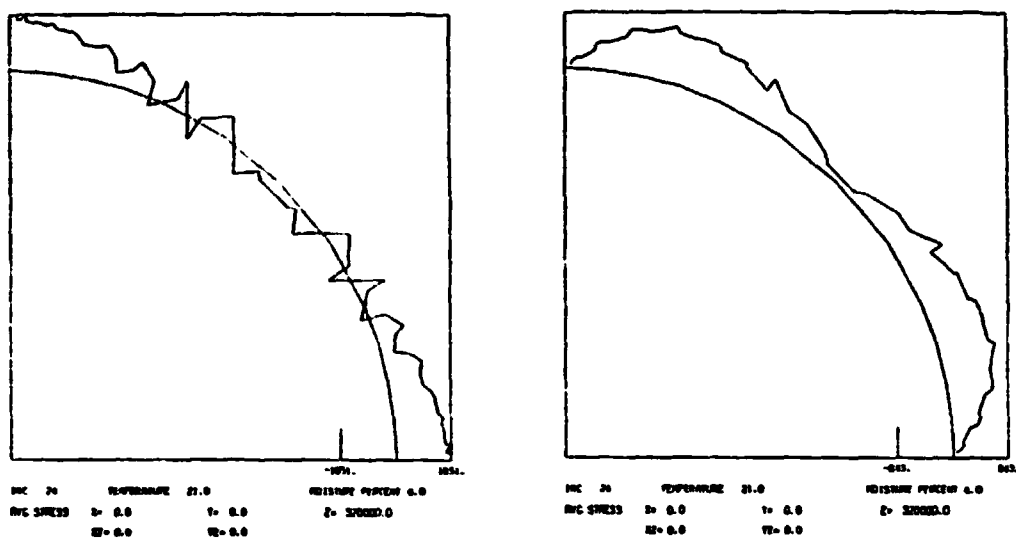


d) Interface Shear Stress (psi)

Figure E25. AS4/2220-3 Graphite/Epoxy Unidirectional Composite, 100 C, Dry (ETD); 2.20 GPa (320 ksi) Longitudinal Tensile Applied Stress.

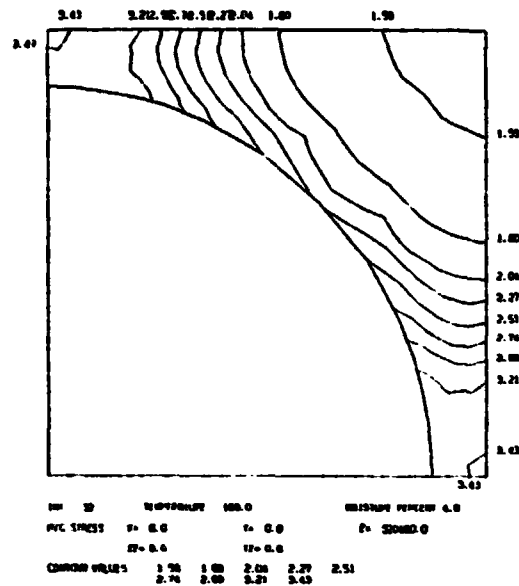


a) Octahedral Shear Stress (ksi) b) Maximum Principal Stress (ksi)

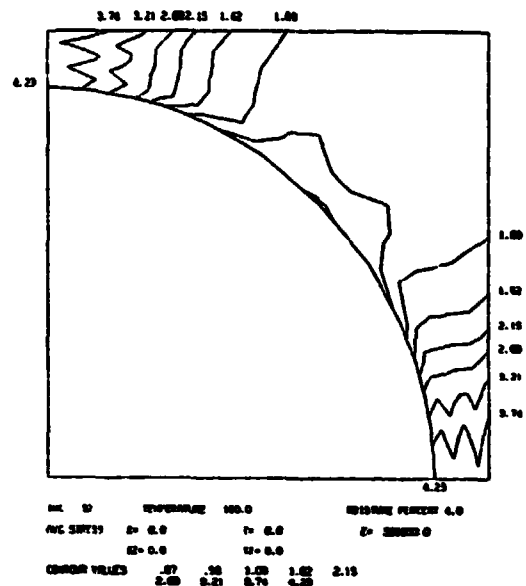


c) Interface Normal Stress (psi) d) Interface Shear Stress (psi)

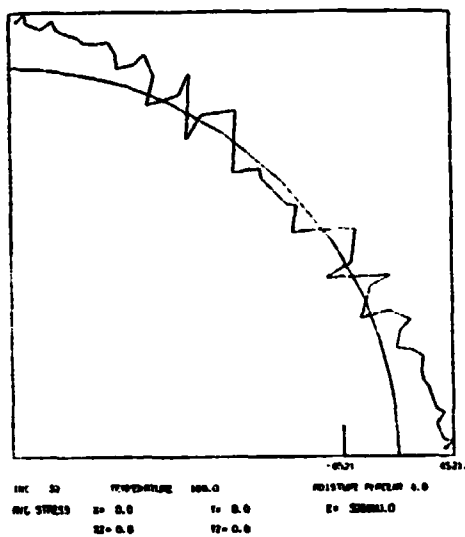
Figure E26. AS4/2220-3 Graphite/Epoxy Unidirectional Composite, Room Temperature, 4.0 Percent Moisture (RTW); 2.20 GPa (320 ksi) Longitudinal Tensile Applied Stress.



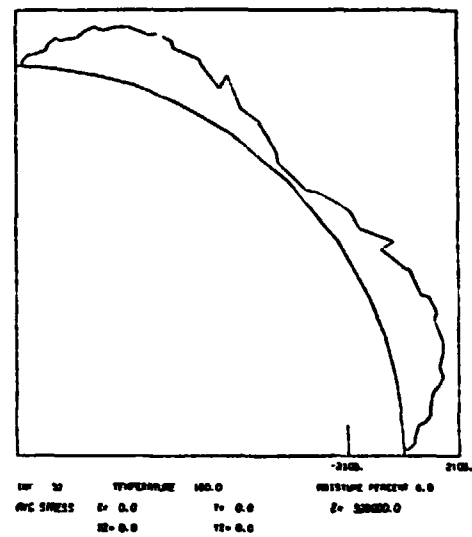
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)



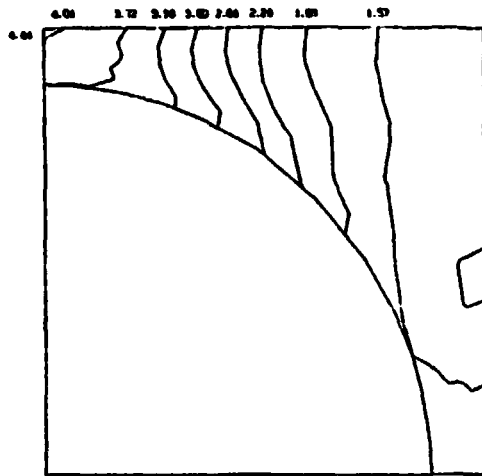
c) Interface Normal Stress (psi)



d) Interface Shear Stress (psi)

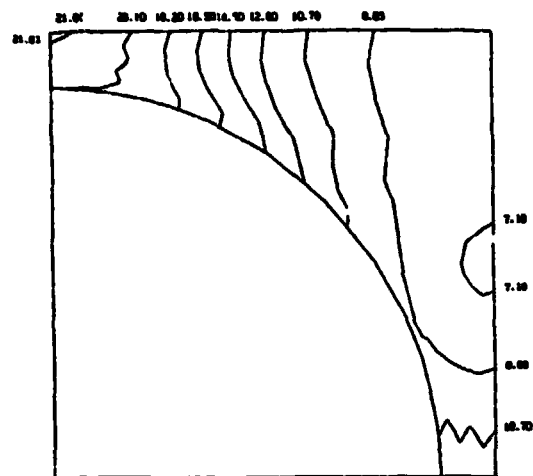
Figure E28. AS4/2220-3 Graphite/Epoxy Unidirectional Composite, 100°C
4.0 Percent Moisture (FTW); 2.20 GPa (320 ksi) Longitudinal
Tensile Applied Stress.

ORIGINAL FILE
OF POOR QUALITY



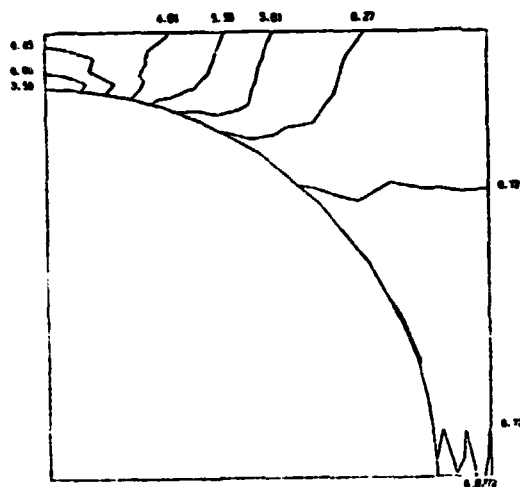
INC 6 TEMPERATURE 21.0 MOISTURE PERCENT 0.0
AVG STRESS S= 4000.0 T= 0.0 Z= 0.0
S2= 0.0 S7= 0.0
CONTOUR VALUES 1.26 1.57 1.88 2.20 2.54
0.50 0.89 0.72 0.00

a) Octetedral Shear Stress (ksi)



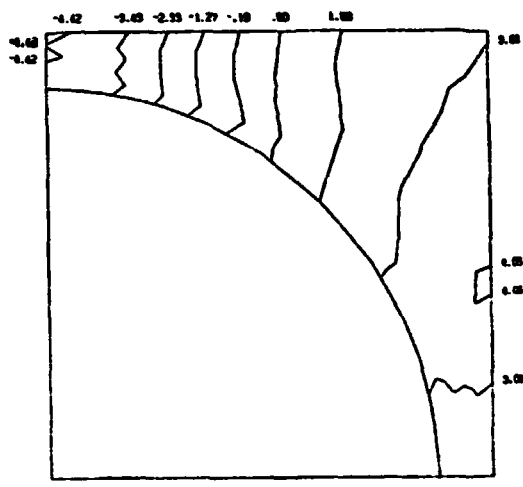
INC 6 TEMPERATURE 21.0 MOISTURE PERCENT 0.0
AVG STRESS S= 4000.0 T= 0.0 Z= 0.0
S2= 0.0 S7= 0.0
CONTOUR VALUES 7.16 0.89 10.70 12.00 14.30
10.00 10.00 10.00 10.00

b) Maximum Principal Stress (ksi)



INC 6 TEMPERATURE 21.0 MOISTURE PERCENT 0.0
AVG STRESS S= 4000.0 T= 0.0 Z= 0.0
S2= 0.0 S7= 0.0
CONTOUR VALUES 2.50 4.00 4.61 4.91 5.36
0.01 0.27 0.72 7.16

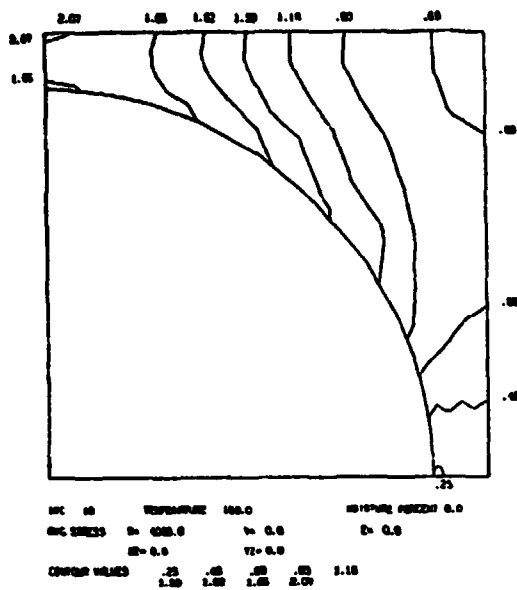
c) Minimum Principal Stress (ksi)



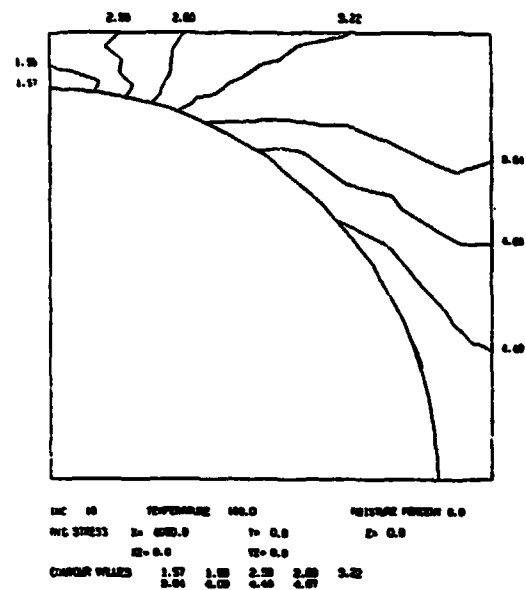
INC 6 TEMPERATURE 21.0 MOISTURE PERCENT 0.0
AVG STRESS S= 4000.0 T= 0.0 Z= 0.0
S2= 0.0 S7= 0.0
CONTOUR VALUES -0.42 -0.42 -2.35 -1.27 -1.10
-0.00 1.00 0.00 0.00

d) Intermediate Principal Stress (ksi)

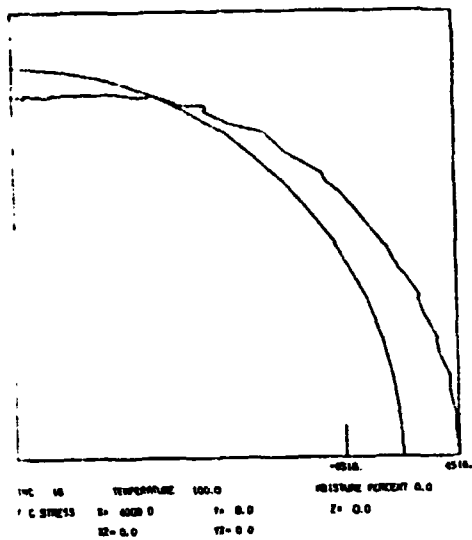
Figure E29. AS4/2220-3 Graphite/Epoxy Unidirectional Composite, Room Temperature, Dry (RTD); 27.6 MPa (4 ksi) Transverse Tensile Applied Stress.



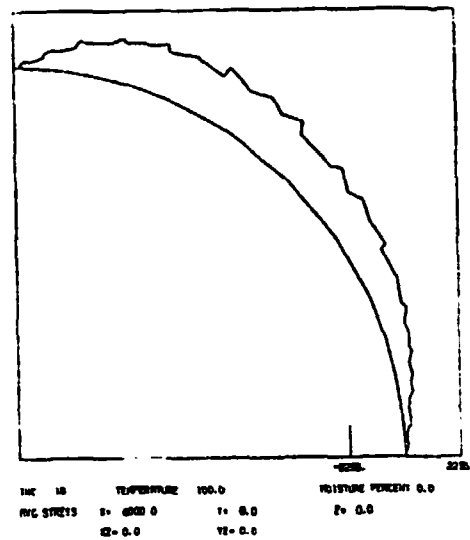
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

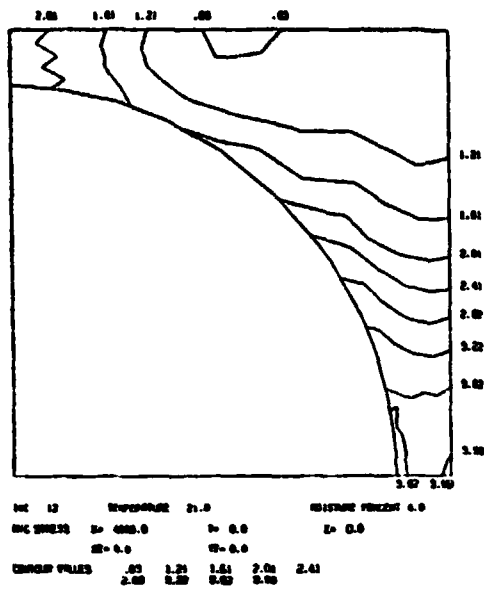


c) Interface Normal Stress (psi)

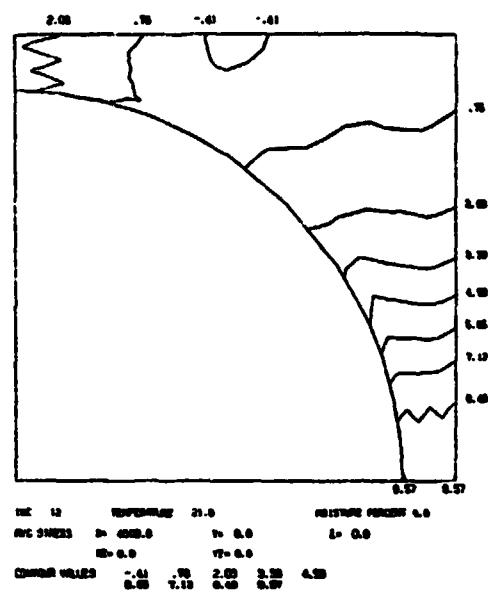


d) Interface Shear Stress (psi)

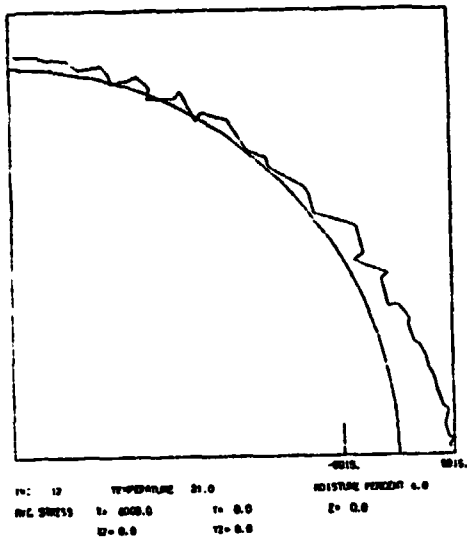
Figure E30. AS4/2220-3 Graphite/Epoxy Unidirectional Composite, 100°C, Dry (ETD); 27.6 MPa (4 ksi) Transverse Tensile Applied Stress.



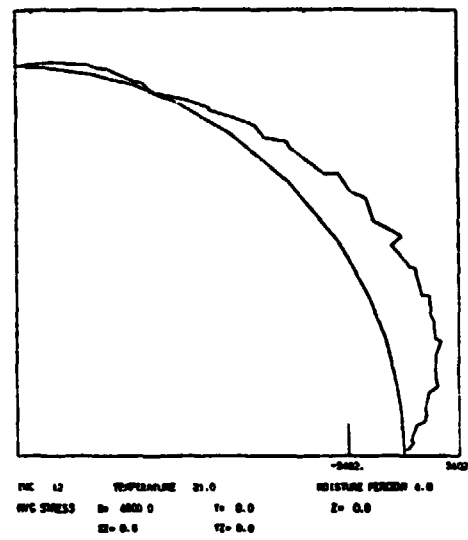
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

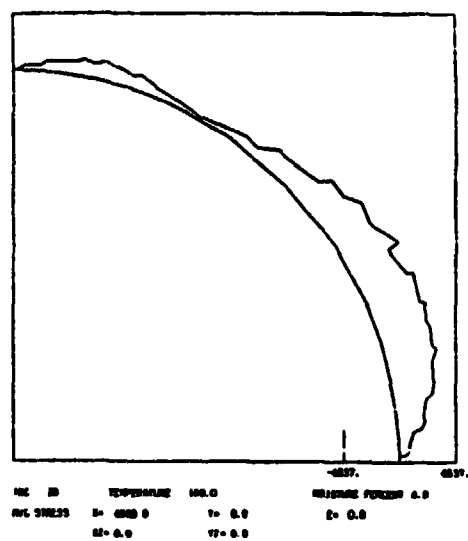
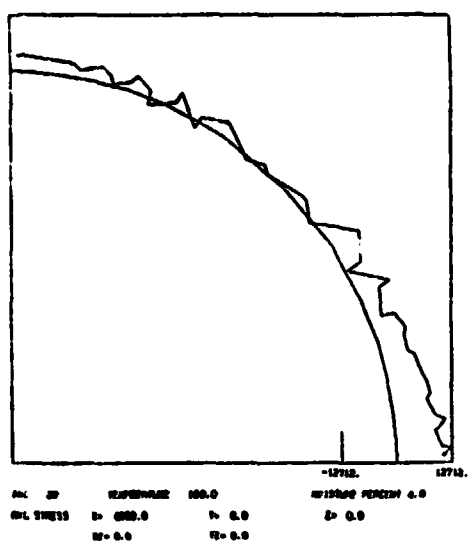


c) Interface Normal Stress (psi)



d) Interface Shear Stress (psi)

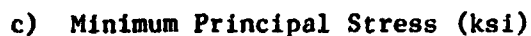
Figure E31. AS4/220-3 Graphite/Epoxy Unidirectional Composite, Room Temperature, 4.0 Percent Moisture (RTW); 27.6 MPa (4 ksi) Transverse Tensile Applied Stress.

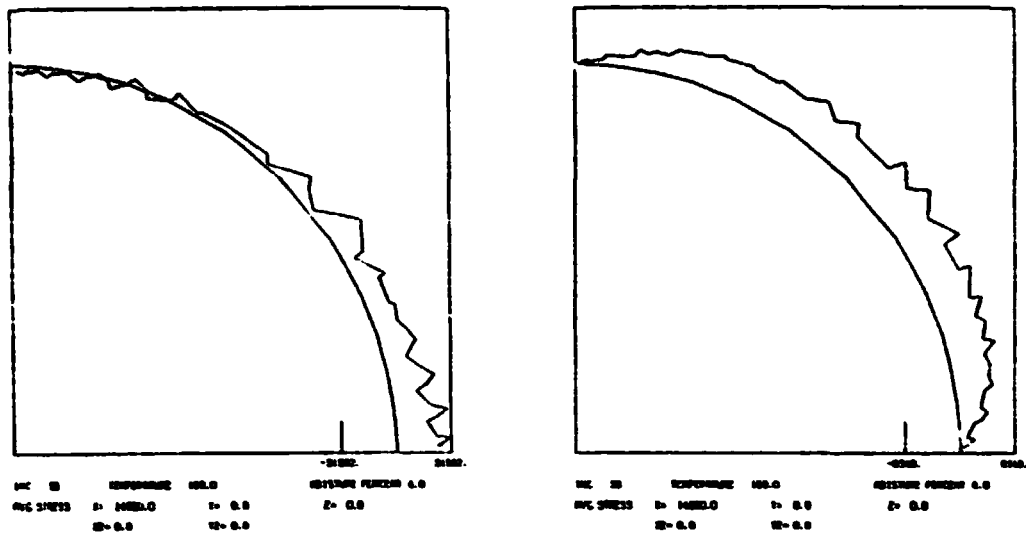


e) Interface Normal Stress (psi)

f) Interface Shear Stress (psi)

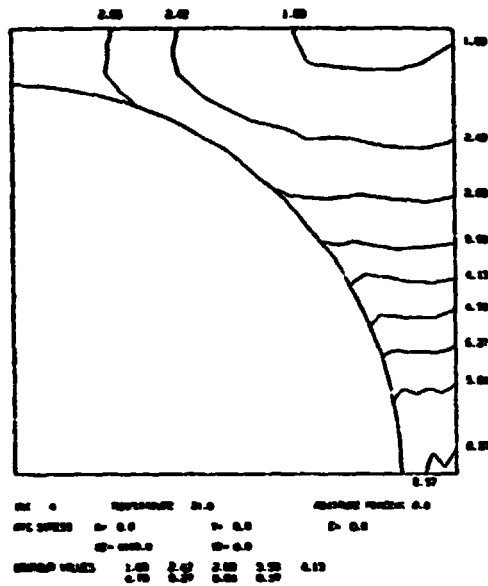
Figure E32 (continued). AS4/2220-3 Graphite/Epoxy Unidirectional Composite, 100°C, 4.0 Percent Moisture (ETW); 27.6 MPa (4 ksi) Transverse Tensile Applied Stress.



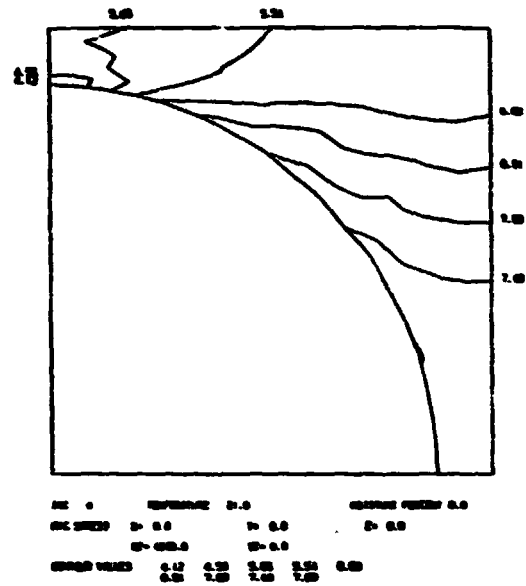


e) Interface Normal Stress (psi) f) Interface Shear Stress (psi)

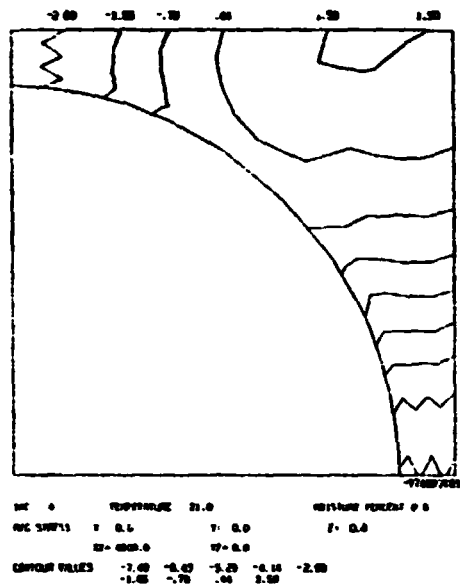
Figure E33 (continued). AS4/2220-3 Graphite/Epoxy Unidirectional Composite, 100°C, 4.0 Percent Moisture (ETW); 97 MPa (14 ksi) Transverse Tensile Applied Stress.



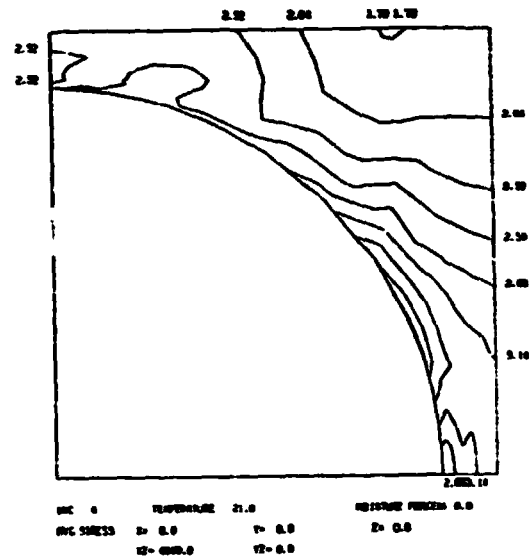
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

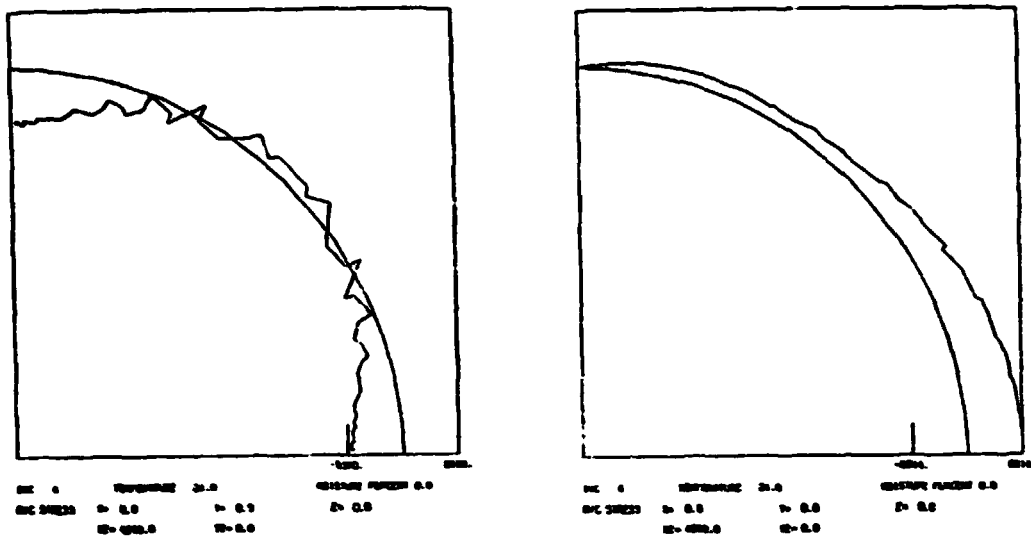


c) Minimum Principal Stress (ksi)



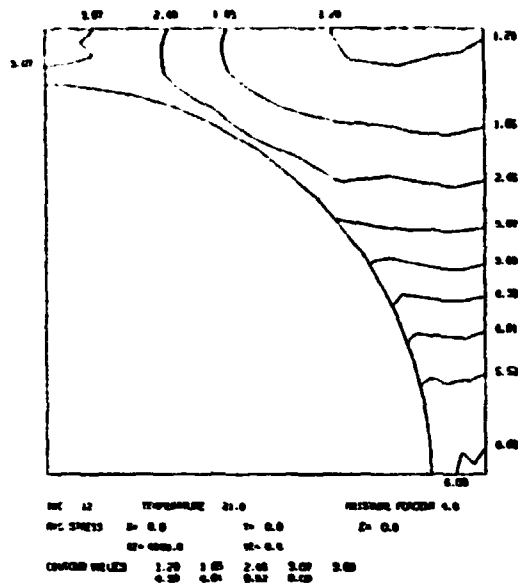
d) Intermediate Principal Stress (ksi)

Figure E34. AS4/2220-3 Graphite/Epoxy Unidirectional Composite, Room Temperature, Dry (RTD); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.

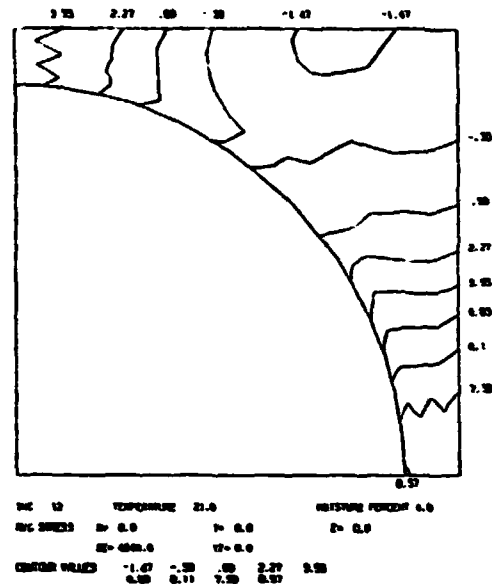


e) Interface Normal Stress (psi) f) Interface Shear Stress (psi)

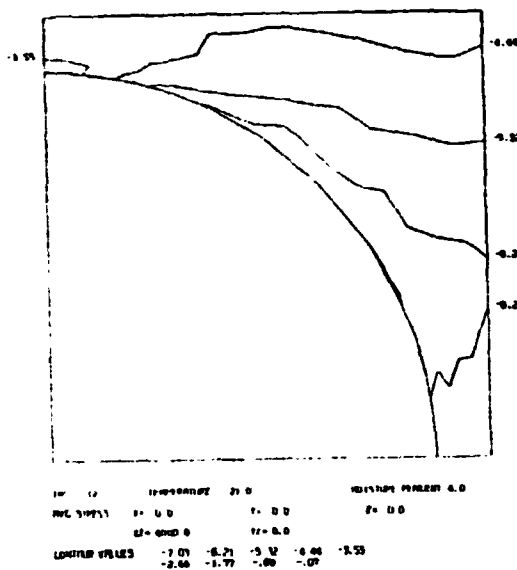
Figure E34 (continued). AS4/2220-3 Graphite/Epoxy Unidirectional Composite, Room Temperature, Dry (RTD); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.



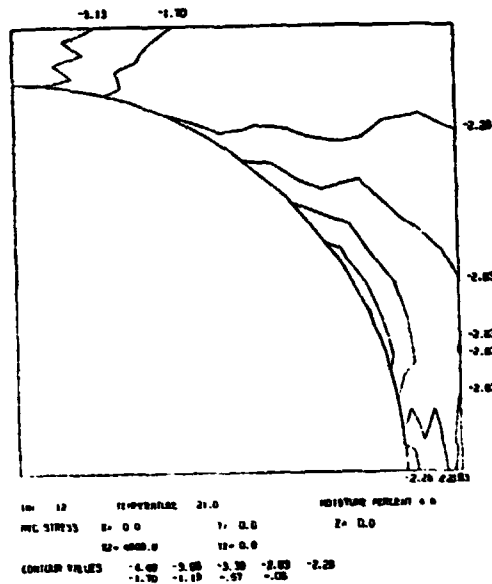
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)



c) Minimum Principal Stress (ksi)



d) Intermediate Principal Stress (ksi)

Figure E36. AS4/2220-3 Graphite/Epoxy Unidirectional Composite, 21°C, 4.0 Percent Moisture (RTW); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.

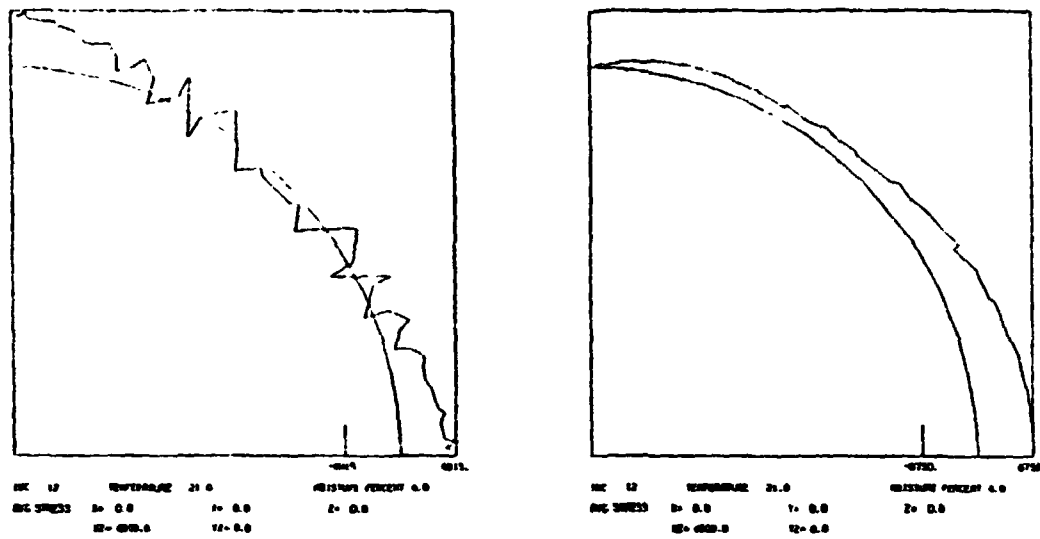
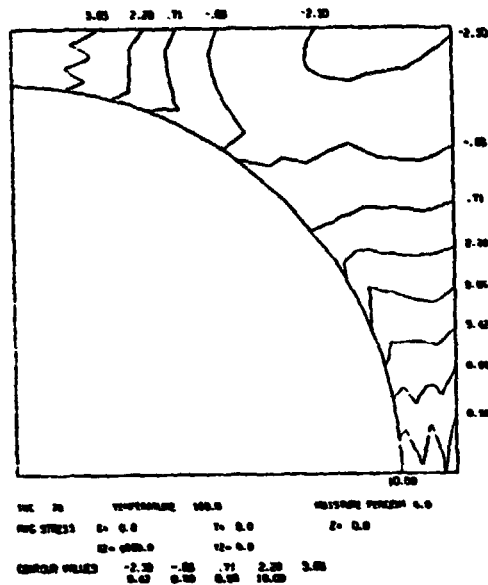
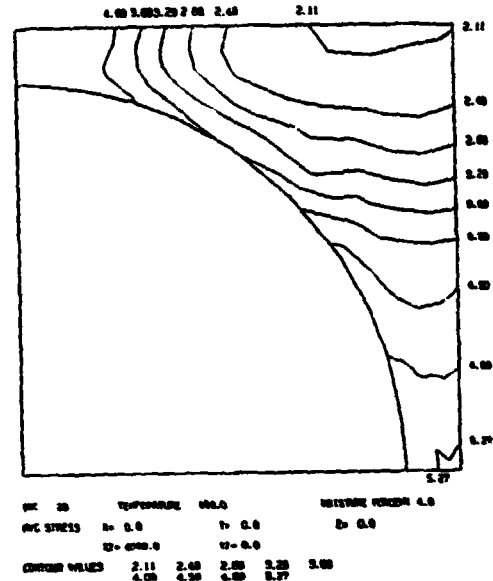


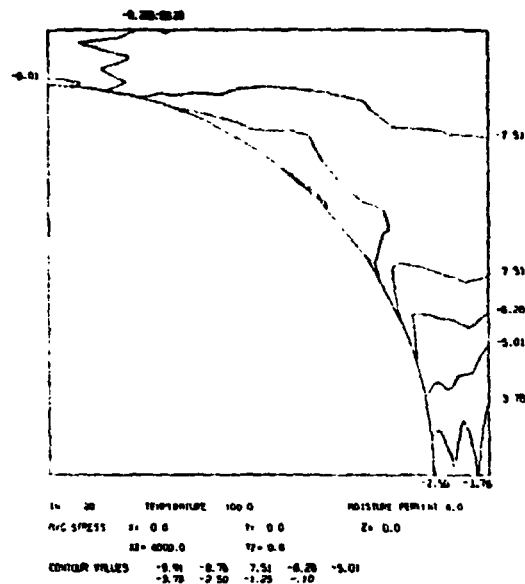
Figure E36 (continued). AS4/2220-3 Graphite/Epoxy Unidirectional Composite, 21°C, 4.0 Percent Moisture (RTW); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.



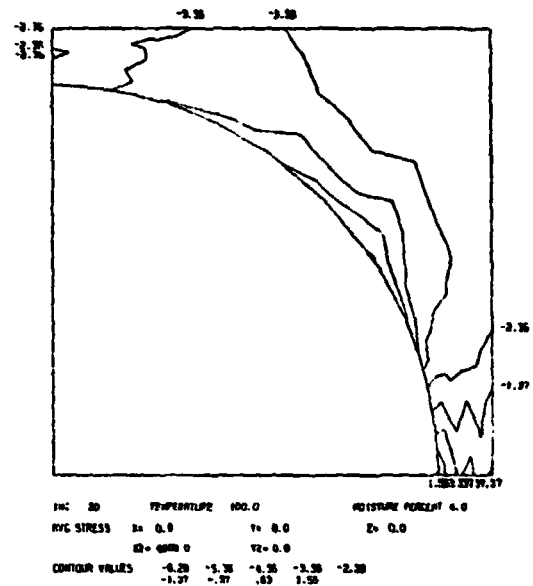
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)



c) Minimum Principal Stress (ksi)

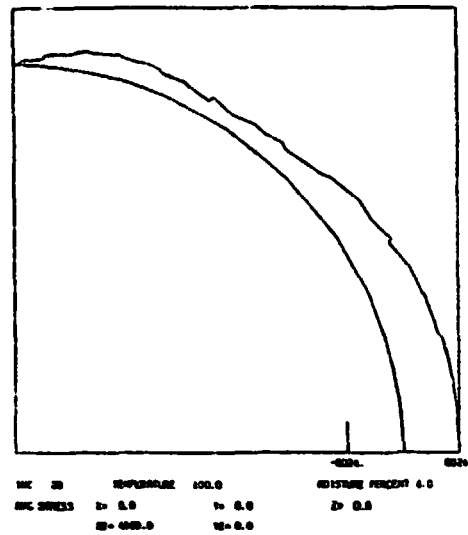


d) Intermediate Principal Stress (ksi)

Figure E37. AS4/2220-3 Graphite/Epoxy Unidirectional Composite, 100°C, 4.0 Percent Moisture (ETW); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.

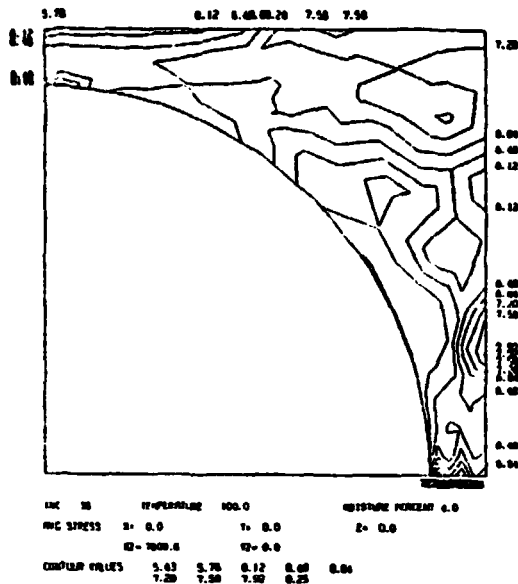


e) Interface Normal Stress (psi)

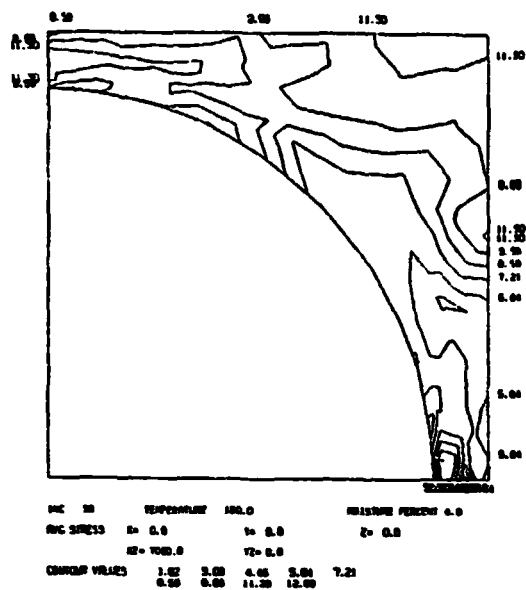


f) Interface Shear Stress (psi)

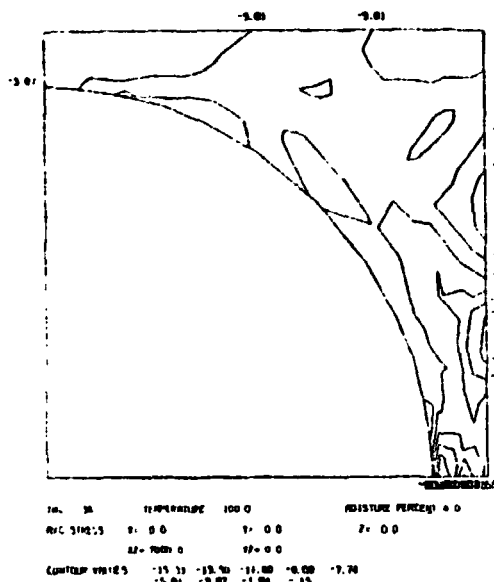
Figure E37 (continued). AS4/2220-3 Graphite/Epoxy Unidirectional Composite, 100°C, 4.0 Percent Moisture (ETW); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.



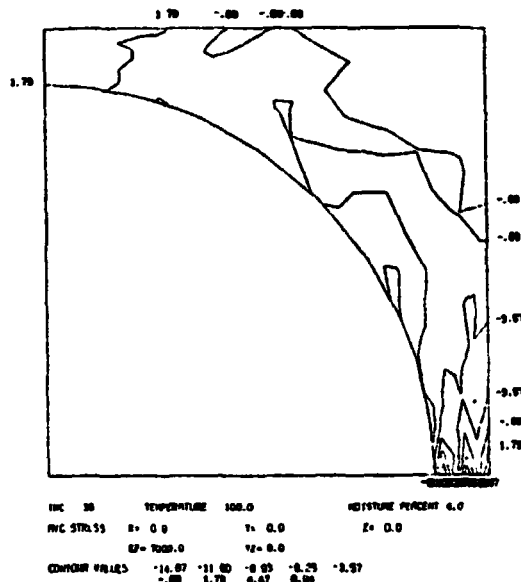
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

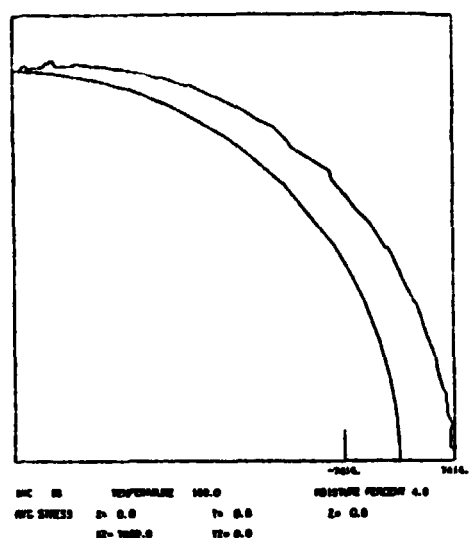
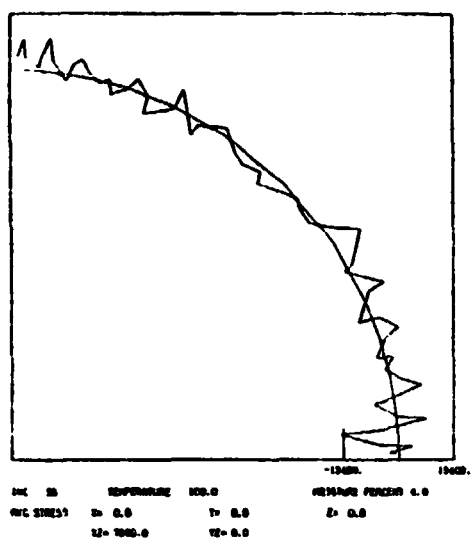


c) Minimum Principal Stress (ksi)



d) Intermediate Principal Stress (ksi)

Figure E38. AS4/2220-3 Graphite/Epoxy Unidirectional Composite, 100°C, 4.0 Percent Moisture (ETW); 97 MPa (14 ksi) Longitudinal Shear Applied Stress.



e) Interface Normal Stress (psi) f) Interface Shear Stress (psi)

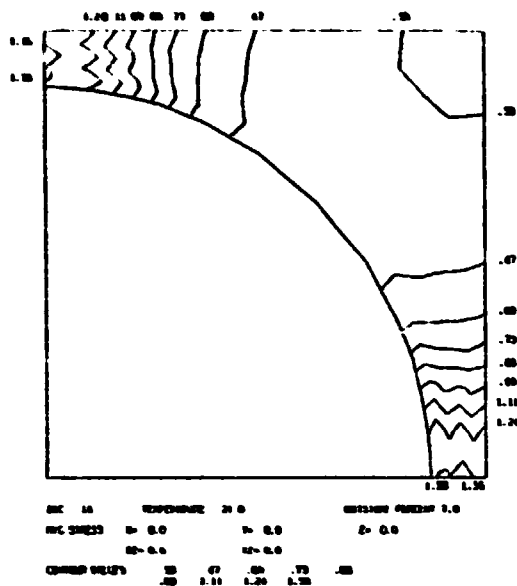
Figure E38 (continued). AS4/2220-3 Graphite/Epoxy Unidirectional Composite, 100°C, 4.0 Percent Moisture (ETW); 27 MPa (14 ksi) Longitudinal Shear Applied Stress.

APPENDIX E3

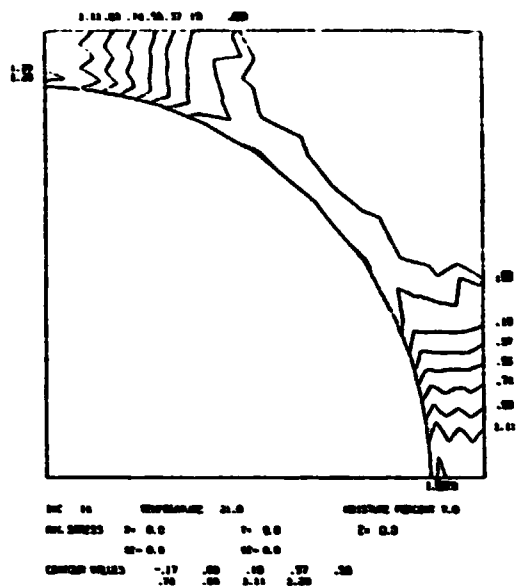
AS4/914 GRAPHITE/EPOXY UNIDIRECTIONAL COMPOSITE



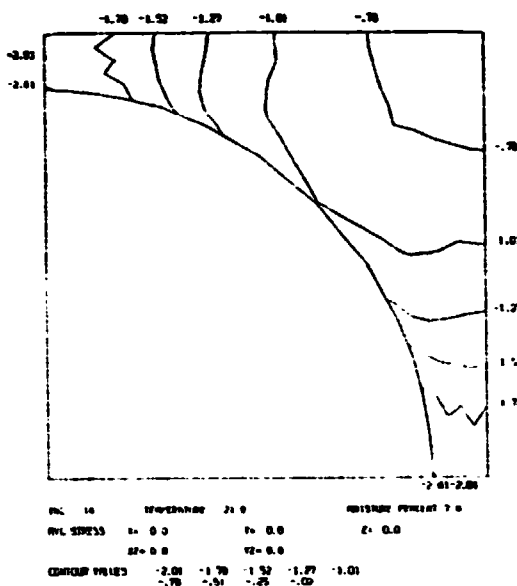
167



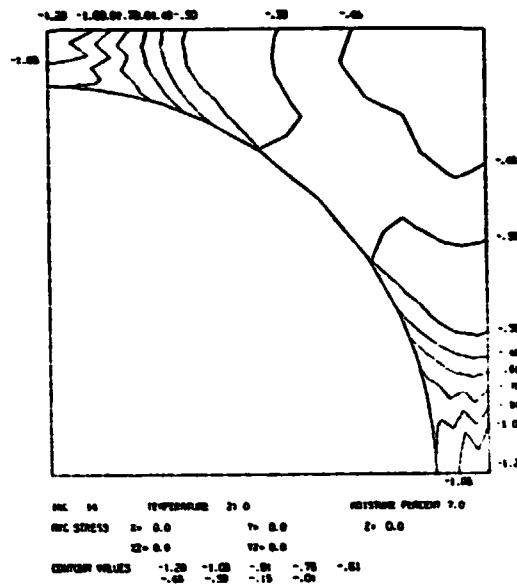
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

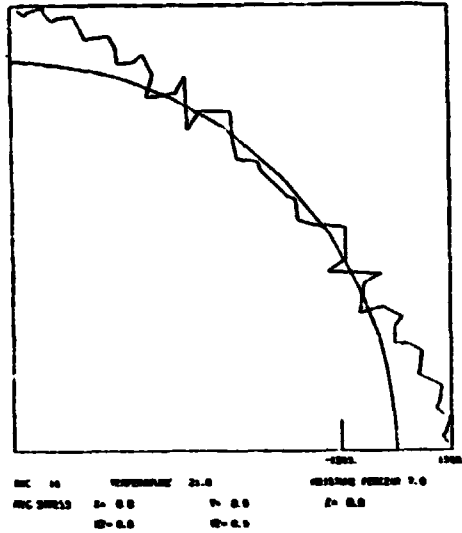


c) Maximum Principal Stress (ksi)

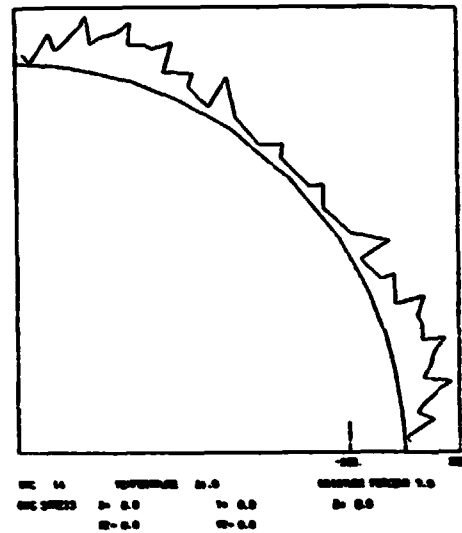


d) Intermediate Principal Stress (ksi)

Figure E41. A 1/4/914 Graphite/Epoxy Unidirectional Composite, Room Temperature, 7.0 Percent Moisture (RTW); No Mechanical Loading.

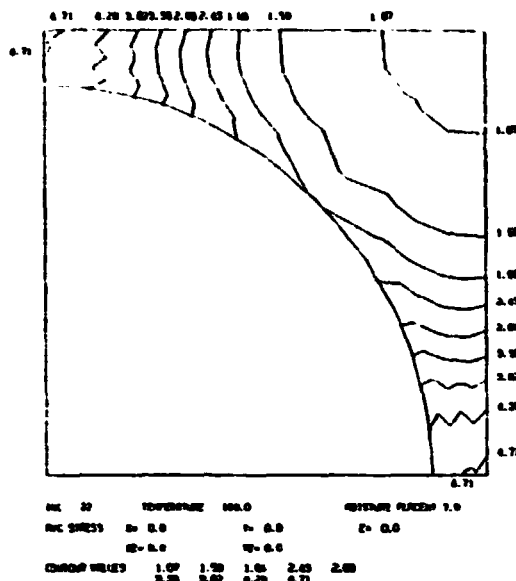


e) Interface Normal Stress (psi)

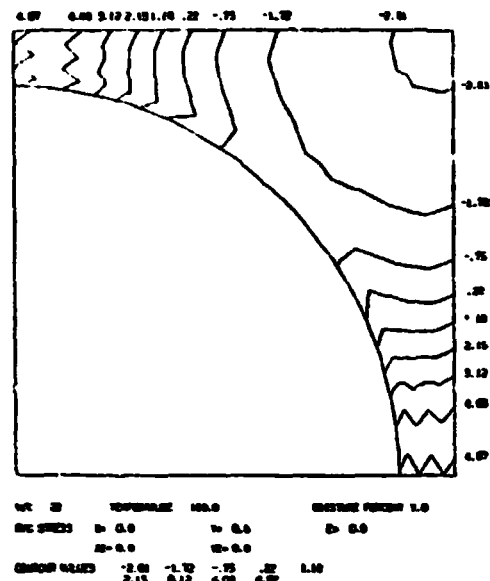


f) Interface Shear Stress (psi)

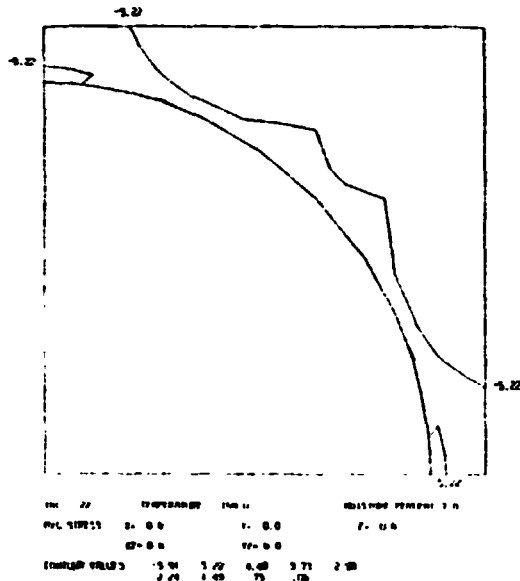
Figure E41 (continued). AS4/914 Graphite/Epoxy Unidirectional Composite, Room Temperature, 7.0 Percent Moisture (RTW); No Mechanical Loading.



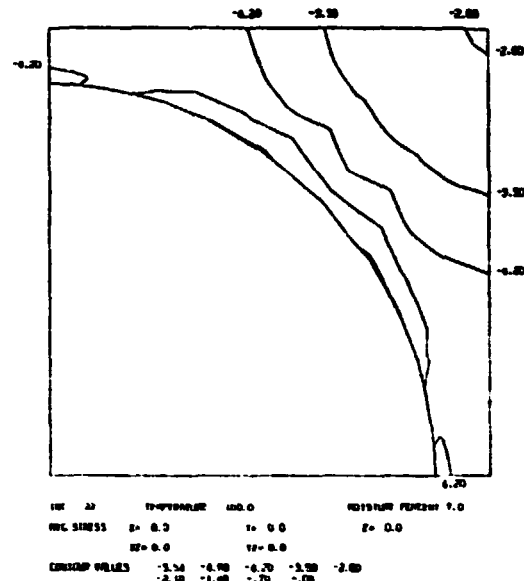
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

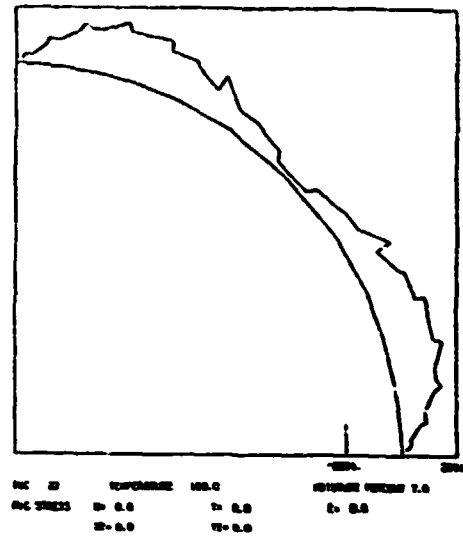


c) Minimum Principal Stress (ksi)



d) Intermediate Principal Stress (ksi)

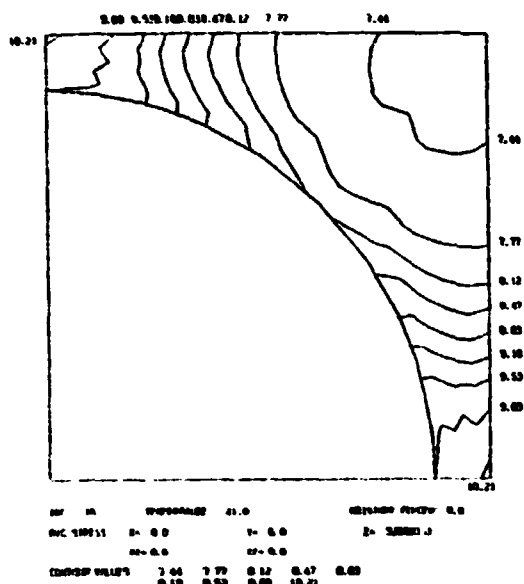
Figure E42. AS4/914 Graphite/Epoxy Unidirectional Composite, 100°C, 7.0 Percent Moisture (ETW); No Mechanical Loading.



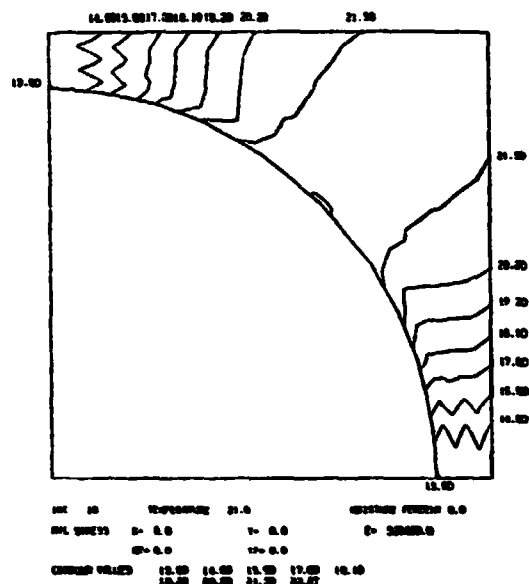
e) Interface Normal Stress (psi)

f) Interface Shear Stress (psi)

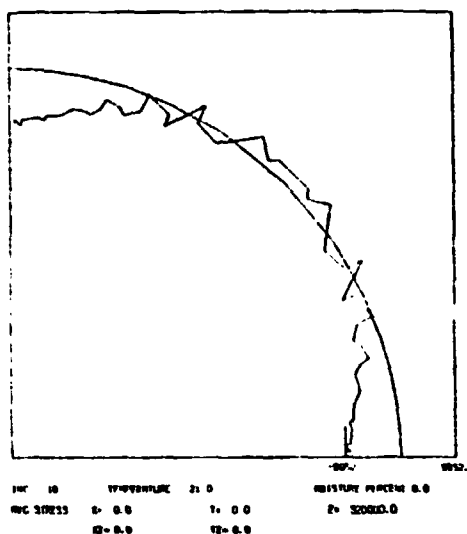
Figure E42 (continued). AS4/914 Graphite/Epoxy Unidirectional Composite, 100°C, 7.0 Percent Moisture (ETW); No Mechanical Loading.



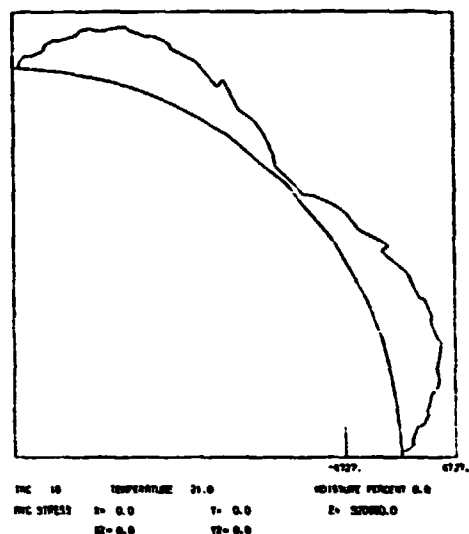
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

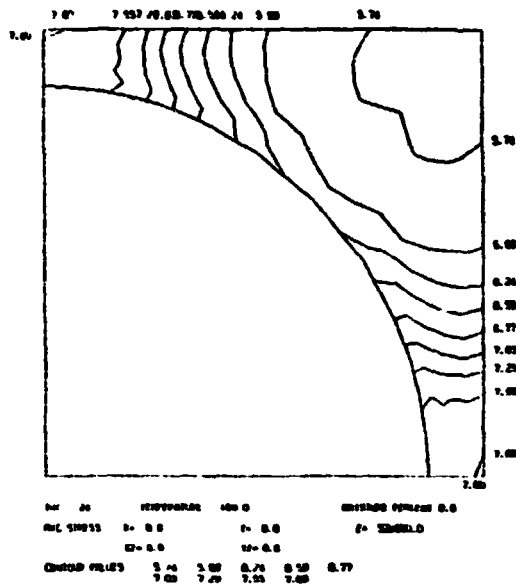


c) Interface Normal Stress (psi)

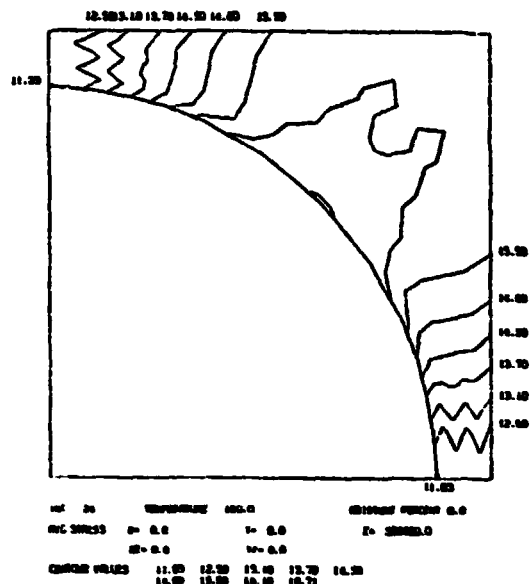


d) Interface Shear Stress (psi)

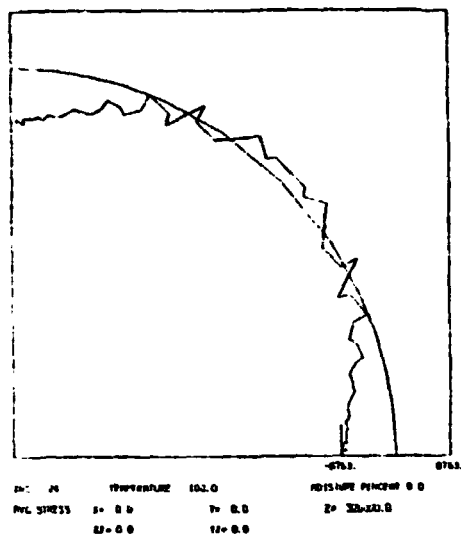
Figure E43. AS4/914 Graphite/Epoxy Unidirectional Composite. Room Temperature, Dry (RTD); 2.20 GPa (320 ksi) Longitudinal Tensile Applied Stress.



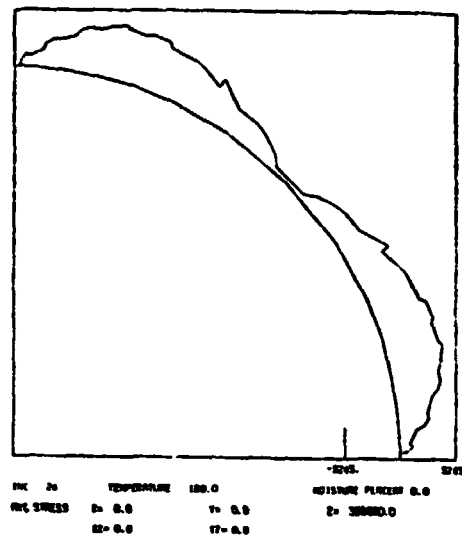
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)



c) Interface Normal Stress (psi)

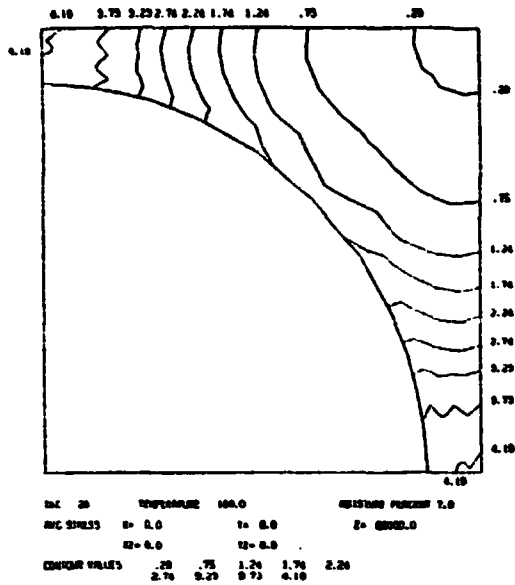


d) Interface Shear Stress (psi)

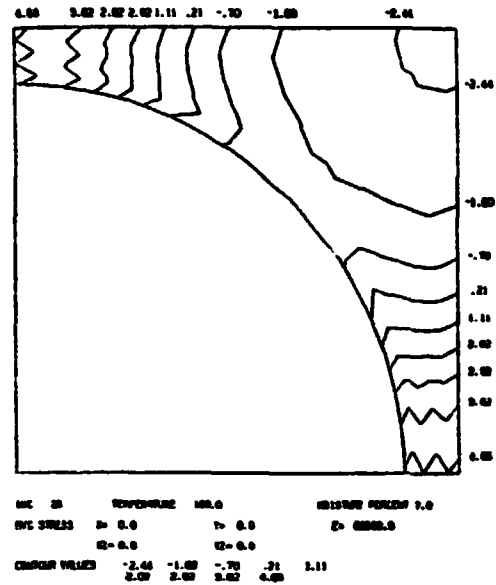
Figure E44. AS4/914 Graphite/Epoxy Unidirectional Composite, 100°C, Dry (ETD); 2.20 GPa (320 ksi) Longitudinal Tensile Applied Stress.



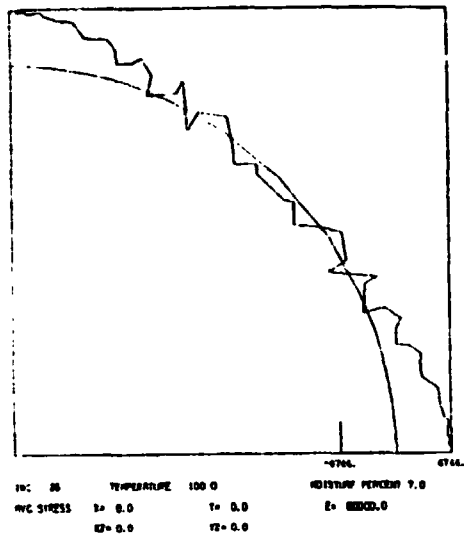
174



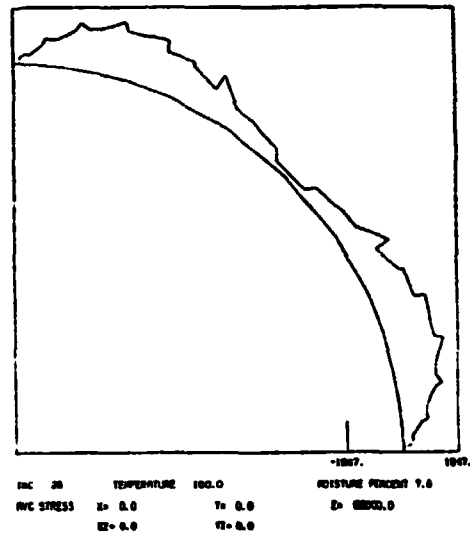
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

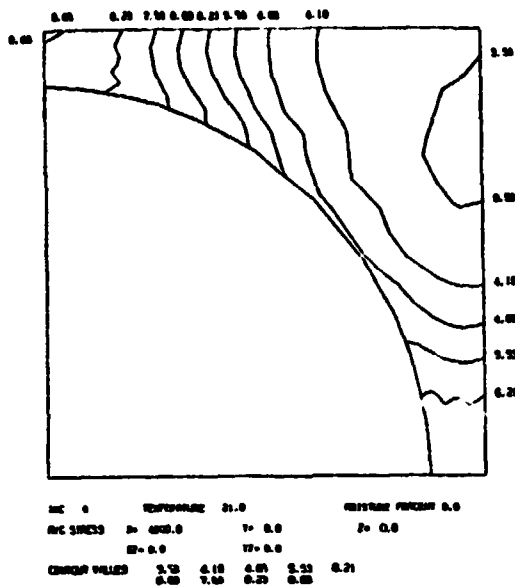


c) Interface Normal Stress (psi)

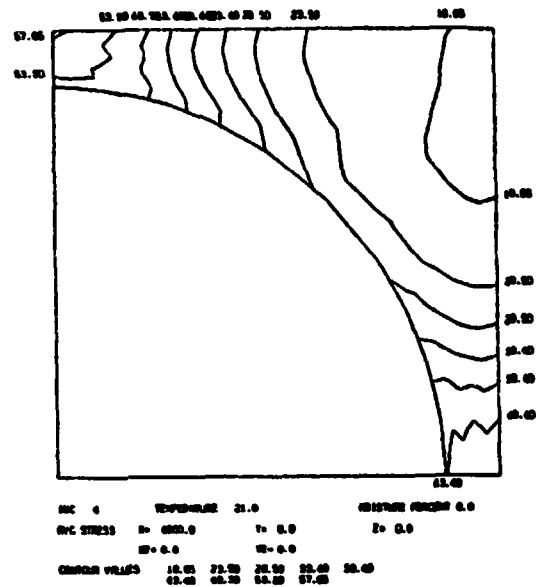


d) Interface Shear Stress (psi)

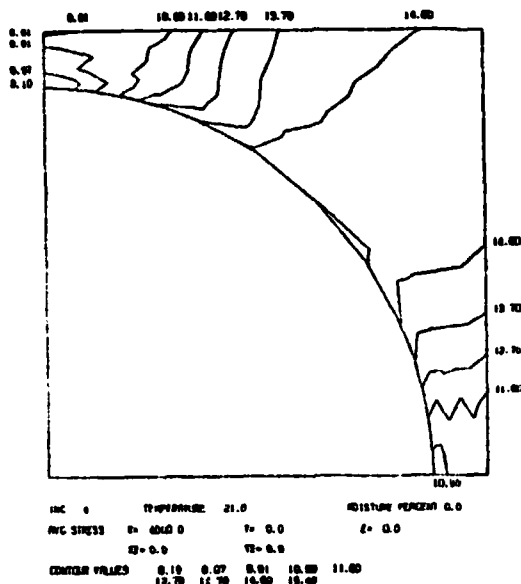
Figure E46. AS4/914 Graphite/Epoxy Unidirectional Composite, 100°C, 7.0 Percent Moisture (ETW); 0.55 GPa (80 ksi) Longitudinal Tensile Applied Stress.



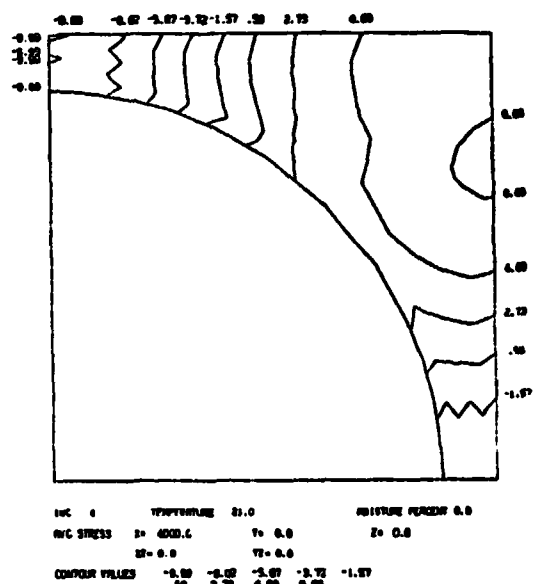
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

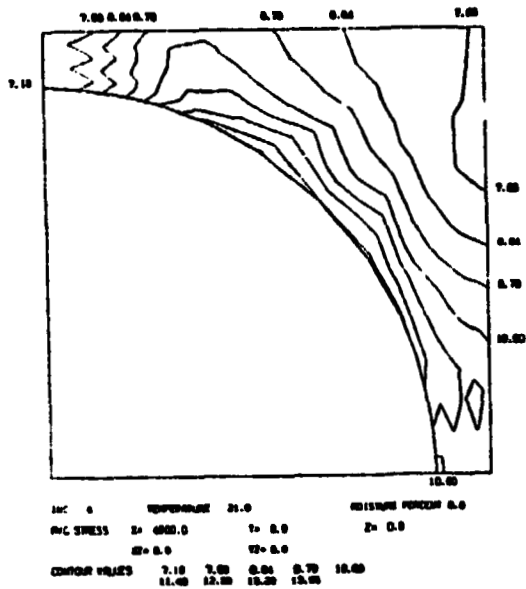


c) Minimum Principal Stress (ksi)

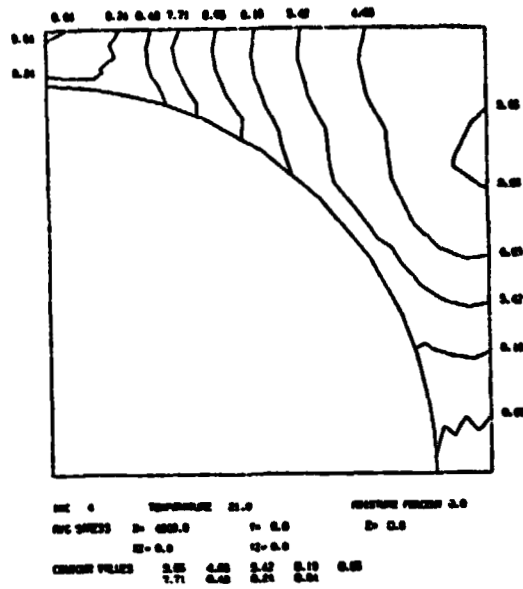


d) Intermediate Principal Stress (ksi)

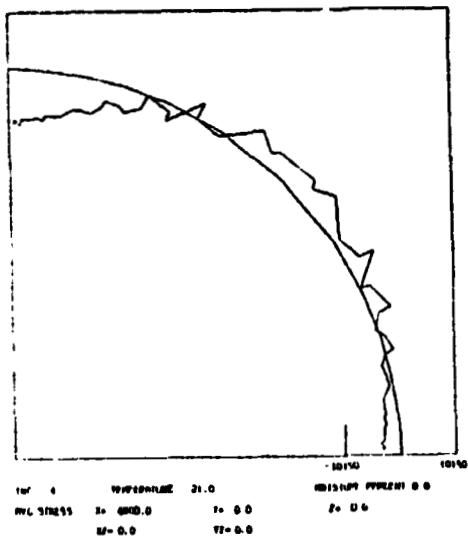
Figure E48. AS4/914 Graphite/Epoxy Unidirectional Composite, Room Temperature, Dry (RTD); 27.6 MPa (4 ksi) Transverse Tensile Applied Stress.



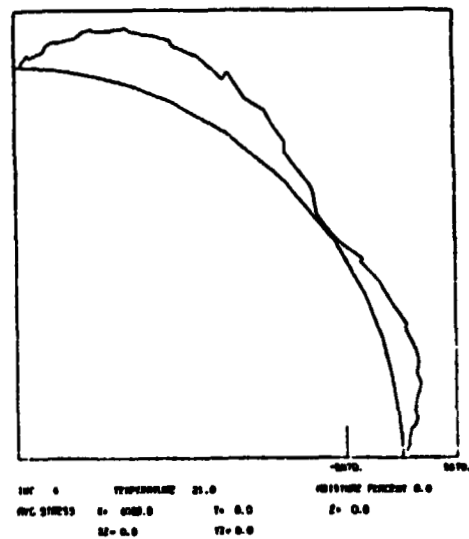
e) Intermediate Principal Stress (ksi)



f) Maximum Shear Stress (ksi)

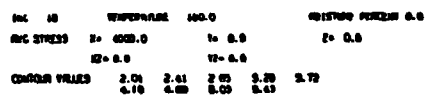


g) Interface Normal Stress (psi)



h) Interface Shear Stress (psi)

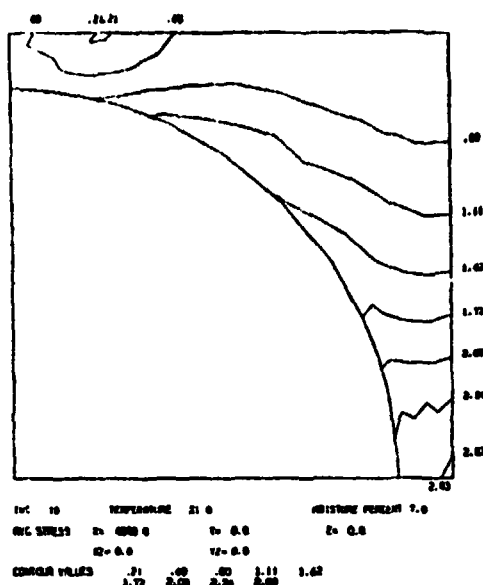
Figure E4C (continued). AS4/914 Graphite/Epoxy Unidirectional Composite, Room Temperature, Dry (RTD); 27.6 MPa (4 ksi) Transverse Tensile Applied Stress.



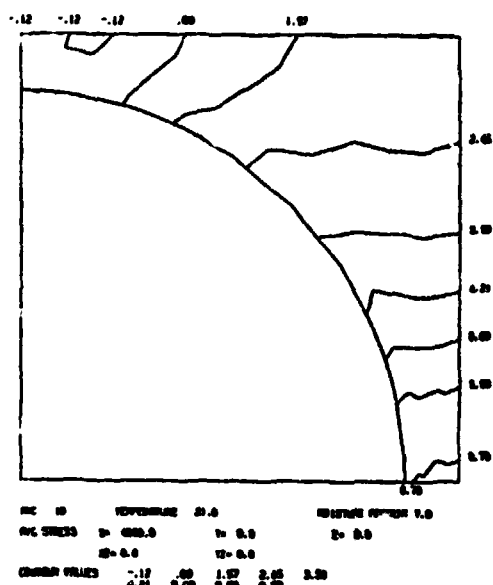
| MC | 10 | PERCENTAGE | 100.0 | PERCENTAGE | 100.0 |
|-------------|------|------------|-------|------------|-------|
| MC STRESS | 10 | 100.0 | 10 | 10 | 100.0 |
| | | 100.0 | | 100.0 | |
| CORRELATION | 0.12 | 0.24 | 0.00 | 0.07 | 0.14 |
| | 0.00 | 0.00 | 0.00 | 0.00 | |

| | | | | |
|------------|-------|-----------|-------|------------------------|
| 100 | 10 | 174000000 | 100.0 | ADDITIONAL PERCENT 0.0 |
| AVC STRESS | 0.0 | 4000.0 | 10.0 | 0.0 |
| | 100.0 | | 100.0 | |

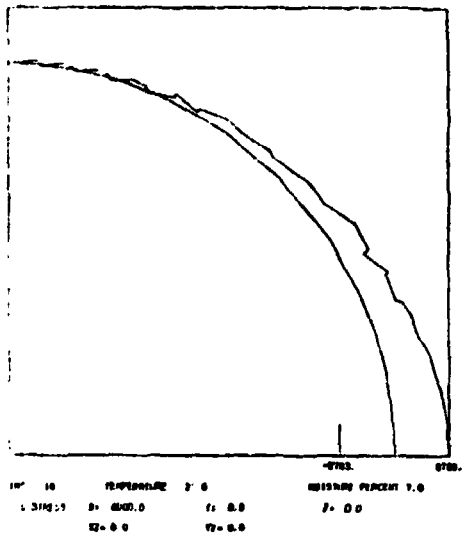
| INSTRUMENT | | -1057. | | 4057. | |
|-------------|-------------|----------|----------|---------|-----|
| INSTR. NO. | TEMPERATURE | 100.0 | POSITION | PERCENT | 0.0 |
| AVG. STRESS | 10 = 4000.0 | 10 = 0.0 | 10 = 0.0 | | |
| | 12 = 0.0 | 12 = 0.0 | | | |



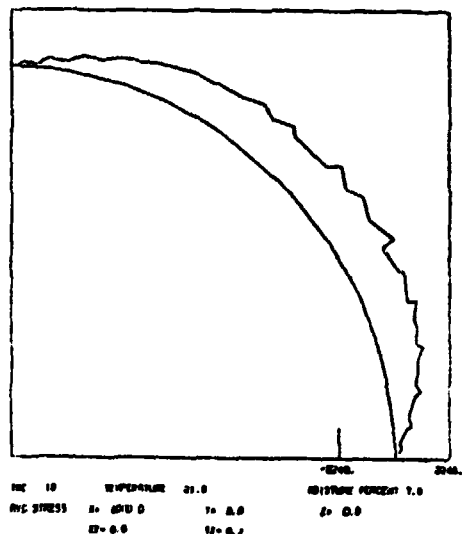
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

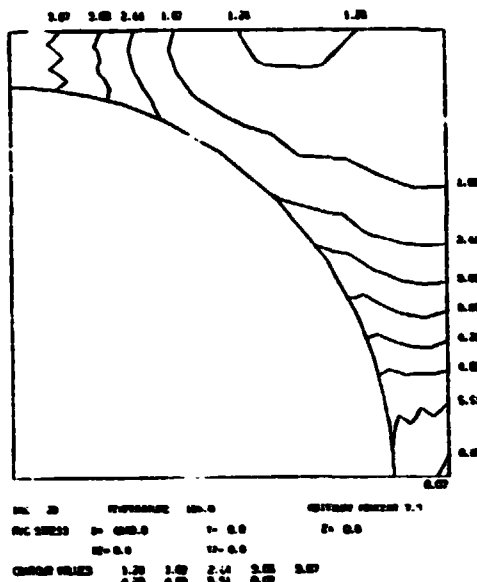


c) Interface Normal Stress (psi)

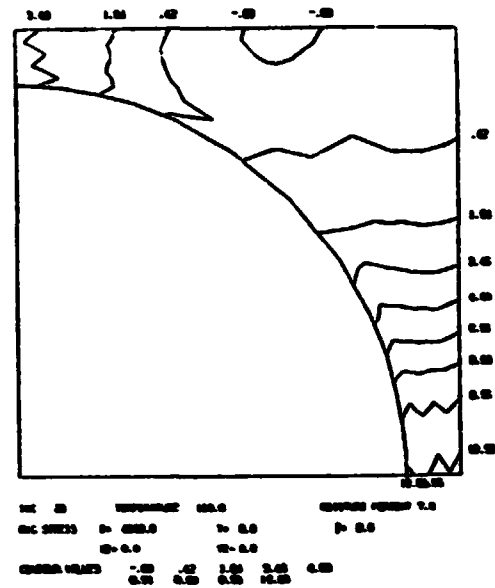


d) Interface Shear Stress (psi)

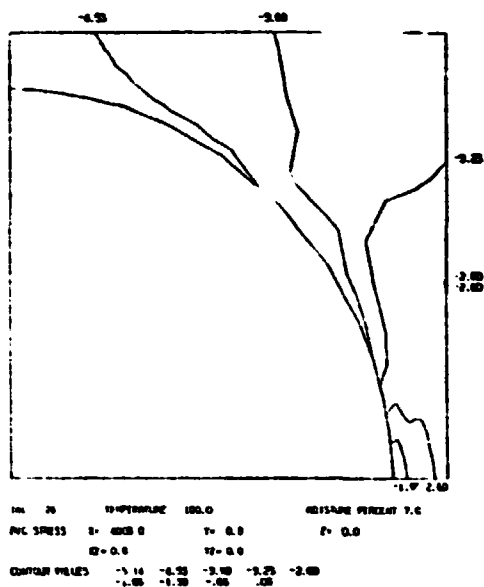
Figure E50. AS4/914 Graphite/Epoxy Unidirectional Composite, Room Temperature, 7.0 Percent Moisture (RTW); 27.6 MPa (4 ksi) Transverse Tensile Applied Stress.



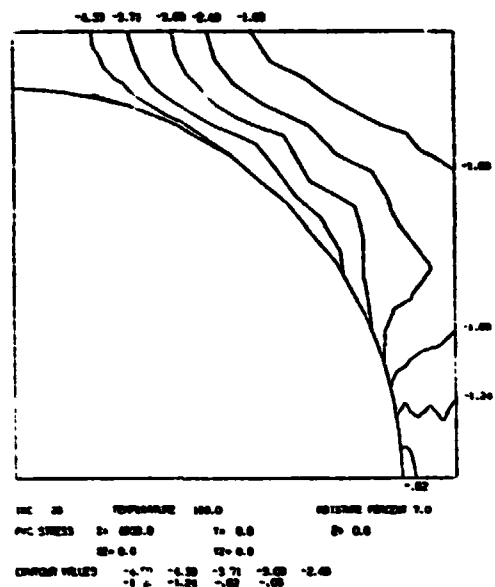
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

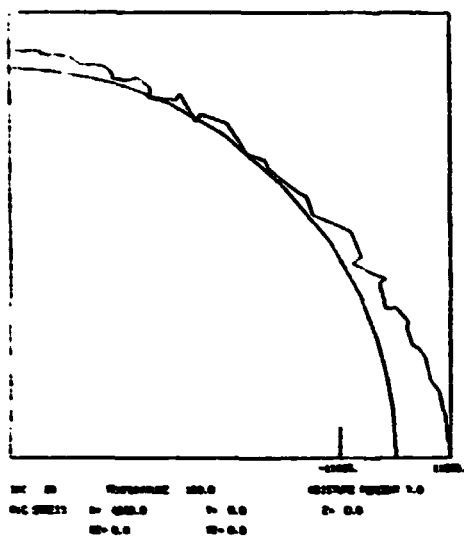


c) Minimum Principal Stress (ksi)

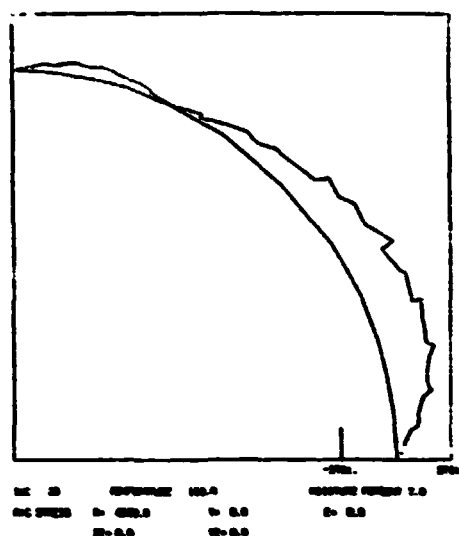


d) Intermediate Principal Stress (ksi)

Figure E51. AS4/914 Craphite/Epoxy Unidirectional Composite, 100°C, 7.0 pc cent Moisture (ETW); 27.6 MPa (4 ksi) Transverse Tensile Applied Stress.

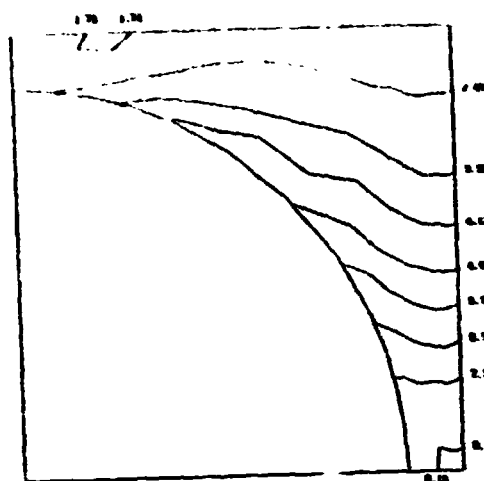


c) Interface Normal Stress (psi)



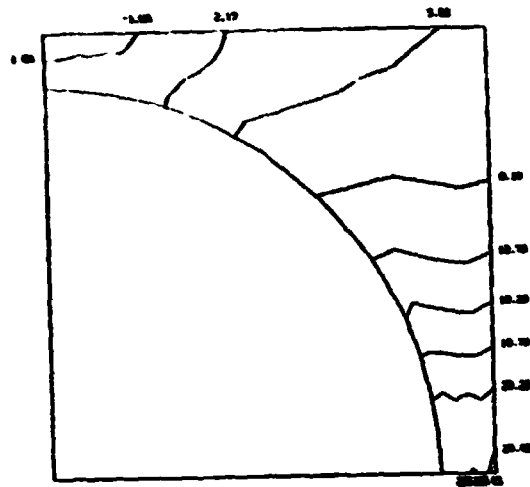
f) Interface Shear Stress (psi)

Figure E51 (continued). AS4/914 Graphite/Epoxy Unidirectional Composite, 100°C, 7.0 Percent Moisture (ETW); 27.6 MPa (4 ksi) Transverse Tensile Applied Stress.



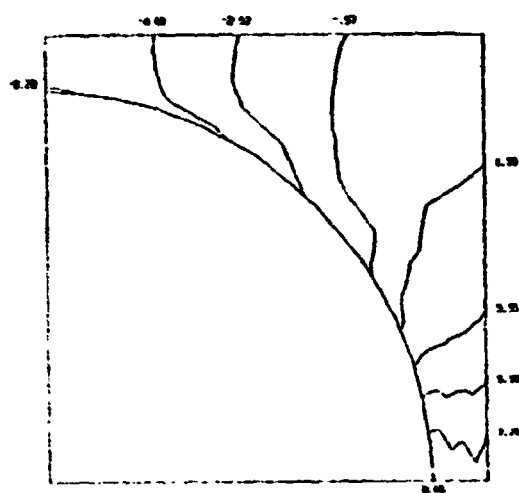
ELEM ID: 1000000000
 ELEM SIZE: 1000000000
 ELEM TYPE: 1000000000
 ELEM VALUE: 1.75 1.50 1.25 1.00 0.75 0.50 0.25 0.00

a) Octahedral Shear Stress (ksi)



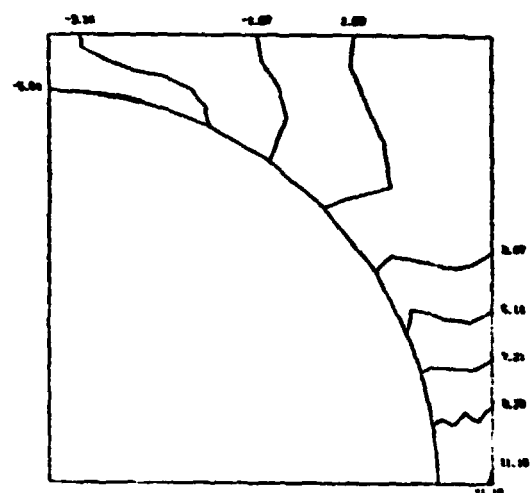
ELEM ID: 1000000000
 ELEM SIZE: 1000000000
 ELEM TYPE: 1000000000
 ELEM VALUE: 1.00 1.25 1.50 1.75 2.00 2.25 2.50 2.75 3.00

b) Maximum Principal Stress (ksi)



ELEM ID: 1000000000
 ELEM SIZE: 1000000000
 ELEM TYPE: 1000000000
 ELEM VALUE: 0.00 0.25 0.50 0.75 1.00 1.25 1.50 1.75

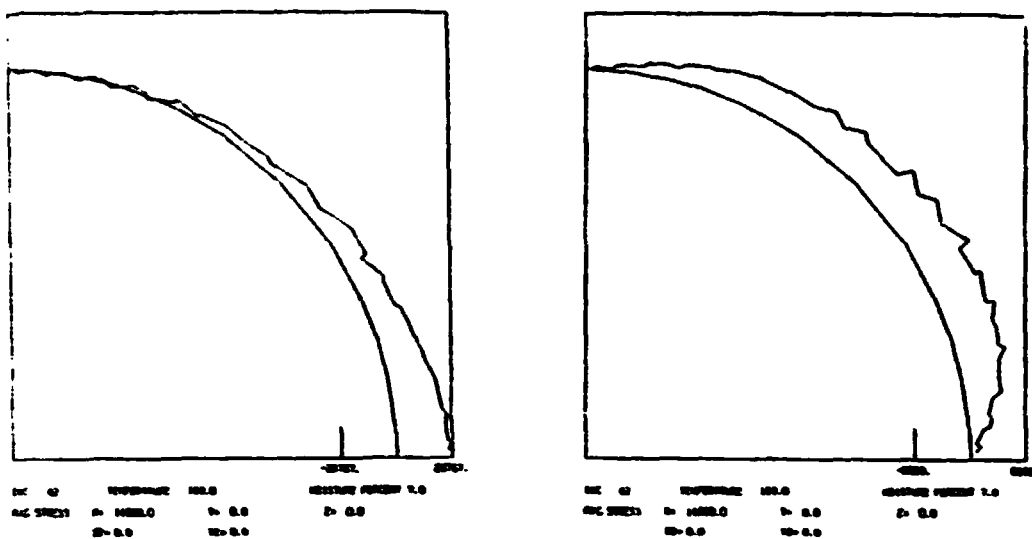
c) Minimum Principal Stress (ksi)



ELEM ID: 1000000000
 ELEM SIZE: 1000000000
 ELEM TYPE: 1000000000
 ELEM VALUE: 0.00 0.25 0.50 0.75 1.00 1.25 1.50 1.75

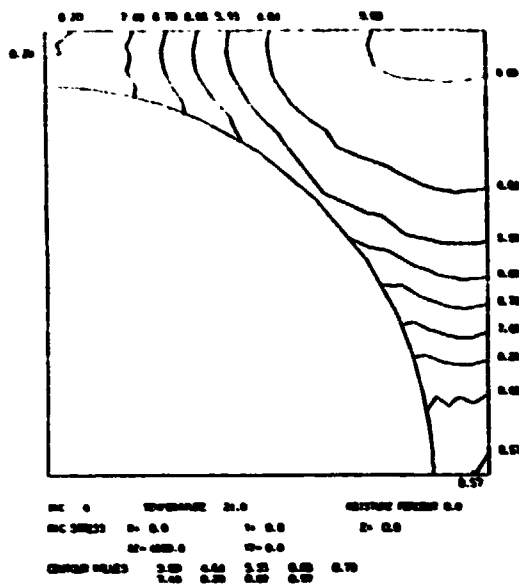
d) Intermediate Principal Stress (ksi)

Figure E52. AS4/914 Graphite/Epoxy Unidirectional Composite, 100°C, 7.0 Percent Moisture (ETW); 97 MPa (14 ksi) Transverse Tensile Applied Stress.

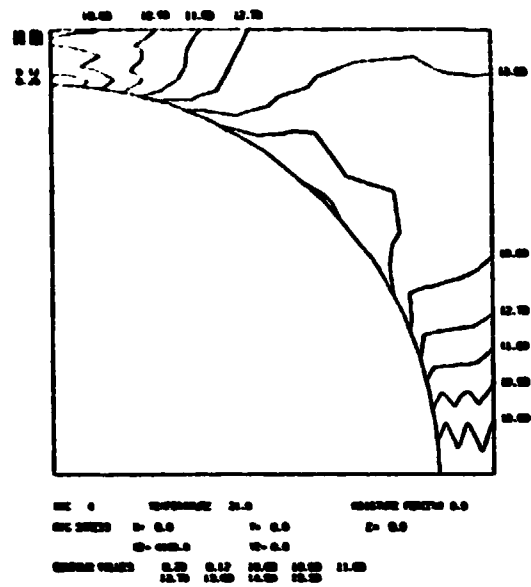


e) Interface Normal Stress (psi) f) Interface Shear Stress (psi)

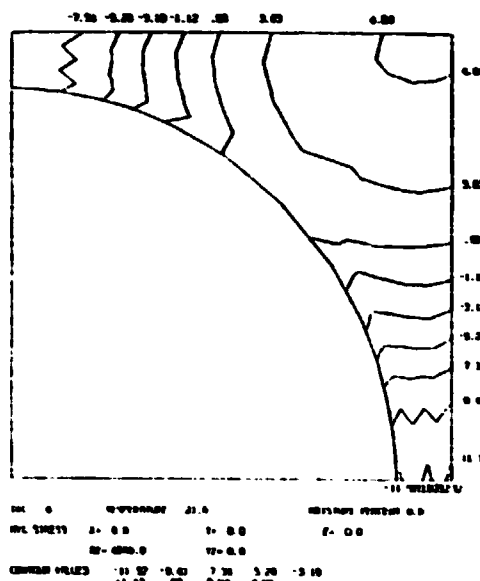
Figure E52 (continued). AS4/914 Graphite/Epoxy Unidirectional Composite, 100°C, 7.0 Percent Moisture (ETW); 97 MPa (14 ksi) Transverse Tensile Applied Stress.



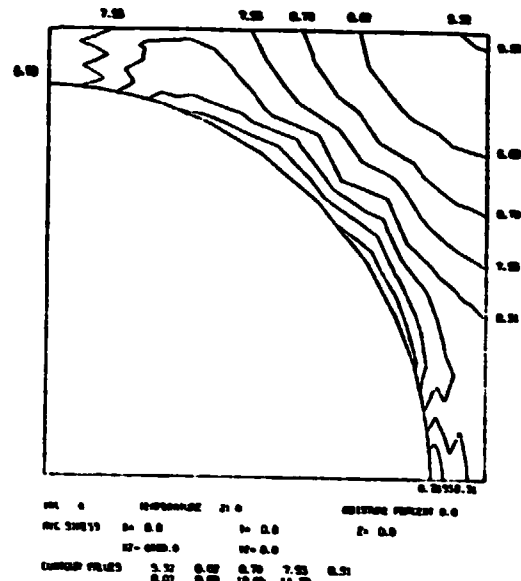
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

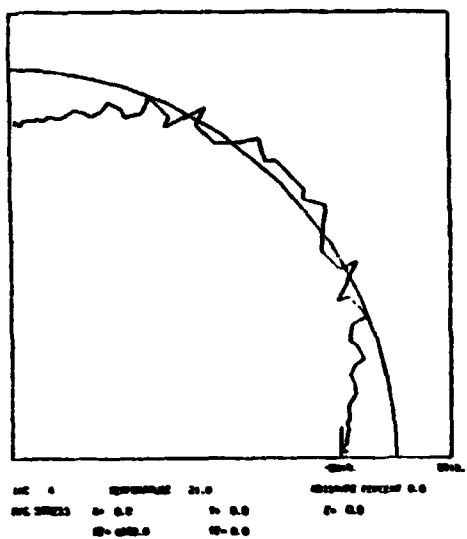


c) Minimum Principal Stress (ksi)

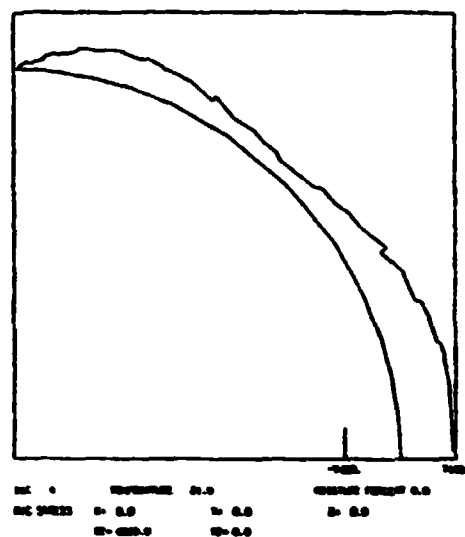


d) Intermediate Principal Stress (ksi)

Figure E53. AS4/914 Graphite/Epoxy Unidirectional Composite, Room Temperature, Dry (RTD); 27.6 MPa (4 ksi) Longitudinal Shear Applied Stress.

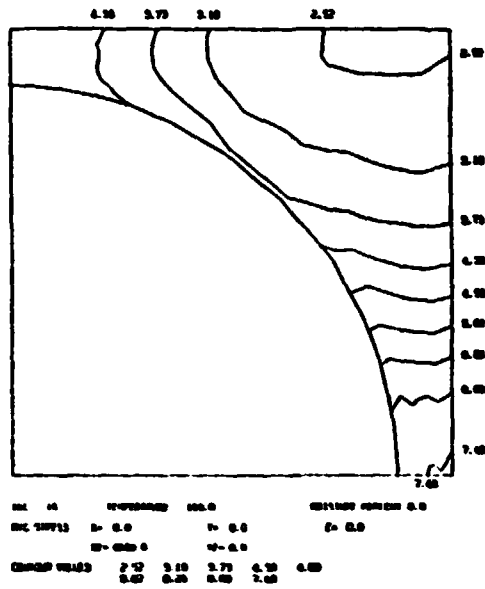


e) Interface Normal Stress (psi)

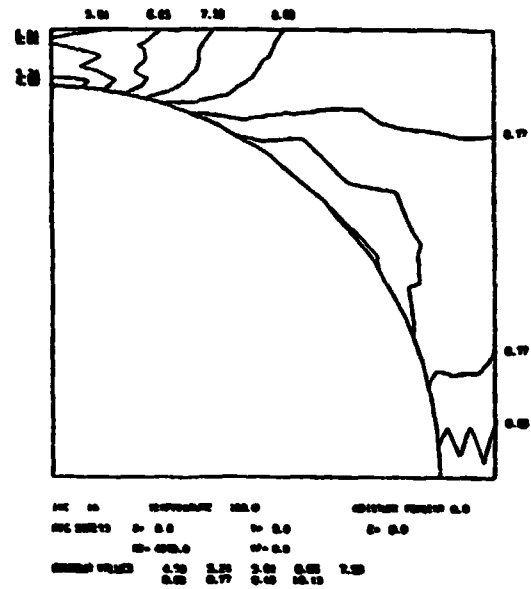


f) Interface Shear Stress (psi)

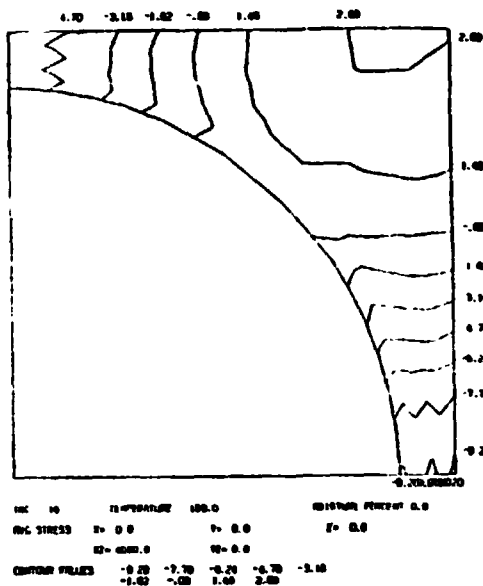
Figure E53 (continued). AS4/914 Graphite/Epoxy Unidirectional Composite, Room Temperature, Dry (RTD); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.



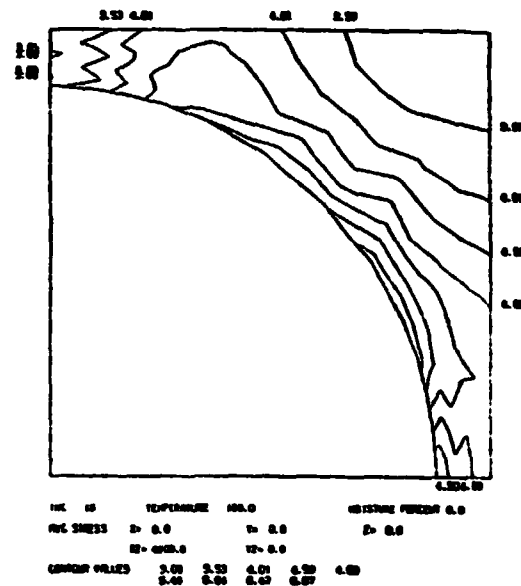
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

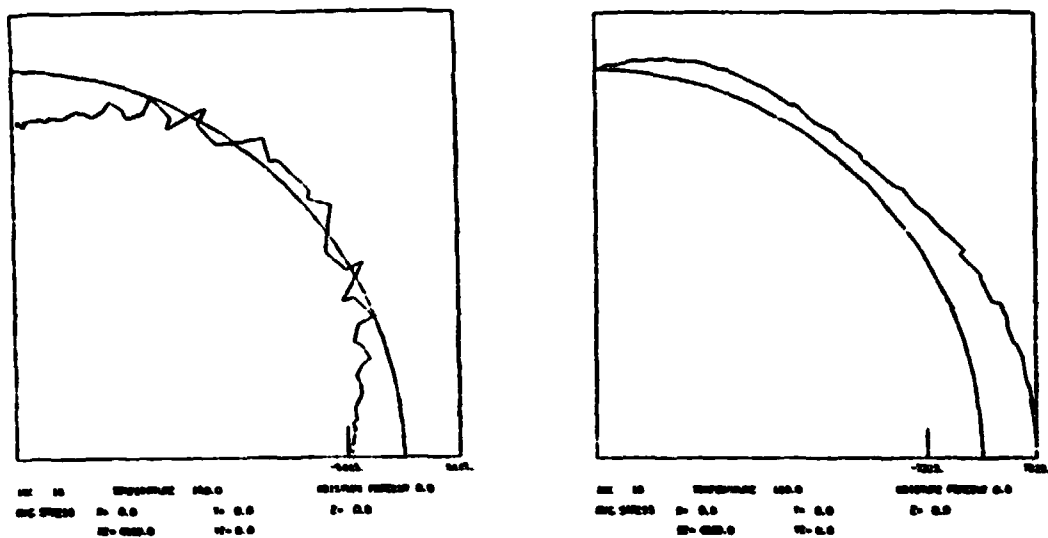


c) Minimum Principal Stress (ksi)



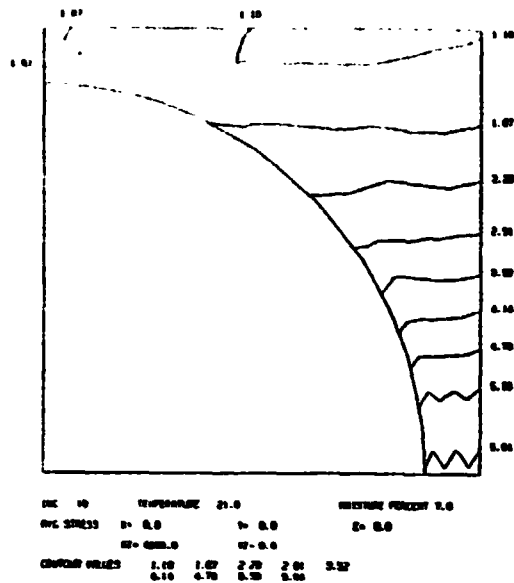
d) Intermediate Principal Stress (ksi)

Figure E54. AS4/914 Graphite/Epoxy Unidirectional Composite, 100°C Dry (ETD); 27.4 MPa (4 ksi), Longitudinal Shear Applied Stress.....

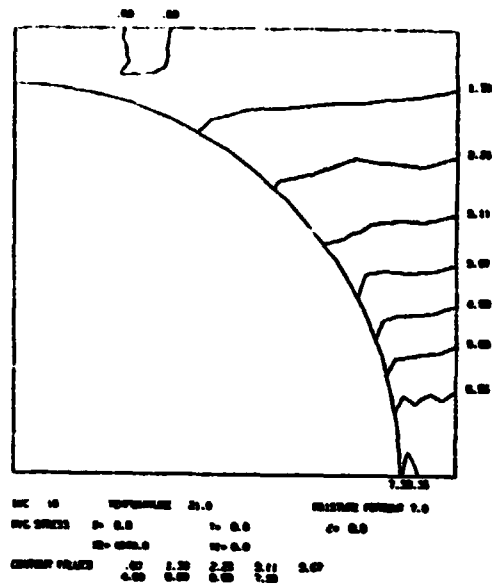


e) Interface Normal Stress (psi) f) Interface Shear Stress (psi)

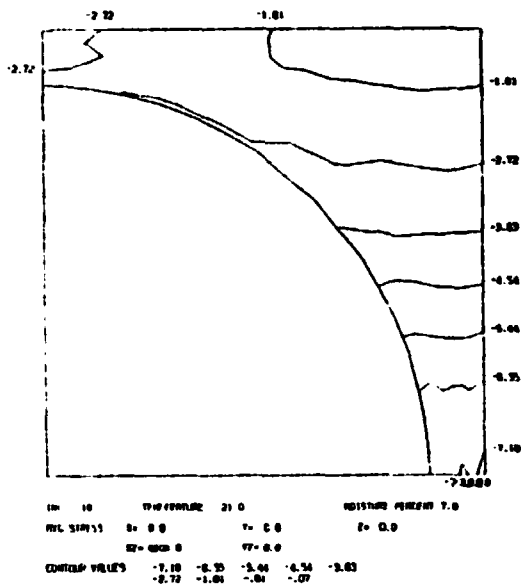
Figure E54 (continued). AS4/914 Graphite/Epoxy Unidirectional Composite, 100°C, 3.8 Percent Moisture; 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.



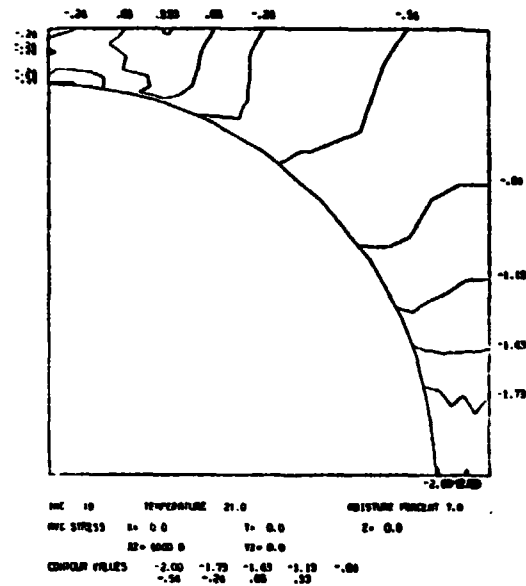
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

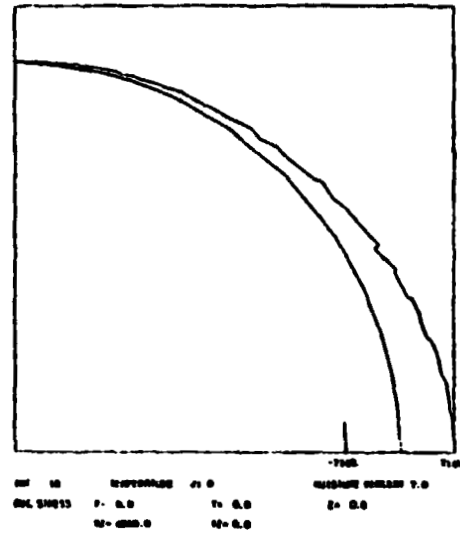
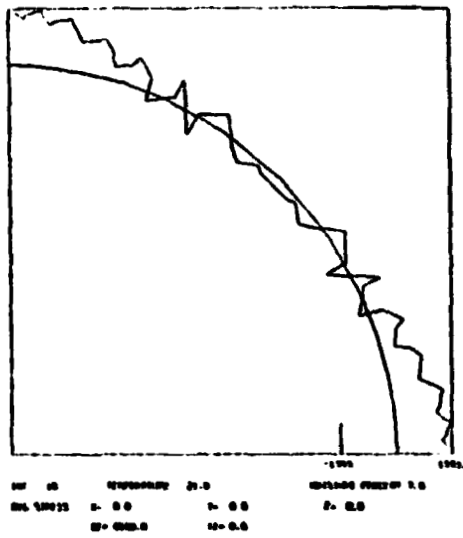


c) Minimum Principal Stress (ksi)



d) Intermediate Principal Stress (ksi)

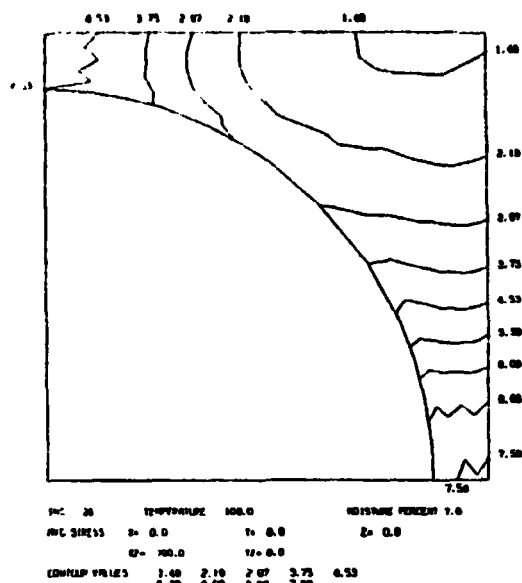
Figure E55. AS4/914 Graphite/Epoxy Unidirectional Composite, 21°C, 7.0 Percent Moisture (RTW); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.



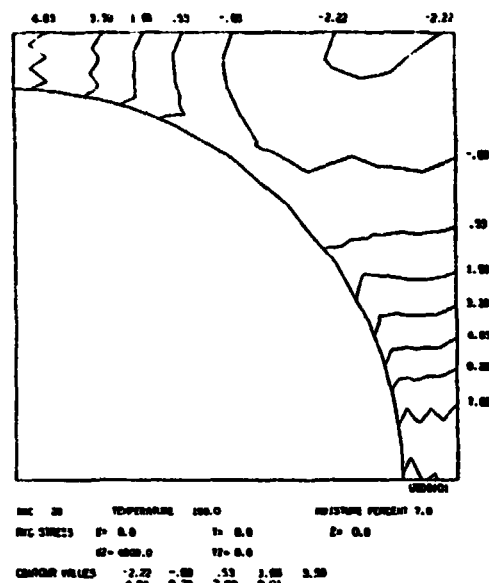
e) Interface Normal Stress (psi)

f) Interface Shear Stress (psi)

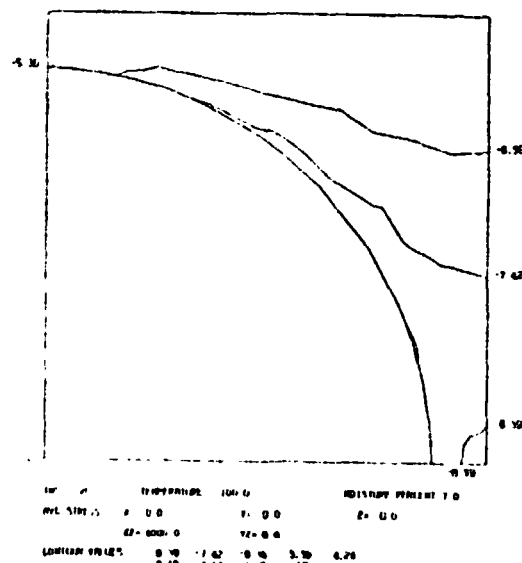
Figure E55 (continued). AS4/914 Graphite/Epoxy Unidirectional Composite, 21°C, 7.0 Percent Moisture (RTW); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.



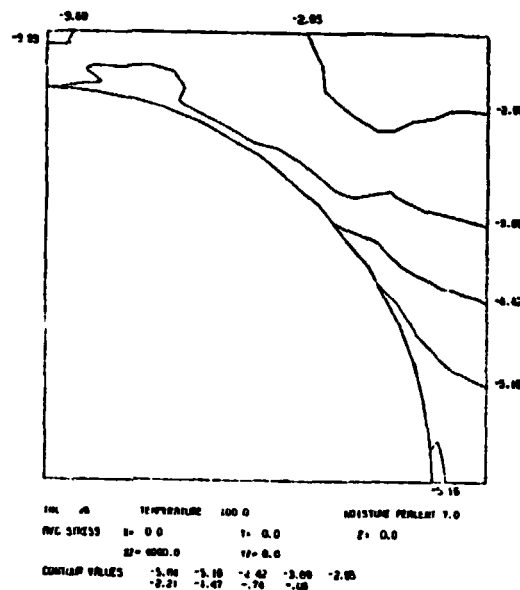
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

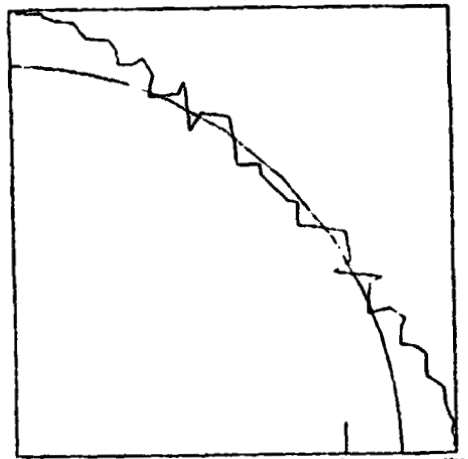


c) Minimum Principal Stress (ksi)

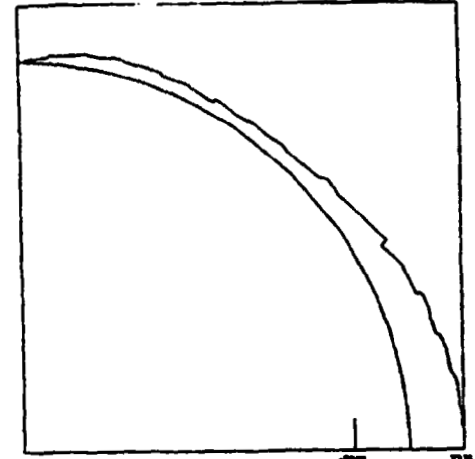


d) Intermediate Principal Stress (ksi)

Figure E56. AS4/914 Graphite/Epoxy Unidirectional Composite, 100°C, 7.0 Percent Moisture (ETW); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.

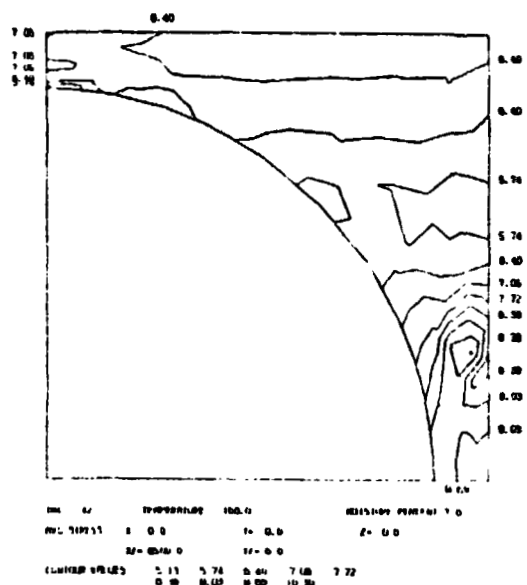


e) Interface Normal Stress (psi)

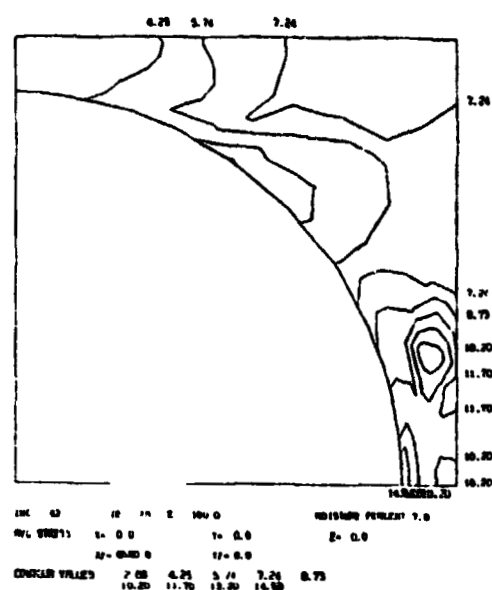


f) Interface Shear Stress (psi)

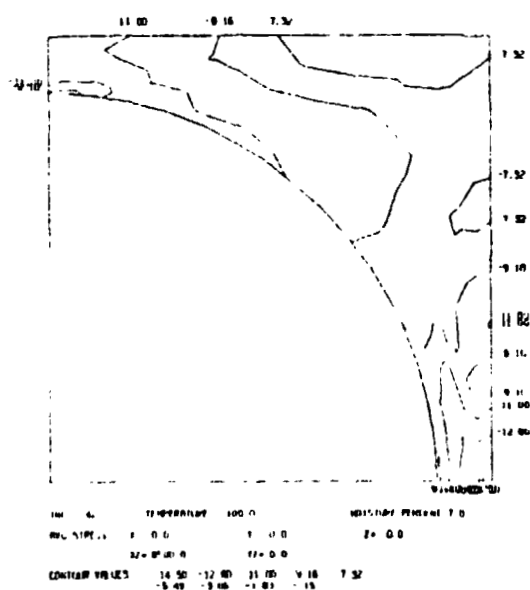
Figure E56 (continued). AS4/914 Graphite/Epoxy Unidirectional Composite, 100°C, 7.0 Percent Moisture (ETW); 27.4 MPa (4 ksi) Longitudinal Shear Applied Stress.



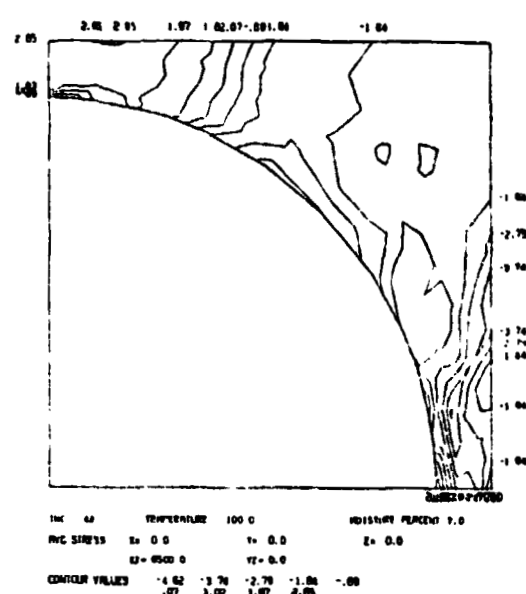
a) Octahedral Shear Stress (ksi)



b) Maximum Principal Stress (ksi)

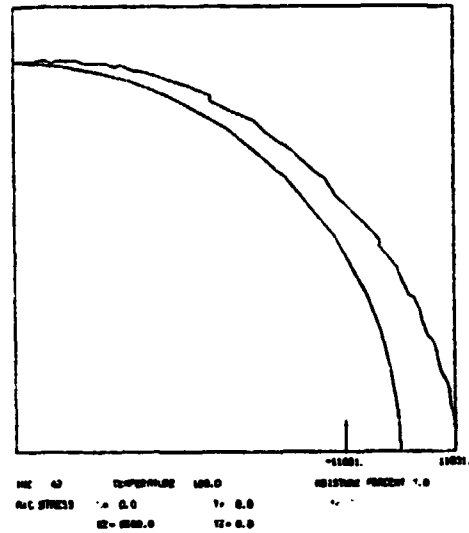
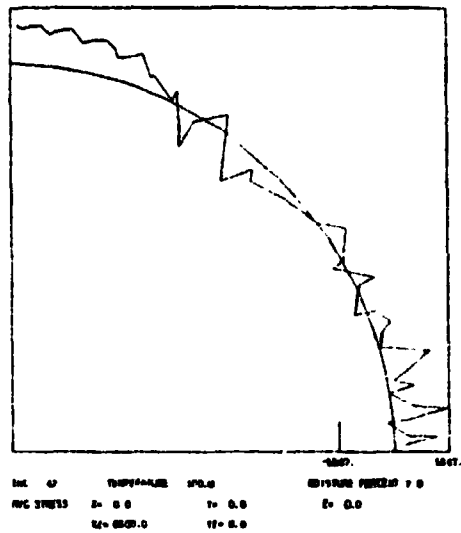


c) Minimum Principal Stress (ksi)



d) Intermediate Principal Stress (ksi)

Figure E57. AS4/914 Graphite/Epoxy Unidirectional Composite, 100°C, 7.0 Percent Moisture (ETW); 97 MPa (14 ksi) Longitudinal Shear Applied Stress.



e) Interface Normal Stress (psi)

f) Interface Shear Stress (psi)

Figure E57 (continued). AS4/914 Graphite/Epoxy Unidirectional Composite, 100°C, 7.0 Percent Moisture (ETW); 97 MPa (14 ksi) Longitudinal Shear Applied Stress.

AND
 DATE
 FEB. 28 1985